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# Design of a Planar, Concentric Coil for the Generation of a Homogeneous Vertical Magnetic Field Distribution

Ben Minnaert, Lieven De Strycker, and Nobby Stevens

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**Abstract** — This paper describes an algorithm to design a planar, concentric coil that generates a homogeneous vertical magnetic field distribution, with a target self inductance. The algorithm consists of several steps. The first step is the random generation of a large set of concentric coils, taken into consideration a number of boundary conditions such as the outer dimensions and the radius of the conducting wire. The optimal coil is found from this set after a twofold selection, namely on the self inductance value and on the uniformity of the vertical magnetic field, which was evaluated using a cost function. The algorithm is straightforward and based on closed expressions, which leads to a rapid execution of the procedure. We evaluated the algorithm on a representative reference configuration. The best coil of the large set was selected and the quasi-uniform field distribution of this coil was confirmed by experimental verification.

**Index Terms** — Magnetic field, planar coil, self inductance, uniform field distribution.

## I. INTRODUCTION

In several growing domains such as inductive wireless powering [1-4], Radio Frequency IDentification (RFID) [5] and Near Field Communication (NFC) [6], two inductively coupled coils are used to transfer energy and to exchange information over a short-range, wireless link. In inductive wireless powering systems, the energy transfer dominates the design, while for RFID or NFC the emphasis is on the data transfer. In both approaches, an increased homogeneity of the magnetic field is a favorable property. For example, if the transmitter coil for an inductive wireless power transfer system generates a more homogeneous field, it enhances the flexibility of positioning the receiver coil [7]. This can be beneficial for the charging of low-power consumer electronics such as smartphones [8-10], but also for higher power applications, such as the on-road charging of electrical vehicles while they are driving [11, 12]. An increased homogeneity of the field is also for RFID and NFC applications beneficial: it increases the readout

region and efficiency of information exchange [13], which is an advantage for, e.g., moving receivers. An example application is the registering of the RFID tags of cattle at an automatic feeding installation [14].

Besides the flat field distribution, it is also important to take the self inductance into account. If one wants to replace an existing coil with an optimized coil with a more homogeneous field, it will have a minimal impact on the rest of the circuitry if the self inductance is not modified.

In this study, we will develop an algorithm to design a concentric, fully planar, transmitter coil that leads to a high and quasi-uniform magnetic field distribution with a specific self inductance value. The result of the algorithm is a set of dimensional parameters that can be used directly to design the coil. Limitations imposed by, e.g., the wire radius, are taken into account.

Interesting work on the design of coils for the generation of quasi-uniform magnetic field generation has been published in recent years by various authors. Kim et al [15] designed a planar coil of multiple loops connected in series and parallel. They mix the current direction in each loop in forward and in reverse, thus generating a more uniform magnetic field distribution. The same alternate winding design is applied by Zhang and Chau [16] to continuously charge a moving receiver. They apply different transmitter coils next to each other with alternating current directions, to minimize the gap in the field distribution between different transmitter coils. In this way they create a more uniform magnetic field which allows for an enhanced energy transfer performance around the gap between the coils.

Casanova et al. [17] designed a dedicated coil of 20 cm by 20 cm for inductive wireless power transfer. The coil design is a flat spiral. The geometry is determined by sweeping all possible parameters, evaluating the magnetic field for all possibilities, and choosing the design that leads to the most flat field distribution using an objective function. An analogous procedure was used by Yinlian et al. [18] to design two coil structures that realize in parallel a flat field distribution.

Liu and Hui [19] have patented a technique, based

on a hybrid coil structure. It consists of a concentric coil and a spiral winding. A number of limitations are posed here, since the spacing between the spiral coil loops must be constant. It is also observed that the spiral coil is not located in the same plane of the concentric coil, which makes the structure inherently non-planar, which can be considered less attractive for most applications due to the increased height of the coil. The same applies for the work of Lee et al. [20] who proposed a technique to bend a rectangular coil further away from the receiving coil at the edges of this structure. By realizing a more uniform magnetic field distribution, they increased the overall transfer efficiency up to 44% compared to a conventional coil, regardless of the location of the receiving coil.

Waffenschmidt [7] developed an excellent algorithm to determine the turn distribution of a planar coil to generate a specified homogeneous field. In the first step, his algorithm calculates a current distribution from a specified magnetic field. In the second step, the corresponding turn distribution to this current distribution is derived. Using the generalized minimal residual method, this algorithm can be extended to more coil turns [21]. Azpúra [22] proposed a semi-analytical method for designing a coil that generates a more uniform field. Starting from the Biot-Savart law, he analytically determines the field distribution of a start configuration. He then uses an iterative algorithm based on TABU search to improve the homogeneity of the field. The number of turns of each coil and the thickness of wire is adjusted at each iteration until the required field is reached.

An important difference between our algorithm and the aforementioned references is that we impose a certain, by the user determined, self inductance for the planar coil. In that way, our improved coil will have a minimal impact on the rest of an existing circuitry since the self inductance is not modified. Previous related work always managed to increase the homogeneity of the field by, among others, changing the inductance. Moreover, as a result of the availability of closed expressions for the field distribution and self inductance, our algorithm can be evaluated in a very short time.

This paper is organized as follows. In Section II, the expressions for the field evaluation, the self inductance and the homogeneity of the field for a circular, concentric, planar coil are elaborated. These expressions are used extensively in the algorithm that is described in detail in Section III, which describes in detail the flow in order to obtain the planar coil with the best field distribution and a specific target self inductance. The algorithm is demonstrated on a representative example in Section IV, where the results are also verified experimentally. The main results are summarized in the conclusions (Section V).

In this study, we limit our attention to a concentric coil. The reason is that when a certain degree of freedom

is offered to the user to position its electronic device, the most flexible solution is the one where there is no rotational preference direction. This obviously leads to a concentric configuration.

## II. FIELD DISTRIBUTION AND INDUCTANCE CALCULATION

The vertical magnetic field generated by a circular current loop with radius  $R$  and current  $I$  is available in a closed form. If the origin of the coordinate set is chosen at the center of the circular loop, with the loop in the  $xy$ -plane, Equation (1) is applicable [23]:

$$H_z(r, z) = \frac{I}{2\pi} \frac{1}{\sqrt{(r+R)^2+z^2}} \cdot \left\{ \mathcal{K}(k) - \frac{r^2-R^2+z^2}{(r-R)^2+z^2} \mathcal{E}(k) \right\}, \quad (1)$$

with

$$k = \frac{2\sqrt{rR}}{\sqrt{(r+R)^2+z^2}}. \quad (2)$$

Here,  $r$  and  $z$  are the cylindrical coordinates of the observation point, while  $\mathcal{K}(k)$  and  $\mathcal{E}(k)$  are the complete elliptic integrals of the first and second kind respectively. When multiple, concentric loops are present, as shown on Fig. 1, the field is found as the sum of the different loop contributions. Without loss of generality (due to linearity), we normalize the field at a current of 1 A. We suppose that the same current is flowing counterclockwise (top view) in the different loops. Equation (3) thus describes the vertical magnetic field generated by  $N$  concentric coils with different radii  $R_n$ :

$$H_z(r, z) = \frac{1}{2\pi} \sum_{n=1}^N \frac{1}{\sqrt{(r+R_n)^2+z^2}} \cdot \left\{ \mathcal{K}(k_n) - \frac{r^2-R_n^2+z^2}{(r-R_n)^2+z^2} \mathcal{E}(k_n) \right\}, \quad (3)$$

with

$$k_n = \frac{2\sqrt{rR_n}}{\sqrt{(r+R_n)^2+z^2}}. \quad (4)$$

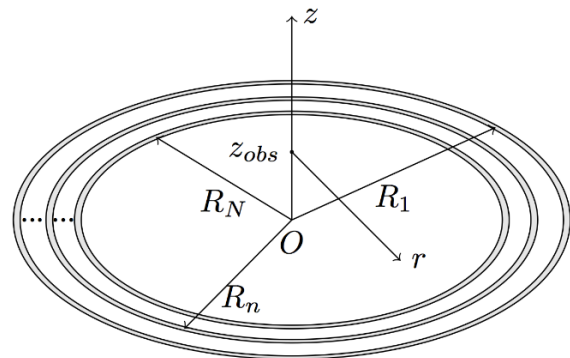


Fig. 1. Configuration under study with the corresponding conventions.

Since we want to design a planar, concentric coil with a certain target self inductance  $L$ , we need a closed expression for this lumped parameter. If the radius of the wire  $r_w$  is much smaller than the radius of the loop  $R_n$ ,

the self inductance  $L$  can be approximated by [24]:

$$L = \sum_{n=1}^N \mu_0 R_n \left[ \ln \left( \frac{8R_n}{r_w} \right) - 2 \right] + \sum_{i=1}^N \sum_{j=1}^N M_{i,j} (1 - \delta_{i,j}), \quad (5)$$

with

$$M_{i,j} = \mu_0 \sqrt{R_i R_j} \left[ \left( \frac{2}{k_{i,j}} - k_{i,j} \right) \mathcal{K}(k_{i,j}) - \frac{2}{k_{i,j}} \mathcal{E}(k_{i,j}) \right], \quad (6)$$

and

$$k_{i,j} = \frac{2\sqrt{R_i R_j}}{R_i + R_j}. \quad (7)$$

$\delta_{i,j}$  is the Kronecker delta (zero for  $i \neq j$  and 1 if  $i = j$ ). In order to have a planar structure, the radial difference between any two different loops should be at least two times the radius of the wire  $r_w$ :

$$|R_i - R_j| \geq 2r_w \text{ for } i \neq j. \quad (8)$$

Based on previous expressions, we can construct a planar, circular concentric coil with a certain self inductance  $L$ . Once the dimensions of the different loops are chosen, the field at any height  $z_{obs}$  can be evaluated making use of expression (3). The goal is however, that an optimized coil with a specific self inductance  $L$  is defined, which leads to a homogeneous vertical field distribution at a specific observation height  $z_{obs}$ . In order to quantify the homogeneity of the field in an interval between  $r_{min}$  and  $r_{max}$ , we define the following cost function  $v$ :

$$v = \int_{r_{min}}^{r_{max}} \left| \frac{dH_z(r, z_{obs})}{dr} \right| dr = \int_{r_{min}}^{r_{max}} |dH_z(r, z_{obs})|. \quad (9)$$

It is clear that  $v$  will be minimal for flat and thus, homogeneous field distributions between  $r_{min}$  and  $r_{max}$ .

### III. ALGORITHM

In our algorithm, we choose the maximal outer dimension of the coil, being  $R_1 + r_w$ , fixed. It has been proven [25] that the *distribution* of the magnetic field for a circular current loop is independent on the outer radius  $R$ . Indeed, expressions (1) and (3) can be rewritten as a function of two dimensionless parameters, the normalized radius and the normalized height, with  $R$  as the normalization variable [25].  $R$  then acts as a scaling factor and does not modify the distribution.

The goal of the algorithm is to find a concentric, planar configuration with a given exterior dimension  $R_1 + r_w$  and a given self inductance  $L$  that generates a homogeneous field distribution. If one wants to design a planar, concentric coil with a specific inductance using as less wire length as possible, it can be derived from expression (5) that all current loops must be located at the utmost exterior. We will call this configuration the reference coil with exterior dimension  $R_1 + r_w$  (Fig. 2 shows such a coil with  $N_{ref} = 10$  number of loops). Figure 3 shows the typical magnetic field distribution for such a configuration. One notices that the magnetic field in the center is lower than near the edges. It is clear that in order to realize a more uniform vertical field distribution, we have to increase the field in the central

region. If we consider a second configuration where we fill the entire central region with current loops (i.e., a “disk configuration”), we obtain a magnetic field that is higher in the center than at the outer edges (Fig. 3). In this latter case, the different loops are closely stacked to each other and the different values (for  $n$  from 1 to  $N$ ) are:

$$R_n = r_w + (N - n)2r_w. \quad (10)$$

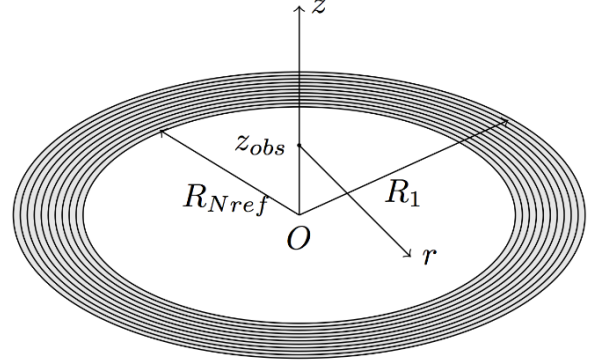


Fig. 2. Reference coil with all loops at the utmost exterior.

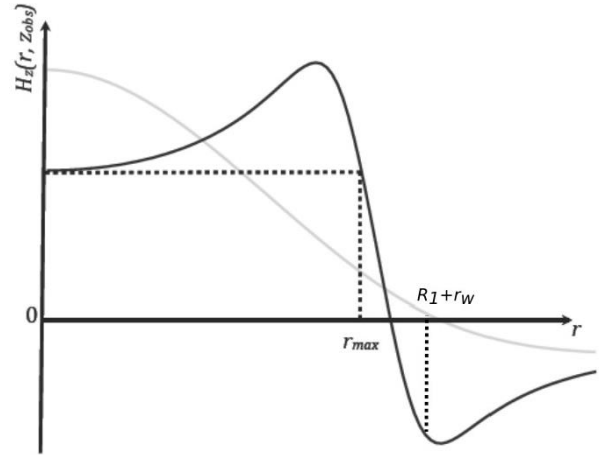


Fig. 3. Typical field distributions: all loops at the outer edge (black curve) and all loops lowest possible radius (gray curve).

Both typical field distributions are also found by other authors, e.g., [7]. Based on the distributions of Fig. 3, we can now propose a strategy to obtain a more homogeneous field distribution, with a given exterior dimension  $R_1 + r_w$  and given self inductance  $L_{ref}$ . If we consider the reference coil and simply add extra loops in the central region, we obtain a more homogeneous field distribution. However, this would increase the self inductance. Therefore, in order to keep the self inductance fixed to the self inductance of the reference coil, we also will have to decrease the number of loops at the outer



edge. One can see, based on Equations (5) and (8), that the total number of loops of the optimized coil  $N_{opt}$  will be larger than  $N_{ref}$ .

The goal of the algorithm is to define a planar, concentric coil that leads to a significant more uniform  $z$ -oriented magnetic field. Therefore, we have to define a lateral region where this condition has to be realized, i.e., an interval over which the cost function  $v$  (Equation (9)) is to be evaluated. Since we want a field as homogeneous as possible over an area as large as possible, we choose the value of  $r_{min}$  to be 0, and let the value of  $r_{max}$  be determined by the reference coil. We define  $r_{max}$  as the radial distance where the decaying field becomes smaller than the value found in the center in the case of the reference coil, as illustrated graphically on Fig. 3. Above this radius  $r_{max}$ , the variation of the field is no longer relevant.

For the input of the algorithm, the following parameters are given by the user: the required self inductance  $L_{ref}$ , the radius of the wire  $r_w$ , the outer dimension of the planar coil  $R_1 + r_w$  and the vertical distance  $z_{obs}$  where the magnetic field is to be observed. The algorithm is split in two steps. First, a set of concentric coils is created that have a self inductance that is close to the value of the required self inductance  $L_{ref}$ . In the second step, the configuration within this set that delivers the most uniform  $H_z(r, z_{obs})$  (where  $r$  runs from 0 to  $r_{max}$ ) is determined.

On Fig. 4, the flowchart of the algorithm is shown. At the start, the user inputs the values for  $L_{ref}$ ,  $r_w$ ,  $R_1 + r_w$  and  $z_{obs}$ . From these data, the algorithm calculates for all possible reference coils (i.e., coils with all loops exterior) the corresponding inductance, based on expression (5). It then selects the coil with the inductance closest to  $L_{ref}$ . We call this the reference coil with number of loops  $N_{ref}$ . From this reference coil, the value of  $r_{max}$  is calculated (using Equation (3) and the given observation height  $z_{obs}$ ).

The next task to be executed is the generation of a large set of planar coils. Therefore, the number and radii of the loops is modified and a random generator determines a new set of radii. The limitation on the random generator is that the difference of the values of the radii  $R_n$  must be at least  $2r_w$  in order to keep the structure planar. The random generator returns uniformly distributed random numbers in the specified interval. We define  $N_{ext}$  as the number of loops that is kept at the most outside position and  $N_{int}$  as the number of loops that have a radius smaller than  $R_1 - 2N_{ext}r_w$ . These loops are thus located in the central part of the coil. Figure 5 shows an example, where  $N_{ext}$  equals 8 and  $N_{int}$  is 3. The total number of loops  $N_{opt}$  is of course  $N_{ext} + N_{int}$ . The only limitation towards the radii of the inner loops is that they must differ by at least  $2r_w$ . The difference between the interior loop radii values can vary.

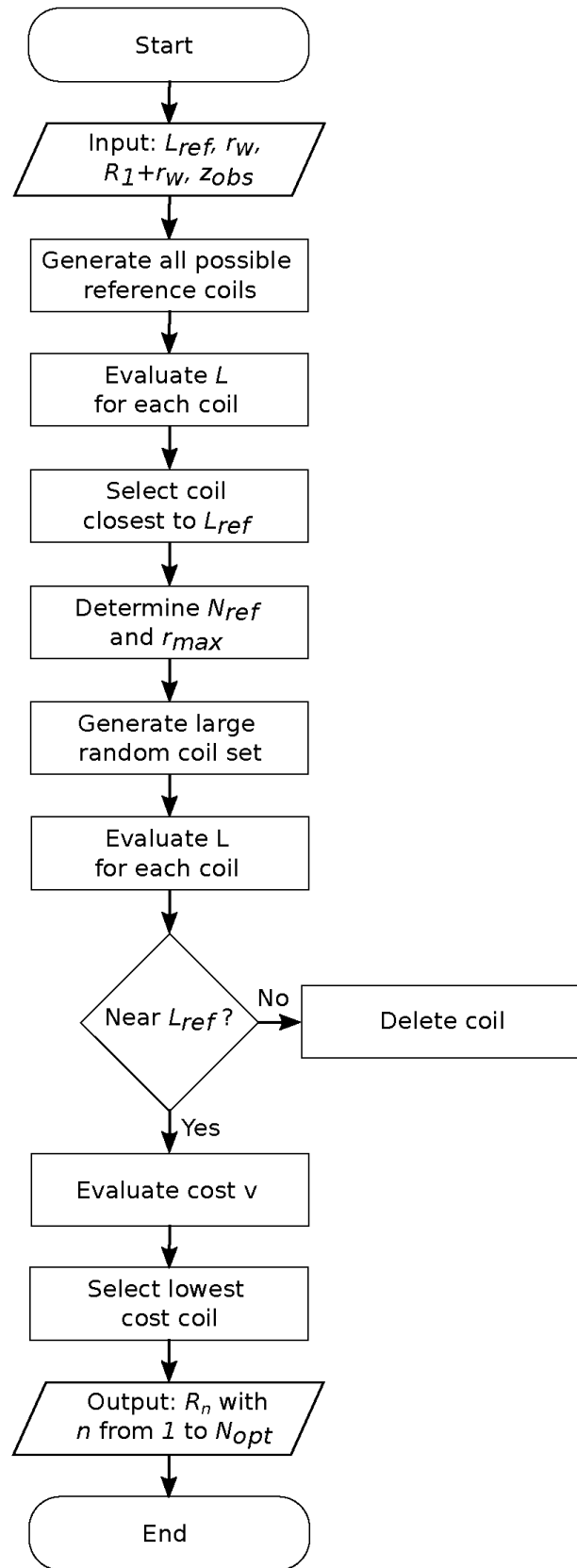


Fig. 4. Flowchart of the algorithm.

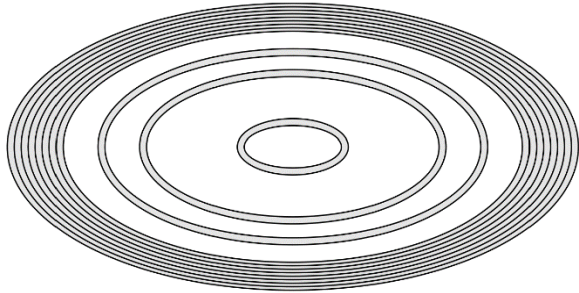


Fig. 5. Planar concentric coil with  $N_{ext} = 8$  and  $N_{int} = 3$ .

Then the self inductance  $L$  of each coil is calculated by using the closed expression (5). Based on the calculated self inductance values, we decide which coils can be deleted and which ones are selected for further evaluation. The user has a certain degree of freedom here, but a relative difference of a couple of percent seemed a reasonable choice. For example, the Agilent 42851 Precision LCR meter has, depending on the measurement conditions, an inductance measurement accuracy of about 2%. Therefore,  $L \in [0.98 L_{ref}, 1.02 L_{ref}]$ , which allows a maximum relative difference of 2%, are the boundaries we will choose in Section IV. At this point of the procedure, we have a large set of concentric coils with a self inductance that is close to the user specified  $L_{ref}$ .

The next step is the evaluation of the cost function  $v$  (expression (9)) for the remaining coils to find the current loop distribution that leads to the most uniform field for  $r$  between 0 and  $r_{max}$ . For the numerical evaluation of  $v$ , we approximate the integral of Equation (9) by a finite sum, as described by equation (11). Here,  $r_0$  equals 0,  $r_{M+1}$  equals  $r_{max}$  and  $M$  is large. We have observed that this sum converges rapidly as  $M$  approaches several hundreds of evaluation points between 0 and  $r_{max}$ :

$$v \approx \sum_{m=0}^M |H_z(r_{m+1}) - H_z(r_m)|. \quad (11)$$

Since this corresponds with the midpoint Riemann sum, the order of magnitude of the error as a consequence of this approximation is  $\mathcal{O}((r_{max}/M)^2)$ . In order to find the design with the most uniform vertical field distribution at the height  $z_{obs}$ , the coil with the smallest cost value  $v$  is selected. The outcome of the procedure delivers us all required parameters of the optimized planar coil: the number of loops  $N_{opt}$ , with the corresponding radii  $R_n$ .

#### IV. EXAMPLE AND EXPERIMENTAL VERIFICATION

To illustrate the algorithm, we apply it on a representative example. The dimensions of the set-up are based on the ‘‘power transmitter design A2’’ [26] from the Qi-standard that allows free positioning. We have constructed a reference coil with the following dimensions:  $2r_w$  equals 0.83 mm (a wire that was

available for measurement verification),  $R_1 + r_w$  is 40 mm and the number of loops  $N_{ref}$  is 10. The exact numbers are not dedicated to our method, it rather allows us to focus the attention on the main, representative results. An observation height ( $z_{obs}$ ) of 5 mm and frequency of 100 kHz was chosen, which are representative for the wireless charging of low power electronic devices [27] (with the exception of biological implants where larger distances and frequencies are more typical).

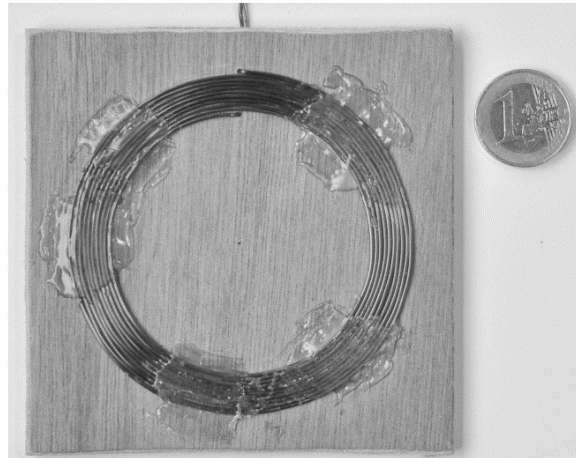


Fig. 6. The reference coil.

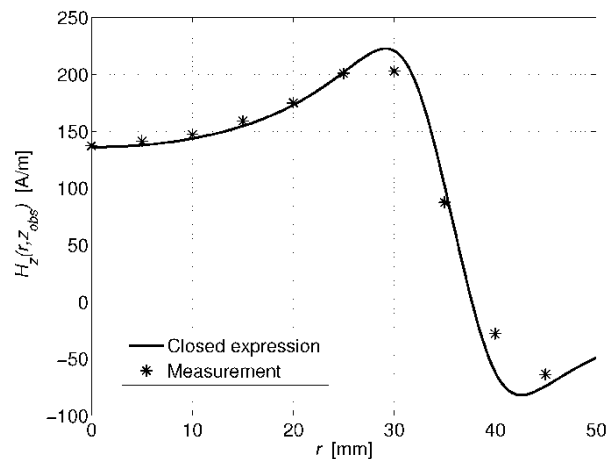


Fig. 7. Vertical magnetic field of the reference coil at the observation height  $z_{obs}$  of 5 mm.

A photograph of this coil is depicted on Fig. 6. The reflecting zones are caused by the glue to keep the loops at the same position. In order to have an estimate on the size of both coils, a one euro coin (that has a diameter of 23.25 mm) is also shown on the picture.

Application of the self inductance formula (Equation (5)) leads to a value of 13.24  $\mu\text{H}$  for  $L_{ref}$ . The vertical magnetic field distribution is shown on Fig. 7, calculated

with closed expression (3), where the excitation current equals 1 A. For the rest of this study, the results are normalized to a current of 1 A. The distance  $r_{max}$ , as defined in the description of the algorithm, corresponds to a value of 34.9 mm in this specific configuration. As mentioned above, the variation of the field above this radius  $r_{max}$  is not relevant.

On Fig. 7, experimental data is also shown. We have used the GM08 Hand-held Gaussmeter from Hirst Magnetic Instruments Ltd. to measure the vertical magnetic field. Very good correspondence is obtained between the analytical formulation and the measured data. Now that we have defined the reference coil and verified its vertical field distribution, we can start the algorithm with the goal to find a planar concentric coil with a much more uniform field distribution at  $z_{obs} = 5$  mm for  $r$  between 0 and  $r_{max}$ . In order to compare the results with our reference coil, we will require that the new planar coil has the same self inductance  $L_{ref}$  as our reference coil (i.e., 13.24  $\mu\text{H}$ ), with a maximum relative difference of  $\pm 2\%$ .

In total, we evaluated  $10^5$  different coils (interior radii random generated) with values for  $N_{ext}$  of 6 to 9, where the maximal value of  $N_{ext} + N_{int}$  was 15. For larger values of  $N_{ext} + N_{int}$ , many small loops were at the interior in order to achieve the same inductance. This increases the field at the center too much so that field homogeneity cannot be reached anymore. With this set, 7488 coils (thus about 7.5%) passed the test of being close to the original  $L_{ref}$ , with a maximal relative difference of  $\pm 2\%$ . From these 7488 coils, the smallest  $\nu$ -number was 97.1 A/m, where the  $\nu$ -number of the reference coil was 179.7 A/m.

The coil that was selected has a theoretical self inductance  $L_{opt}$  of 13.27  $\mu\text{H}$ , which deviates only 0.23% from the  $L_{ref}$  of 13.24  $\mu\text{H}$ . The optimized coil has 9 exterior loops ( $N_{ext}$ ) and 3 interior loops ( $N_{int}$ ). The values of  $R_1$  to  $R_9$  are straightforward, since they are part of the exterior loops.  $R_n$  equals in our case  $40 - 0.83/2 - (n - 1)0.83$  mm, where  $n$  runs from 1 to 9. For the three other, interior loops we have a value of 26.41 mm for  $R_{10}$ , 20.80 mm for  $R_{11}$  and 12.95 mm for  $R_{12}$ . We fabricated this optimized coil, as can be seen on Fig. 8.

As a first verification, we measured the self inductance of our coils using an Agilent 42851 Precision LCR meter. We measured a value of 12.20  $\mu\text{H}$  and 12.42  $\mu\text{H}$  for respectively the reference coil and the optimized coil. Two observations can be made. In the first place, the measured self inductance deviates from the theoretical predictions based on the formula (5). On the one hand, this theoretical prediction is an approximation. On the other hand, one also has to take into account that due to practical reasons, it is not always possible for the entire circumference to reduce the distance between any two

adjacent loops to zero. This will thus lead to some loss of self and mutual inductance. However, the most important observation, apart from the value itself, is that the two measured self inductances are close to each other, within a relative difference of 2%.

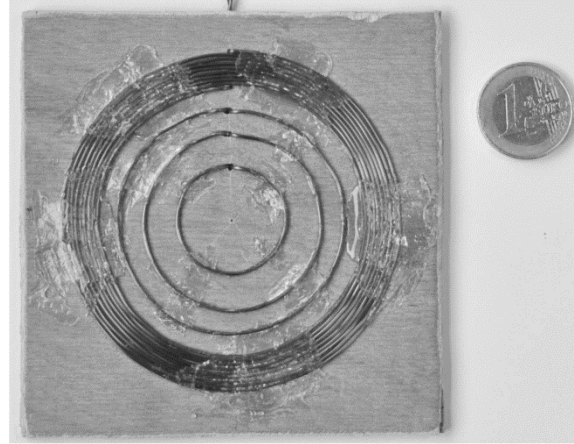


Fig. 8. The optimized coil.

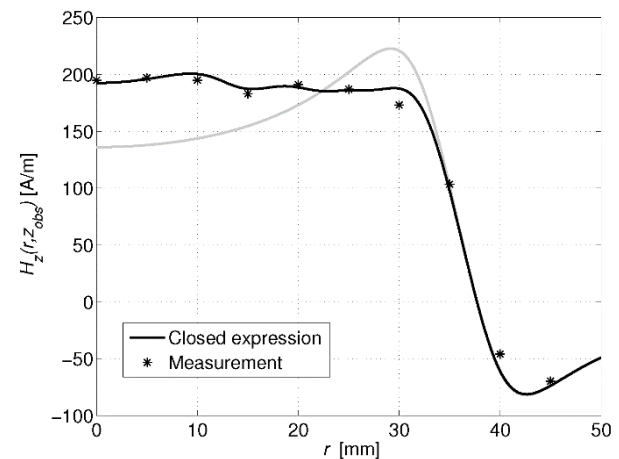


Fig. 9. Vertical magnetic field of the optimized coil at the observation height  $z_{obs}$  of 5 mm. The gray line shows the vertical magnetic field for the reference coil as comparison.

On Fig. 9, we have shown the field distribution of the coil with the minimal cost  $\nu$ , with the corresponding measurement result. One can see that the field itself, is indeed much more uniform over the  $r$ -region between 0 and  $r_{max}$ , which means that the cost  $\nu$  is a good selection criterion. A second observation, is that the field values are also high over the entire region, it is only at the outer regions of the optimized coil that the field values are smaller than the reference coil field values. On Fig. 9, a gray line with the result of the reference coil is added to facilitate the comparison. Remark that by minimizing the

cost  $v$ , we actually only targeted to obtain a uniform field, without directly imposing conditions on the value of this field. It turns out that a minimized cost also leads to a high field value. Table 1 summarizes the results for the reference and optimized coil.

Table 1: Overview of different parameters for the reference and optimized coil

	Reference Coil	Optimized Coil
$N_{ext}$	10	9
$N_{int}$	0	3
$L_{theoretical}$	13.24 $\mu\text{H}$	13.27 $\mu\text{H}$
$L_{measured}$	12.20 $\mu\text{H}$	12.42 $\mu\text{H}$
$v$	179.7 A/m	97.1 A/m

The algorithm for this example was executed in the numerical environment Matlab<sup>®</sup> on a modest computer, i.e., a MacBook Pro with an Intel Core i7 processor of 2.6 GHz with 4 GB 1600 MHz DDR3 SDRAM. It took 57.6 s for the algorithm to produce the end results for this representative example. The short algorithm time is due to the available closed-form expressions at the different steps. The time-consuming part is almost entirely attributed to the fact that many ( $10^5$ ) coils need to be evaluated.

Because the algorithm chooses the best coil (with regard to the cost function  $v$  from a large coil set), the procedure always converges and produces a useful result. The user implementing our method has to be aware of two possible issues:

- Since the algorithm selects the optimal coil from a random generated coil set, the exact same solution will not be reproduced when the program is executed more than once. However, if the user sets the number of generated coils high enough ( $10^5$  in our example), the end result will lead to as equally homogeneous magnetic field distribution, well within any practical measurement errors. Given the fast execution time of the algorithm on a modest machine, the user will not experience any significant obstacles by choosing a very large coil set.
- The novelty of this algorithm, compared to previous work, is that it imposes a by the user chosen self inductance  $L$  of the coil. This allows the user to replace an existing coil with the calculated coil, allowing a more homogeneous field with minimal impact on the rest of the network. The algorithm requires an interval for this self inductance  $L$ , for example a relative difference of 2% as chosen in this example. However, it might be beneficial to choose a larger interval, thus allowing more freedom to generate a uniform field distribution. Depending on the application, the user can, by circuital simulation,

study the influence of varying the coil inductance and adapt the interval accordingly.

## VI. CONCLUSION

In this paper, an algorithm to design a concentric, planar coil that leads to a homogeneous vertical magnetic field distribution is described. The novelty of this approach is that the procedure takes a certain target self inductance into account, so that the optimized coil can easily replace another coil without impact on the rest of the circuitry. As a consequence of the availability of closed expressions for the field distribution and self inductance, a very large set of random generated coils can be evaluated in a very short time. The result of the algorithm leads to all parameters required to realize the optimized planar coil, taken into account the radius of the wire in order to keep the structure planar. The method is applied on a representative coil. The results were verified experimentally and excellent correspondence was obtained.

## ACKNOWLEDGMENT

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# Mutual Coupling Reduction of Closely Spaced MIMO Antenna Using Frequency Selective Surface based on Metamaterials

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**Abstract** — A new structure for mutual coupling reduction, which is using the application of Frequency Selective Surface (FSS) based on metamaterials is presented in this paper. In this method, first a custom-designed unitcell is presented that provides a proper  $S_{12}$  response for the mutual coupling reduction and then, this unitcell is used in a two-element array of wideband dipoles. According to the results, this unitcell provides a maximum reduction of 15dB in the frequency response of the antenna array, while it does not have a considerable effect on the reflection coefficient and radiation pattern of the antenna. To verify the results, the antenna is fabricated and measured and there is a very good agreement between the simulation and measurement.

**Index Terms** — End-fire antenna, frequency selective surfaces, metamaterial, MIMO antenna, mutual coupling.

## I. INTRODUCTION

Today, MIMO antennas are the key elements in modern wireless communication systems, because they can minimize the interference, improve the link quality and channel capacity without the need of increasing the bandwidth. One major drawback in designing MIMO antennas is the interactions of electromagnetic wave between the adjacent elements which degrade the antenna parameters such as bandwidth and radiation pattern. To solve this issue, a lot of works have been carried out to decrease the effect of mutual coupling between antennas such as EBG, DGS, as well as metamaterials [1-5]. The abovementioned techniques have been applied for broadside radiation type. For instance, two inverted-L shaped branches and a rectangular slot with one circular end was etched on the ground plane of a folded monopole antenna, which covers different standards suitable for mobile phone applications [6]. With this proposed technique the mutual coupling is better than 15dB and 20dB in the lower and upper band respectively. However, for some applications it is necessary not to modify the ground plane so as to integrate the antenna element to RF front-end systems. Another approach to minimize the effect of

mutual coupling is to integrate arrays of split-ring resonators as a metamaterial inclusion between each adjacent element [7]. Recently, the authors in [8] have presented a compact S-shaped EBG structure to reduce the E-plane mutual coupling between two patch antennas by 15 dB at 5.25 GHz. However, the drawback of this configuration is the use of via which adds a complexity to achieve S-shaped configuration.

The authors in [9] have introduced three interdigital lines that excite the orthogonal polarization mode for reducing the mutual coupling between adjacent patch antennas. With this method the isolation corresponds to -20dB. However, using vias in the interdigital lines adds the complexity to the antenna configuration. Applying a T-shaped slot impedance transformer between two-closely spaced PIFA antennas [10] is another approach with isolation of 19.2 and 22.8 for WLAN and WiMAX frequency band respectively. Utilizing the wall loaded by a coplanar strip and defected split-ring resonator is another viable approach to reduce the mutual coupling between closely distanced patch antennas [11]. However, the presence of the wall makes the antenna structure bulky, which is useless for modern wireless communication systems.

In this paper, we propose a simple structure to suppress the effect of mutual coupling between two planar end-fire dipole antennas. The proposed antenna operates in the wide frequency bandwidth of 1.8-4.2 GHz with edge to edge distance of 2mm. To reduce the interaction between antennas, an array of modified FSS metamaterials [12] is integrated on the back side of the dipole antenna. To verify the results, a prototype of the antenna is fabricated indicating there is a good agreement between the simulation and measurement. The measured isolation between antennas is below -40dB over frequency band of 2.15-3.1 GHz, which makes this structure a good candidate for MIMO antenna applications.

## II. UNITCELL LAYOUT

Figure 1 shows the schematic view of the proposed design for the FSS metamaterial unitcell. Compared to

other structures, this unitcell does not have any vias and it is easy to fabricate. This unitcell is implemented on a Rogers RT/5880 substrate with the effective permittivity of 2.2 and the thickness of 1.575mm.

To calculate the frequency response of the unitcell, it is simulated in the HFSS software using the PEC and PMC walls along X and Z directions and two ports located in the Y direction, and the S parameter results are plotted in Fig. 2.

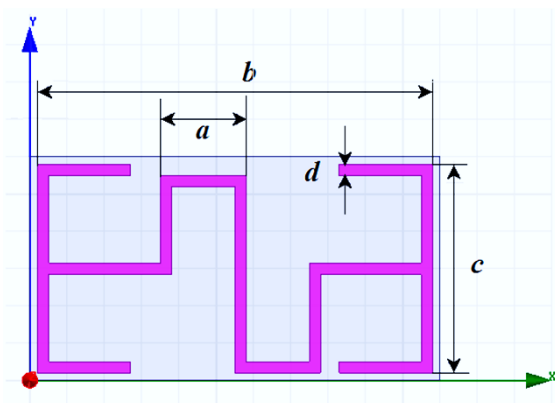


Fig. 1. Schematic view of the unitcell ( $a=2.3$ ,  $b=10.6$ ,  $c=5.6$ , and  $d=0.3$  mm).

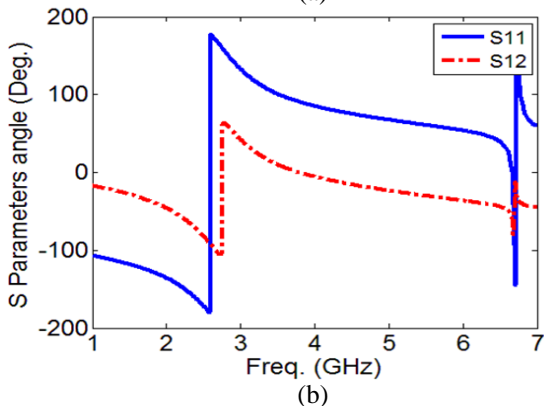
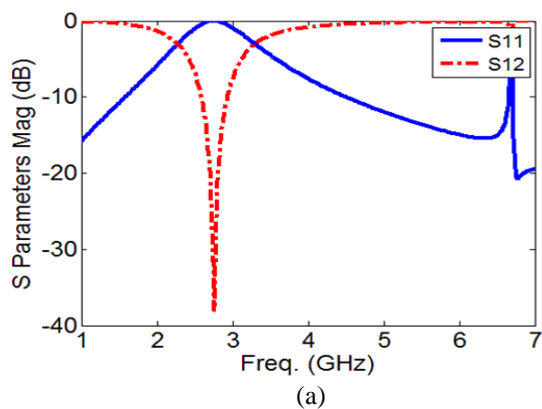


Fig. 2. Simulated S-parameters of the unitcell: (a) magnitude and (b) phase.

According to the results, there is a very deep fall-off in the  $S_{12}$  response of the unitcell from 2.28 to 3.30 GHz. As mentioned in the previous section, the final objective of this paper is to apply this unitcell on a planar antenna array structure and use its sharp fall-off as a mutual coupling reduction mechanism.

### III. ANTENNA DESIGN

Figure 3 depicts the schematic view of the presented antenna array with the unitcells printed as the mutual coupling reduction elements. The reference antenna is derived from [11] where a special balun structure is used to obtain a wideband result.

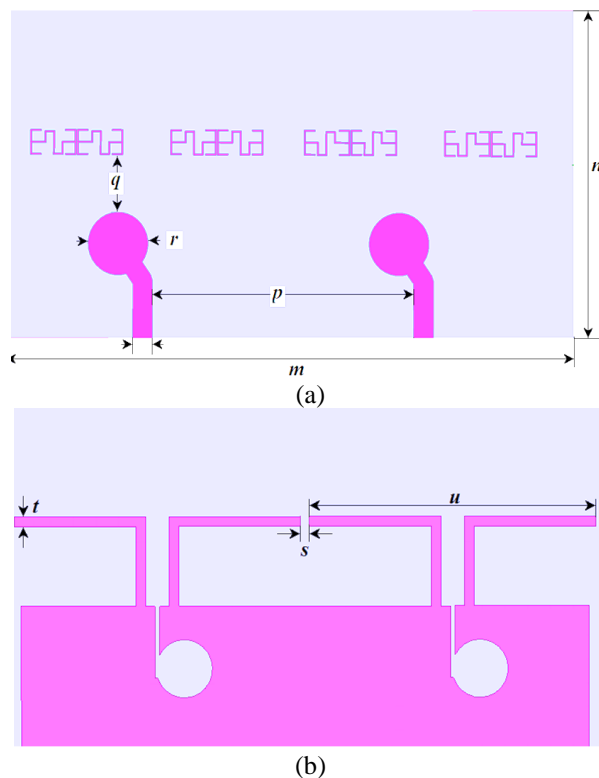


Fig. 3. Schematic view of the antenna with unitcells: (a) top, (b) bottom (dimensions in millimeters are:  $m=130$ ,  $n=71.7$ ,  $p=60.5$ ,  $q=12.7$ ,  $r=13.5$ ,  $s=1.9$ ,  $t=2$ ,  $u=63.1$ ).

This two-element antenna array is also implemented on the Rogers RT/5880 substrate with the thickness of 1.575mm. The effect of FSS metamaterial unitcells on the mutual coupling of the antenna is demonstrated in Fig. 4.

According to Fig. 4, a reduction in the  $S_{12}$  parameter of the antenna is observed from 1.7 to 3 GHz, the maximum amount of mutual coupling reduction is 15dB, which is achieved at the frequency of 2.45 GHz. The other important feature of the metamaterial inclusions in front of antenna is their negligible effect on the antenna matching as demonstrated in Fig. 5.

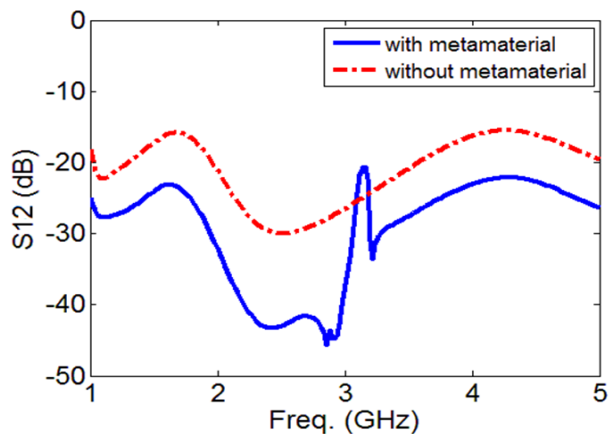


Fig. 4. Simulated  $S_{12}$  of the antenna with and without FSS metamaterial.

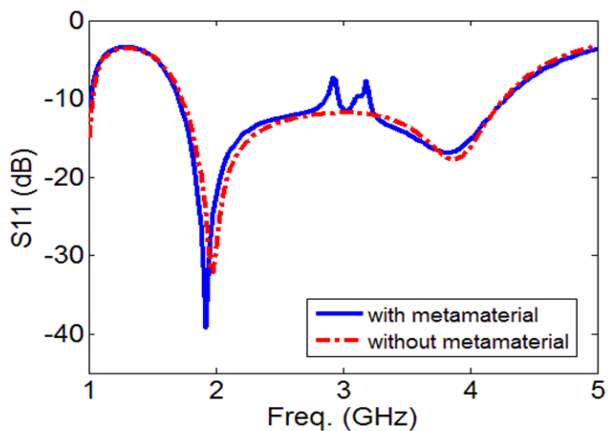


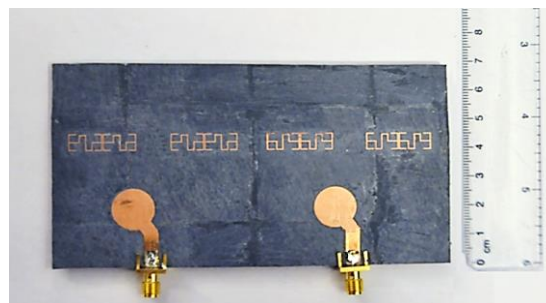
Fig. 5. Reflection coefficient of antenna with and without FSS metamaterial.

According to Fig. 5, the FSS metamaterial loading has no effect on antenna reflection coefficient and the  $S_{11}$  parameter is almost the same with and without FSS metamaterial. The antenna gain is 6dB when the unitcells are used which is the same as the antenna with no unitcells.

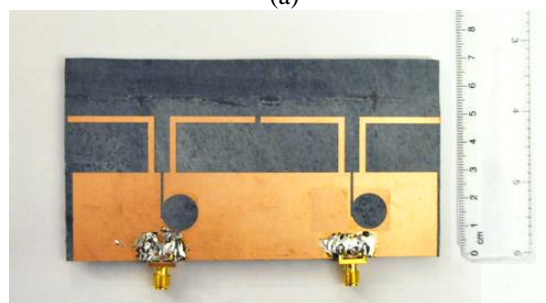
#### IV. EXPERIMENTAL RESULTS

To verify the simulation results, the antenna is fabricated and measured. A photograph of the fabricated prototype is shown in Fig. 6.

The measured reflection coefficient of the antenna with the decoupling elements is plotted in Fig. 7, which is in a very close agreement with the simulation results. Moreover, the measured radiation pattern of the fabricated prototype totally complies with the simulation in both E- and H-planes as depicted in Fig. 8.

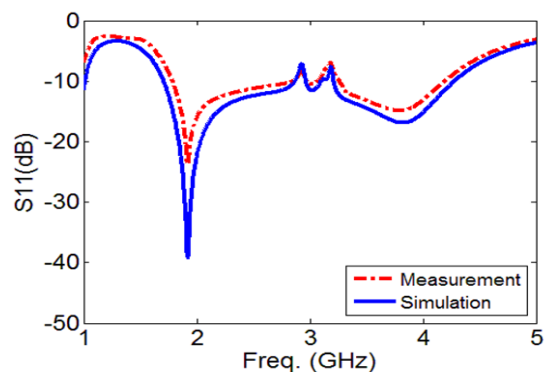


(a)

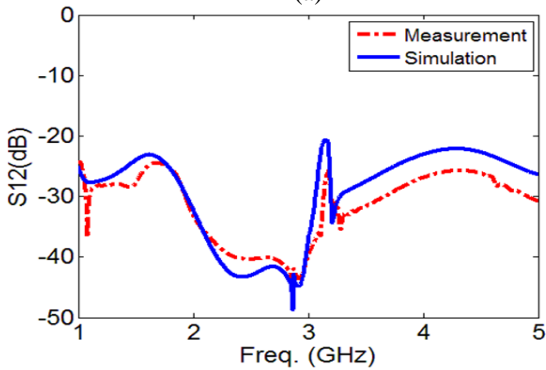


(b)

Fig. 6. The fabricated prototype: (a) top and (b) bottom.



(a)



(b)

Fig. 7. Measured S-parameters of the antenna: (a)  $S_{11}$  and (b)  $S_{12}$ .



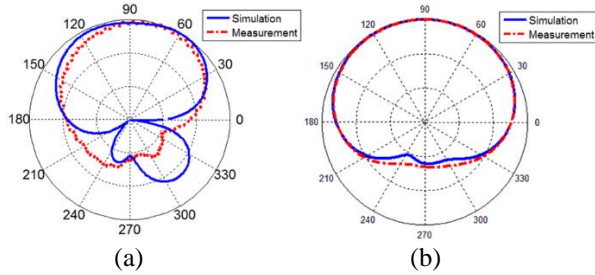


Fig. 8. Measured radiation pattern at 2.45 GHz: (a) E-plane and (b) H-plane.

## V. CONCLUSION

In this paper, a new structure for mutual coupling reduction has been introduced. In this structure, a special FSS metamaterial unitcell has been designed to produce a deep fall-off in the desired frequency range and then the presented unitcell is used in the structure of a planar two-element antenna array. According to the results, the presented technique leads to a reduction of 15dB in the  $S_{12}$  parameter of the antenna array while not affecting the antenna matching and radiation pattern. A prototype of the antenna array with the embedded unitcells is fabricated and the measurement results are very close to the simulation. The presented unitcell provides a simple, low-profile and effective method for mutual coupling reduction of antenna arrays and can be used in modern MIMO applications.

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## A Double Modal Parameter Tracking Method To Characteristic Modes Analysis

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**Abstract** — Mode tracking plays an important role in characteristic modes analysis. However, mode tracking in a proper and efficient way is still a challenging work. Based on modal linear correlation and modal stability, a double modal parameter tracking method is proposed in this paper, which is to track modes by correlating eigenvector and calculating the stability of characteristic angle simultaneously. To eliminate ambiguity case of multi-mode mapping one mode, two sorted mode with max Pearson correlation coefficient are identified same mode by utilizing saving best function. In order to verify accuracy and efficiency of the proposed method, four representative structures are analyzed. It can be observed that the proposed tracking method works better than the traditional methods.

**Index Terms** — Characteristic modes analysis, double modal parameter tracking method, modal linear correlation, modal stability.

### I. INTRODUCTION

The Theory of Characteristic Modes (TCM) was first presented by Garbacz in his doctoral dissertation [1] and then refined by Harrington in 1970s [2, 3]. TCM shows deep insight into the nature feature of an object, which is based on the method of moment (MoM). But researchers originally didn't pay much attention to TCM. As the extreme development of computer, TCM has attracted many researchers' interest now. Applying the CMA, Chen an Wang show a UAV platform integrated pattern reconfigurable antenna [4], and a HF band shipboard antenna [5], where the radiation efficiency has been improved. Using the orthogonality between different modes, many antennas with good isolation in MIMO system have been designed by researchers [6-9].

CMA is generally performed in spectrum. The same mode at different frequency samples needs to be identified firstly, which is called mode tracking. Mode

tracking is the basis of characteristic mode application. However, mode tracking in a proper and efficient way is still a challenging work. There are a few of papers involving mode tracking. The tracking methods in these papers are all only tracking one modal parameter: a) Tracking method in paper [10, 11] is based on tracking eigenvalues. b) Tracking method in paper [12-14] is based on correlating eigenvectors. c) Tracking method in paper [15] is based on correlating modal far-field pattern. Tracking methods in a) and b) are both based on one parameter deriving from generalized eigenvalue equation, and no additional calculation is needed. Therefore, these methods have high computational efficiency. However, they can't deal well with complex structures since mode swapping always exists. Tracking method in c) utilizes stability of far-field pattern to obtain better results than a) and b), but a lot of computation has to be added for calculating far-field.

In this paper, a new tracking method named as double modal parameter tracking method (DMPTM) is proposed, which is based on tracking two modal parameters at the same time.

The remainder of this paper is organized as follows. In Section 2, the theory and the formulations in characteristic modes analysis are given briefly. In Section 3, DMPTM is introduced in detail. Four numerical experiments are presented in Section 4 to show the accuracy and efficiency of the proposed method. Section 5 concludes this paper.

### II. BRIEF INTRODUCTION TO CMA

Linear combination of a set of characteristic modes can well approximately represent the solution of an electromagnetic problem. These characteristic modes correspond to the inherent properties of electromagnetic objects and have orthogonality between different modes [2]. For the sake of easy reference, the following paragraphs are a brief introduction to characteristic

modes analysis.

As presented in [2], characteristic modes can be obtained by solving the following generalized eigenvalue problem:

$$XJ_n = \lambda_n R J_n, \quad (1)$$

where  $X$  and  $R$  are imaginary part and real part of impedance matrix  $Z$ , which is derived from the well known method of moment (MoM).  $J_n$  and  $\lambda_n$  are named as eigencurrents (or eigenvectors), and eigenvalues  $\lambda_n$  represents the radiation or scattering properties of the corresponding modes. If  $\lambda_n > 0$ , the mode stores magnetic energy. If  $\lambda_n < 0$ , the mode stores electric energy. And if  $\lambda_n = 0$ , the mode is at resonance. It is notable that characteristic modes are independent of a specific source or excitation, and only rely on the shape, size, material and working frequency band of the object.

Because of its large range of eigenvalues [12], characteristic angles  $\alpha_n$  are introduced as following:

$$\alpha_n = 180^\circ - \arctan(\lambda_n). \quad (2)$$

Obviously, the range of values for  $\alpha_n$  is from  $90^\circ$  to  $270^\circ$ . If  $\alpha_n = 180^\circ$ , the corresponding modes are at resonance.

In theory, the linear combination of infinite modes is needed to describe the electromagnetic properties of the study object. Fortunately, only several mode with small eigenvalues are needed to describe the electromagnetic behavior for electrically small or moderate objects [16].

### III. DOUBLE MODAL PARAMETER TRACKING METHOD

In this section, double modal parameters tracking method is presented in detail.

#### A. Main ideas

Eigenvalues, eigenvectors and characteristic far fields are three main parameters in CMA, and they have different features: The range of values for eigenvalues is  $(-\infty, +\infty)$ , and they vary fast with frequencies and are easily affected by numerical accuracy, so eigenvalues are not suitable to be tracked. Characteristic far fields are the most stable among modal parameters, so they can be chosen to track mode. However its calculation cost is expensive and its calculation results are also affected by spatial angular step. The eigenvectors stability is between eigenvalues and characteristic far fields, and they are obtained directly through solving generalized eigenvalue equation, so mode tracking based correlating eigenvectors is a balanced choice between efficiency and accuracy.

However, mode tracking based only one modal

parameter [10-15] can work well for simple structure, but they can not deal well with complex object.

This paper proposed a double modal parameter tracking method based modal correlation and modal stability. Its main ideas are:

- 1) Tracked mode satisfying both model correlation requirement and modal stability requirement can be identified to same mode.
- 2) To eliminate ambiguity situation where multi-modes map one mode, the pair of modes with the max correlation coefficient are mapped to same mode.

#### B. Modal correlation requirement

Modal correlation requirement is referred that correlation coefficients of same modes at different frequency samples are larger than specified linear correlation threshold. In this paper, linear correlation is measured by Pearson correlation coefficient, its formulation is followed [14]:

$$r_{m,n} = \frac{\left| \frac{\sum_{i=1}^N (x_i - \bar{x})(y_i - \bar{y})}{\sqrt{\sum_{i=1}^N (x_i - \bar{x})^2} \sqrt{\sum_{i=1}^N (y_i - \bar{y})^2}} \right|}{\quad} \quad (3)$$

In formulation (3),  $x = (x_1, x_2, \dots, x_N)^T$  represents  $m^{\text{th}}$  eigenvector at frequency  $f_p$ ,  $\bar{x}$  represents the mean value of its components.  $y$  and  $\bar{y}$  represent  $n^{\text{th}}$  eigenvector and its mean value at frequency  $f_q$  respectively, and  $p \neq q$ .

The range of values for  $r_{m,n}$  is [0,1]. It represents that the two eigenvectors are linearity independent if  $r_{m,n}$  equals 0, and it represents that the two eigenvectors are linear correlation if  $r_{m,n}$  equals 1. Because eigenvectors are function of frequency, the correlation coefficients between same modes vary with frequency change, so a threshold for linear correlation is needed to be specified. According to our experience, the correlation threshold  $R_g$  is set to 0.8~0.9, the default value of  $R_g$  is 0.8. It means that sorted mode satisfied correlation requirement if their correlation coefficient is larger than  $R_g$ .

#### C. Modal stability requirement

Modal stability requirement is referred that modal parameters of same mode varying with frequencies are smaller than specified stability threshold. Stability is a universal characteristic in macroscopic physics word, so modal stability is inherent feature. In the aforementioned literatures, modal stability hasn't been paid sufficient attention, and the tracking methods in [10-15] don't make use of the modal stability.

In order to measure modal stability, characteristic

angle stability is defined as follows:

$$s = \left| \frac{\Delta\alpha}{\alpha_e} \right| = \left| \frac{\alpha_e - \alpha_c}{\alpha_e} \right|. \quad (4)$$

In formulation (4),  $\alpha_e$  represents extrapolated characteristic angle derived from known characteristic angle array, by utilizing Matlab function spline. The  $\alpha_c$  represents directly calculated characteristic angle according to formulation (2).

Tracked modes satisfy modal stability requirement if the modal stability of  $s$  is smaller than the specified threshold  $S_g$ . Because the range of values for characteristic angle is  $[90^\circ, 270^\circ]$ , so the range of values for  $s$  is also large. According to our experiences,  $S_g$  is set to 0.2~0.5 which is changed according the complexity of structure. The default value of  $S_g$  is 0.5. Figure 1 shows the schematic diagram of modal stability.

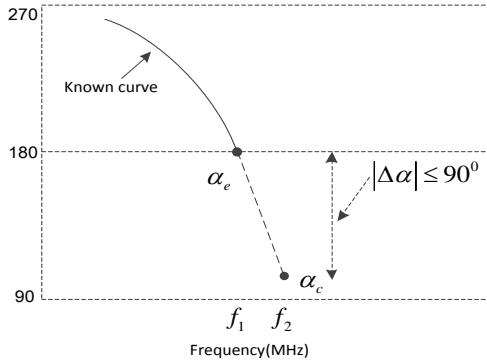


Fig. 1. Schematic diagram of modal stability.

## D. The proposed tracking method

The proposed double modal parameter tracking method is divided into three stages: pre-processing stage, mode tracking stage, and post-processing stage.

### D1. Pre-processing stage

There are several tasks in the pre-processing stage. Firstly,  $K$  eigen-pairs of each frequency, where  $M$  frequencies samples are setup at prior, are calculated and ordered in ascending order of eigenvalues. Secondly, Pearson correlation coefficients between eigenvectors of adjacent frequencies are calculated, according to formulation (3). Thirdly, Index-Table array with dimension of  $M_{rows} \times K_{columns}$ , where each column maps one mode and each row maps one frequency is initialized by its first row filled with 1,2,..., $K$  and others are filled with "NaN". The elements of the first row in Index-Table represent the  $K$  modes of the first frequency  $F_1$ , which does not need to be tracked.

The Matlab function corr is used to calculate

Pearson correlation coefficient between every eigenvector at frequency  $F_i$  and all  $K$  eigenvectors at frequency  $F_{i-1}$ , resulting in an array with  $K$  elements. After calculating correlation coefficients between all eigenvectors at adjacent frequencies, an array with dimension of  $M \times M \times K$  will be obtained.

### D2. Mode tracking stage

Mode tracking stage consists 4 tracking function: primary tracking function, rescuing function, opening new mode function, and saving best function, as is shown in Fig. 2. Mode tracking start with  $K$  eigenvectors at 2nd frequency sample  $F_2$ , then  $K$  eigenvectors at 3rd frequency sample  $F_3$ , until  $K$  eigenvector at last frequency sample  $F_M$ . The 4 tracking function will be performed at these frequencies except that primary tracking function, opening new mode function and saving best function are performed at 2nd frequency sample  $F_2$ .

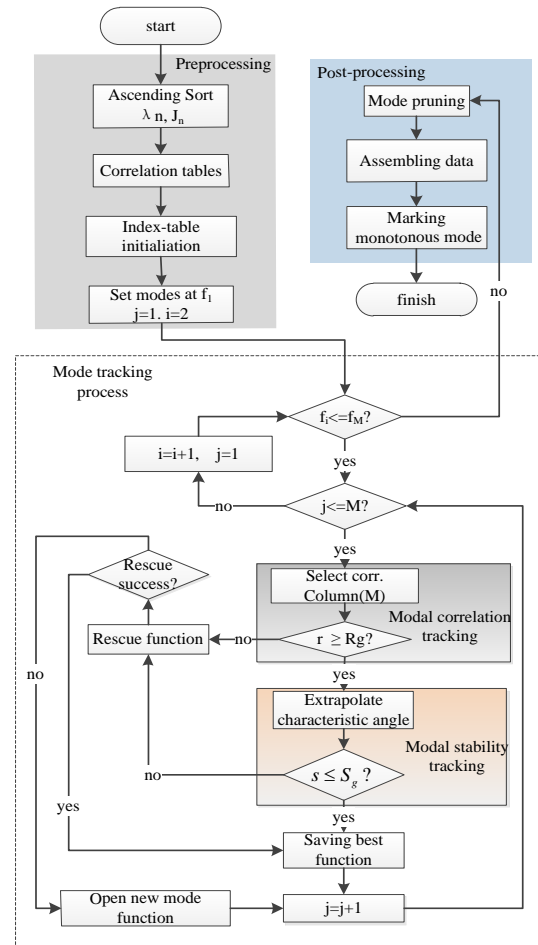


Fig. 2. Flow chart of proposed double modal parameter tracking method.

### D2.1. Primary tracking function

Firstly, one determines whether there is any correlation coefficient between the  $j^{\text{th}}$  eigenvector at  $F_i$  (called current mode) and  $K$  eigenvectors at  $F_{i-1}$  larger than  $R_g$ . If there is none of them, tracking process comes to rescuing function. And if there is any, tracking process comes to extrapolate characteristic angle. This is called modal correlation tracking. Then tracking process comes to determine whether the modal stability coefficient is smaller than  $S_g$ . If the modal stability requirement is satisfied too, tracking process comes to saving best function, and if not, tracking process comes to rescue function again. This is called modal stability tracking.

### D2.2. Rescuing mode function

Firstly, for the sake of convenience, the indicator of previous frequency sample is set as  $p = i - 2$ . Secondly, Pearson correlation coefficients between current eigenvector and  $K$  eigenvectors at previous frequency are calculated. Thirdly, modal correlation tracking and modal stability tracking are performed. If both modal correlation tracking and modal stability tracking are successful, then rescuing function success, and tracking process comes to saving best function, otherwise tracking process comes to opening new mode function.

### D2.3. Opening new mode function

The method of opening new mode function is similar to the one in paper [12]. It is to add a new column at right side of the last column of current Index-Table array. The  $i^{\text{th}}$  element of the new added column is filled with mode index, and the other elements are filled with "NaN", as is shown in Fig. 3.

### D2.4. Saving best function

If tracking process comes to saving best function, it means that both modal correlation requirement and modal stability requirement are satisfied. Without loss of generality, we suppose that  $j^{\text{th}}$  eigenvector at frequency  $F_i$  (called current mode) and  $m^{\text{th}}$  eigenvector at frequency  $F_{i-1}$  do so, and suppose that the mode index of  $m$  at frequency  $F_{i-1}$  (called previous mode) is kept at  $2^{\text{nd}}$  column in Index-Table, as is shown in Fig. 3. If the element of  $2^{\text{nd}}$  column corresponding frequency  $F_i$  is "NaN", it will be directly replaced by the mode index  $j$ . Otherwise, supposing the element is "h", it means that the  $h^{\text{th}}$  eigenvector at frequency  $F_i$  (called kept mode) has been mapped to previous mode. This scenario is known as the ambiguity case of multi-mode mapping one mode. In order to eliminate ambiguity case, one

needs to compare the correlation coefficient between current mode and previous mode with the correlation coefficient between kept mode and previous mode, then choose the pair of modes with larger correlation coefficient to map into same mode. Then the failed mode (mode h in Fig. 3.) in competing will be opened new mode if saving best function is performed in rescuing function, or will be put into rescuing function if saving best function is performed in primary tracing function.

	mode1	mode2	mode3	mode4	mode5	...	mode K	new mode
$F_i$	1	2	3	4	5	...	K	NaN
$\vdots$	$\vdots$	$\vdots$	$\vdots$	$\vdots$	$\vdots$	$\vdots$	$\vdots$	NaN
$F_{i-2}$	1	2	3	5	4	...	K	NaN
$F_{i-1}$	1	m	3	5	4	...	K	NaN
$F_i$	NaN	j	h	NaN	NaN	NaN	NaN	h
$\vdots$								NaN

Fig. 3. Example of saving best function and opening new mode function.

### D3. Post-processing

There are three works during post-processing in this paper, which are pruning Index-Table array, assembling data of eigen-pairs and marking monotonous modes. These three works are similar to the works in paper [12], it will be briefly described here.

Firstly, pruning Index-Table array means that columns with number of non-nan elements less than three will be cut off.

Secondly, assembling data of eigen-pairs means that the remaining modes in Index-Table after pruning are subsequently fulfilled with the real eigenvalues and eigenvectors, which are corresponding to mode index in Index-Table array.

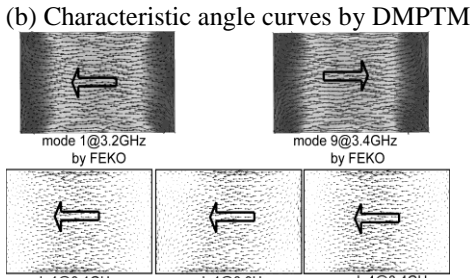
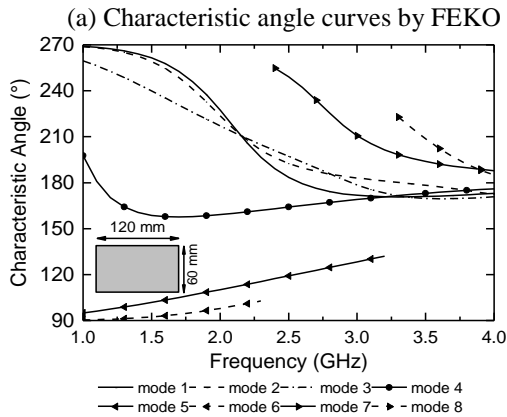
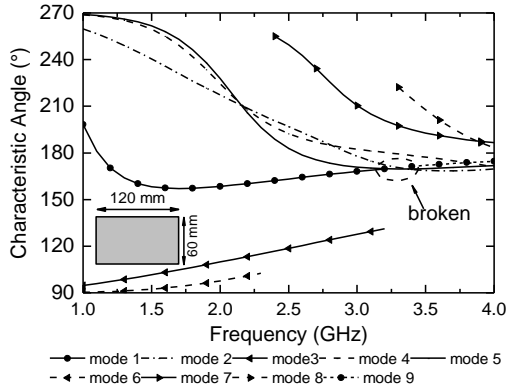
Lastly, monotonous modes means that the modes with negligible meaning within the computed frequency band. It implies that characteristic angle of monotonous modes is smaller than  $110^\circ$  or larger than  $250^\circ$  within the computed frequency band. So the last work is to find out the index of monotonous modes.

## IV. NUMERICAL RESULTS

In this section, four examples are presented to demonstrate the capability of the proposed method, the material in each example is perfect conductor, and the same triangle meshes are used for different tracking methods in each example.

**A. Mode tracking of rectangular plate**

The first example is a rectangular plate with size of 120×60 mm<sup>2</sup>, as is shown in Fig. 4. The whole structure is divided into 798 triangles, resulting in 1158 RWG basis functions. The computed frequency band is from 1 GHz to 4 GHz with 0.1 GHz frequency step, and 6 eigen-pairs are required at each frequency sample, and  $R_g$  is 0.8 and  $S_g$  is 0.5.



(c) Modal current distribution (big arrows indicate the direction of each model current)

Fig. 4. Characteristic angle curves and modal current distribution of metal rectangular plate.

Figure 4 (a) shows the tracking result by FEKO, and Fig. 4 (b) shows the tracking result by proposed method. It can be seen that the characteristic angle curves are similar except that the order of the modes is different. However, there are still obviously differences

between them:

a) The FEKO gives 9 characteristic angle curves for different modes. However the DMPTM gives 8 characteristic angle curves for different modes.

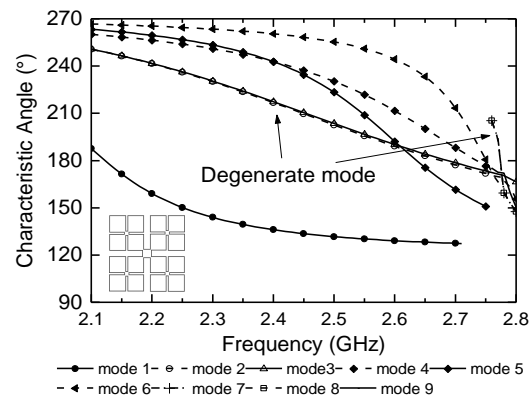
b) The characteristic angle curve of mode 1 in Fig. 4 (a) is closed at 3.2 GHz, and mode 9 is opened at 3.4 GHz. However, the characteristic angle curve of mode 4 in Fig. 4 (b), which corresponds the characteristic angle curve of mode 1 in Fig. 4 (a), is continuous in the entire analysis frequency band.

In order to check tracking results, the mode current distribution is shown in Fig. 4 (c). It is found that the current distribution of mode 1 at 3.2 GHz is the similar to that of mode 9 at 3.4 GHz, which are tracked by FEKO, so they should be the same mode. In contrast, the mode current distribution of mode 4 tracked by DMPTM is similar at 3.1 GHz, 3.3 GHz and 3.4 GHz, so they are the same mode. Therefore, the proposed method in this paper gives the correct tracking results.

**B. Mode tracking of Minkowski fractal structure**

The second example is Minkowski fractal structure with outer size of 71.3×71.3 mm<sup>2</sup>, as is shown in Fig. 5. The whole structure is divided into 664 triangles, resulting in 856 RWG basis functions. The computed frequency band is from 2.1 GHz to 2.8 GHz with 10 MHz frequency step, and 6 eigen-pairs are required at each frequency sample, and  $R_g$  is 0.8 and  $S_g$  is 0.5.

Figure 5 (a) shows the tracking result by FEKO, Fig. 5 (b) shows the tracking result by proposed method, Fig. 5 (c) shows the local enlargement of Fig. 5 (a), and Fig. 5 (d) shows the local enlargement of Fig. 5 (b). Characteristic angle cures in Fig. 5 (a) are very similar to the curves in Fig. 5 (b) except different mode order. It is can be seen from Fig. 5 (c) and Fig. 5 (d) that, there are 2 swapping modes and 2 non-continuous behaviors in Fig. 5 (a), but there are none of them in Fig. 5 (b). This is due to modal stability tracking in the double modal parameters tracking method. Therefore, the proposed tracking method in this paper gives better results than FEKO does in this example.



(a) Characteristic angle curves by FEKO

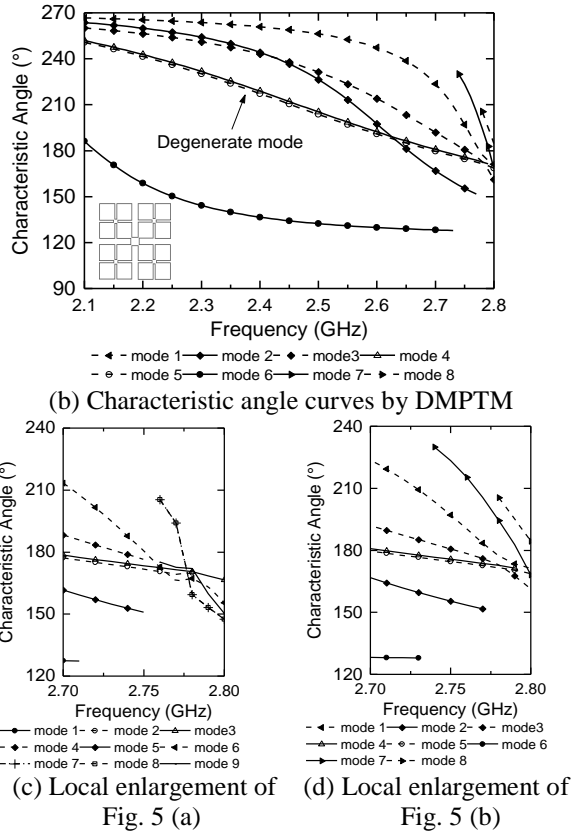


Fig. 5. Characteristic angle curves of Minkowski fractal structure.

**C. Mode tracking of open cavity structure**

The third example is open complex cavity structure. The structure is an embedded conformal omni-directional antenna without any feed [17], as is shown in Fig. 6. In the metal cavity of diameter 138.0 mm and height 45.0 mm, there are narrow circular rings with a width 7.0 mm and a wide circular ring with width 40.0 mm. The two rings are located on aperture plane of the cavity. There is one shorting post connecting narrow circular ring and bottom of the cavity at 0° and 180° respectively, and there is one shoring post connecting wide circular ring and the bottom of the cavity at 90° and 270° respectively. The four shorting posts form eight "T" junctions.

The whole structure is divided into 1289 triangles, resulting in 1834 RWG basis functions. The computed frequency band is from 0.4 GHz to 0.7 GHz with 10 MHz frequency step, and 6 eigen-pairs are required at each frequency sample, and  $R_g$  is 0.8 and  $S_g$  is 0.5.

Figure 6 (a) shows the tracking result by FEKO, Fig. 6 (b) shows the tracking result by proposed method, Fig. 6 (c) shows the local enlargement of Fig. 6 (a), and Fig. 6 (d) shows the local enlargement of Fig. 6 (b). It can be seen that characteristic angle cures in Fig. 6

(a) are also very similar to curves in Fig. 6 (b) except different mode order. As is pointed in Fig. 6 (a) and Fig. 6 (b), both of them well identify two pairs of degenerate mode. However, the curves of mode 5 and mode 6 in Fig. 6 (c) are broken at 0.67 GHz, resulting that two additional curves of mode 7 and mode 8 are opened at 0.68 GHz. So, there are 2 non-continuous behaviors in Fig. 6 (a), but there are none of them in Fig. 6 (b).

Therefore, proposed tracking method gives better results than FEKO does in this example. This is also due to the double modal parameter tracking method.

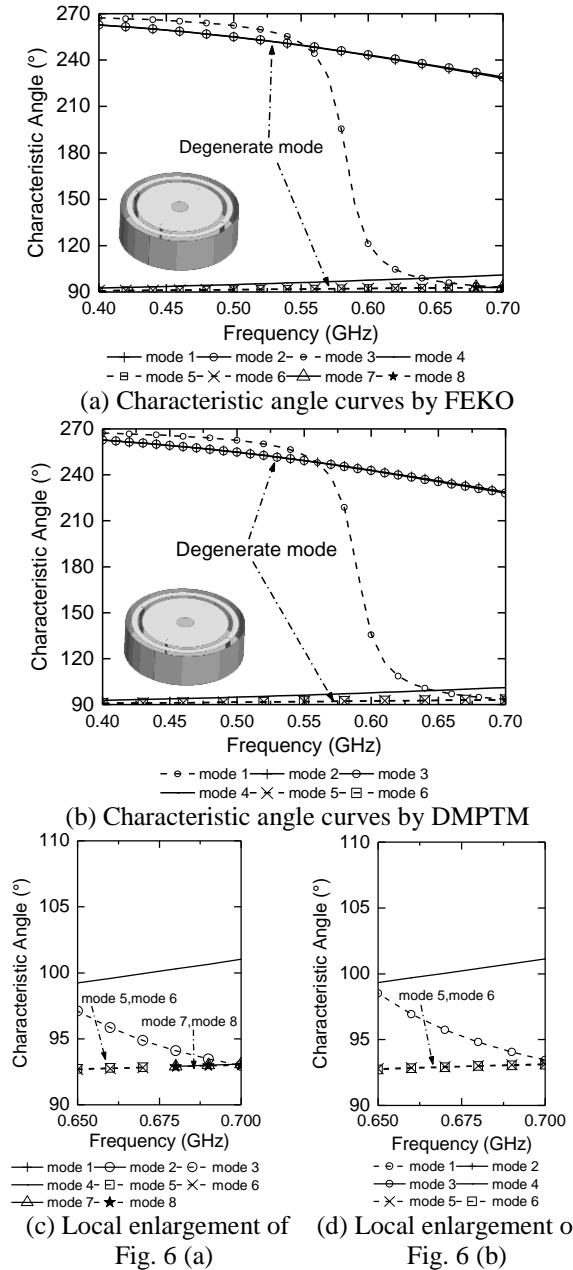


Fig. 6. Characteristic angle curves of open cavity structure.



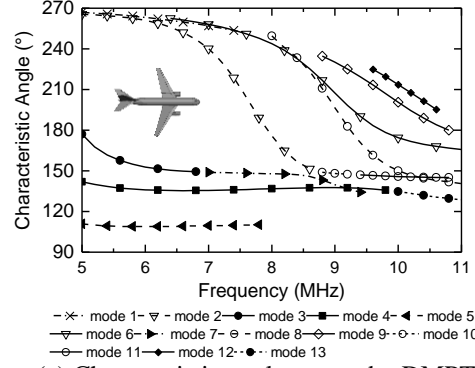
**D. Mode tracking of a plane**

The fourth example is an approximate model of commercial plane A320, which is with body length of 37 m and wingspan size of 39 meters.

The whole structure is divided into 4180 triangles, resulting in 6270 RWG basis functions. The computed frequency band is from 5 MHz to 11 MHz with 0.2 MHz frequency step, and 6 eigen-pairs are required at each frequency sample, and  $R_g$  is 0.8 and  $S_g$  is 0.2.

Figure 7 (a) shows the tracking result by FEKO, Fig. 7 (b) shows the tracking result by tracking method based on correlating eigenvector, and Fig. 7 (c) shows the tracking result by proposed method. In this example, the three figures show completely different results: there are 4 swapping modes and 2 non-continuous behaviors in Fig. 7 (a), and there are 2 swapping modes and 2 non-continuous behaviors in Fig. 7 (b), but there are none of them in Fig. 7 (c). Therefore, the proposed tracking method gives better results than FEKO and tracking method based correlating eigenvector.

Table 1 shows the detailed comparison of the proposed tracking method and FEKO, including the number of non-continuous behavior and the number of mode swapping. From Table 1, it can be seen that there is no non-continuous behavior and mode swapping for DMPTM in the four examples, but there are some for FEKO in these examples. Therefore, the proposed tracking method gives more correct tracking results than FEKO does.



(c) Characteristic angle curves by DMPTM

Fig. 7. Characteristic angle curves of A320.

Table 1: Comparison of DMPTM and FEKO

Numeric Example	Number of Mode Swapping		Number of Non-continuous Behavior	
	FEKO	DMPTM	FEKO	DMPTM
Rectangular plate	0	0	1	0
Minkowski fractal structure	2	0	2	0
Open cavity structure	0	0	2	0
A320	4	0	2	0

**V. CONCLUSION**

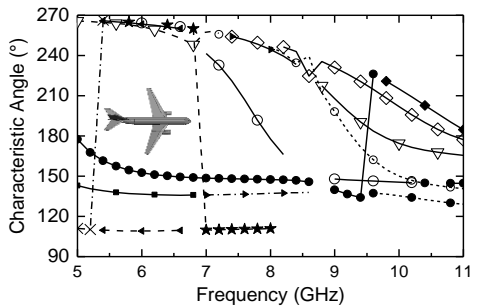
In this paper, a double modal parameter tracking method (DMQTM) is proposed, which tracks modes based correlating eigenvectors and calculating the stability of characteristic angle simultaneously. In order to eliminate ambiguity case, saving best function is introduced. To verify accuracy and efficiency of the proposed DMQTM, four examples are analyzed by FEKO, tracking method based on correlating eigenvector and proposed method respectively. The numerical results demonstrate that the proposed method efficiently gives more accurate mode tracking curves than traditional tracking method does. In order to find out the exact resonant frequency for characteristic mode, our future work is to add adaptive frequency.

**ACKNOWLEDGMENT**

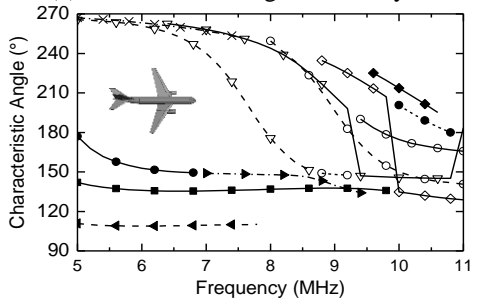
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(a) Characteristic angle curves by FEKO



(b) Characteristic angle curves by correlating eigenvector

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# Analysis of Transient Scattering from a PEC Coated with Thin Dispersive Dielectric Layer

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**Abstract** — A marching-on-in-time (MOT) based time domain thin dielectric sheet (TDS) method is proposed for analyzing the transient electromagnetic scattering from the perfect electrically conducting (PEC) body coated with thin dispersive material. The sources in the thin dielectric layer are all replaced by the current densities of PEC. A recursive convolution method is utilized to deal with the dispersive properties. The performance of accuracy and stability are investigated numerically.

**Index Terms** — Dispersive dielectric, marching-on-in-time, thin dielectric sheet, transient electromagnetic scattering.

## I. INTRODUCTION

Analysis of transient electromagnetic scattering from the thin dispersive dielectric coated perfect electrically conducting (PEC) has been paid considerable attention due to its wide range of applications. For instance, thin dispersive dielectric materials are often utilized to reduce radar cross section (RCS) in military. Many numerical methods are available for such analysis accurately. Compared with differential equation method, time domain integral equation (TDIE) method is a better way for open domain transient scattering analysis because the truncated boundary condition is needless. For the coated PEC structure, time domain volume-surface integral equation (TD-VSIE) [1]-[3] and time domain surface integral equation (TD-SIE) [4]-[5] are both valid. However, both methods will consider the unknowns of dielectric materials, and with the increasing electrical size of target, the available computational resource will be swamped quickly, which is troublesome. When the thickness of the coated dielectric layer is thin enough (in the third part, several numerical results are proposed to demonstrate that reasonable accuracy can be achieved when the thickness of the dielectric is less than 0.09 wavelength at the

highest frequency of the incident wave), the TDS approximation is a remarkable choice, all the sources in the thin dielectric coated on the closed PEC part are represents in the terms of current densities on PEC surface. The unknowns of the dielectric parts can be saved, and its efficiency has been proven [6]-[11].

In this paper, a modified MOT based approximation time domain TDS method is proposed. Compare with the conventional method presented in [11], which can only be used to compute the non-dispersive medium. The proposed method extended the applied range to analyze the transient electromagnetic scattering problems from thin dispersive material coated PEC structure. The practicality of the time domain TDS method is improved. Rao-Wilton-Glisson (RWG) basis function and time shifted Lagrange interpolant are chosen for the expansion in space and time domain. The sources in the thin dispersive material layer are all replaced by the current densities on PEC surface, so the spatial unknowns can be only considered on PEC surface. With the number of unknowns reduced markedly, the computational resource will also be saved. Here, a recursive convolution scheme [12] is introduced to handle the dispersive relationship efficiently. The investigation of performance for the proposed scheme is given in numerical results.

This paper is organized as follows. In Section II, the basic theory and formulations of time domain TDS method are presented. Several numerical results are given to demonstrate the accuracy, stability and efficiency of the proposed method in Section III. Conclusions are summarized in Section IV.

## II. THEORY AND FORMULATIONS

Consider a transient incident wave  $\mathbf{E}^{\text{inc}}(\mathbf{r}, t)$  illuminate upon an arbitrarily shaped closed PEC structure coated with an isotropic, nonmagnetic and dispersive thin dielectric layer. The geometry of the

composite structures is shown in Fig. 1.

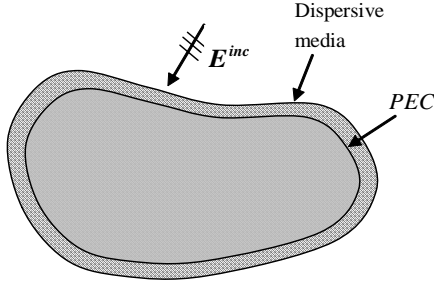


Fig. 1. Geometry of composite structure.

Enforcing the boundary condition on PEC surface  $S$ , time domain electric field integral equation can be expressed as:

$$\left[ \mathbf{E}^{\text{inc}}(\mathbf{r}, t) + \mathbf{E}^{\text{sca}}(\mathbf{r}, t) \right]_{\text{tan}} = 0 \quad \mathbf{r} \in S. \quad (1)$$

Here, the time domain scattering electric field  $\mathbf{E}^{\text{sca}}(\mathbf{r}, t)$  can be expressed as [11]:

$$\begin{aligned} \mathbf{E}^{\text{sca}}(\mathbf{r}, t) &= \mathbf{E}_{\text{pec}}^{\text{sca}}(\mathbf{r}, t) + \mathbf{E}_{\text{die}}^{\text{sca}}(\mathbf{r}, t) \\ &= -\frac{\mu_0}{4\pi} \frac{\partial}{\partial t} \int_S \frac{\mathbf{J}_s(\mathbf{r}', t-R/c)}{R} ds' - \frac{1}{4\pi\epsilon_0} \nabla \int_S \frac{\rho_s(\mathbf{r}', t-R/c)}{R} ds' \\ &\quad - \frac{\mu_0}{4\pi} \frac{\partial}{\partial t} \int_V \frac{\mathbf{J}_{\text{pol}}(\mathbf{r}', t-R/c)}{R} dv' \\ &\quad - \frac{1}{4\pi\epsilon_0} \nabla \left[ \int_S \frac{\rho_{\text{pol,down}}(\mathbf{r}', t-R/c)}{R} ds' \right. \\ &\quad \left. + \int_{S^\Delta} \frac{\rho_{\text{pol,up}}(\mathbf{r}', t-R/c)}{R} ds'^\Delta \right], \end{aligned} \quad (2)$$

where  $\mathbf{E}_{\text{pec}}^{\text{sca}}$  and  $\mathbf{E}_{\text{die}}^{\text{sca}}$  denote the scattering field generated from PEC and dispersive dielectric, respectively.  $S^\Delta$  is the upper surface of dielectric,  $V$  is the volume element of dielectric.  $\mathbf{J}_s$  and  $\rho_s$  are the surface current and charge on the conductor.  $\rho_{\text{pol}}$  and  $\mathbf{J}_{\text{pol}}$  are polarization charge and current of the dielectric layer, respectively.

The relationship between  $\mathbf{J}_s$  and  $\rho_s$  on the conductor can be deduced from the current continuity condition as:

$$\rho_s(\mathbf{r}, t) = -\int_{-\infty}^t \nabla \cdot \mathbf{J}_s(\mathbf{r}, t') dt' \quad \mathbf{r} \in S. \quad (3)$$

As shown in Fig. 2, the relationship between  $\mathbf{J}_{\text{pol}}$  and  $\rho_{\text{pol}}$  at the boundary of different materials can be expressed as:

$$\left[ \mathbf{J}_{\text{pol},1}(\mathbf{r}, t) - \mathbf{J}_{\text{pol},2}(\mathbf{r}, t) \right] \cdot \mathbf{n}(\mathbf{r}) = -\partial_t \rho_{\text{pol}}(\mathbf{r}, t). \quad (4)$$

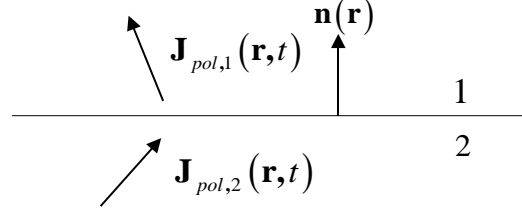


Fig. 2. The interface between two different materials.

If the part 1 is dielectric domain, and part 2 is PEC domain, (4) could change to:

$$\mathbf{J}_{\text{pol}}(\mathbf{r}, t) \cdot \mathbf{n}(\mathbf{r}) = -\partial_t \rho_{\text{pol,down}}(\mathbf{r}, t). \quad (5)$$

Similarly, if the part 1 is free space, part 2 is dielectric domain, we have that,

$$\mathbf{J}_{\text{pol}}(\mathbf{r}, t) \cdot \mathbf{n}(\mathbf{r}) = \partial_t \rho_{\text{pol,up}}(\mathbf{r}, t). \quad (6)$$

In the volume of dispersive dielectric, we have:

$$\begin{aligned} \mathbf{J}_{\text{pol}}(\mathbf{r}, t) &= \partial_t \mathbf{D}(\mathbf{r}, t) - \epsilon_0 \partial_t \mathbf{E}(\mathbf{r}, t) \\ &= \partial_t \mathbf{D}(\mathbf{r}, t) - \epsilon_0 \partial_t \left[ \gamma(t) \otimes \mathbf{D}(\mathbf{r}, t) \right], \end{aligned} \quad (7)$$

where,  $\mathbf{D}$  denotes the electric flux density.  $\gamma(t)$  is the medium susceptibility and defined in [12]. Substituting the normal boundary condition on PEC surface:

$$\mathbf{n}(\mathbf{r}) \cdot \mathbf{D}(\mathbf{r}, t) = \rho_s(\mathbf{r}, t), \quad (8)$$

to (7) and utilizing the relationship between  $\mathbf{J}_s$  and  $\rho_s$  mentioned in (3), the relationship between  $\mathbf{J}_{\text{pol}}$  and  $\mathbf{J}_s$  can be obtained as:

$$\begin{aligned} \mathbf{J}_{\text{pol}}(\mathbf{r}, t) &= -\left[ \nabla \cdot \mathbf{J}_s(\mathbf{r}, t) \right] \cdot \mathbf{n}(\mathbf{r}) \\ &\quad + \epsilon_0 \left\{ \gamma(t) \otimes \left[ \nabla \cdot \mathbf{J}_s(\mathbf{r}, t) \right] \right\} \cdot \mathbf{n}(\mathbf{r}). \end{aligned} \quad (9)$$

Utilizing the relationship of (9) to (5) and (6),  $\rho_{\text{pol}}(\mathbf{r}, t)$  is then obtained by following equations:

$$\begin{aligned} \rho_{\text{pol,down}}(\mathbf{r}, t) &= \int_{-\infty}^t \left[ \nabla \cdot \mathbf{J}_s(\mathbf{r}, t') \right] dt' \\ &\quad - \epsilon_0 \left\{ \gamma(t) \otimes \left[ \int_{-\infty}^t \nabla \cdot \mathbf{J}_s(\mathbf{r}, t') dt' \right] \right\}, \end{aligned} \quad (10)$$

$$\begin{aligned} \rho_{\text{pol,up}}(\mathbf{r}, t) &= -\int_{-\infty}^t \left[ \nabla \cdot \mathbf{J}_s(\mathbf{r}, t') \right] dt' \\ &\quad + \epsilon_0 \left\{ \gamma(t) \otimes \left[ \int_{-\infty}^t \nabla \cdot \mathbf{J}_s(\mathbf{r}, t') dt' \right] \right\}. \end{aligned} \quad (11)$$

According to the TDS theory, the sources of thin dispersive materials are all replaced by the current densities of PEC surface. Substituting (2), (9), (10) and (11) to (1), TD-EFIE can be converted to:

$$\begin{aligned}
& \left. \begin{aligned}
& \frac{\mu_0}{4\pi} \int_S \frac{\partial_t \mathbf{J}_s(\mathbf{r}, t - R/c)}{R} ds - \frac{1}{4\pi\epsilon_0} \nabla \int_S \int_{-\infty}^{t-R/c} \frac{\nabla \cdot \mathbf{J}_s(\mathbf{r}, t')}{R} dt' ds \\
& + \frac{\mu_0}{4\pi} \int_V \left\{ \frac{-\partial_t [\nabla \cdot \mathbf{J}_s(\mathbf{r}, t - R/c)]}{R} \right. \\
& \quad \left. + \frac{\epsilon_0 \partial_t \left\{ \gamma(t) \otimes [\nabla \cdot \mathbf{J}_s(\mathbf{r}, t - R/c)] \right\}}{R} \right\} \cdot \mathbf{n}(\mathbf{r}) dv \\
& + \frac{1}{4\pi\epsilon_0} \nabla \int_S \int_{-\infty}^{t-R/c} \frac{\nabla \cdot \mathbf{J}_s(\mathbf{r}, t')}{R} dt' ds \\
& - \frac{1}{4\pi} \nabla \int_S \left\{ \frac{\gamma(t) \otimes \left[ \int_{-\infty}^{t-R/c} \nabla \cdot \mathbf{J}_s(\mathbf{r}, t') dt' \right]}{R} \right\} ds \\
& - \frac{1}{4\pi\epsilon_0} \nabla \int_{S^\Delta} \int_{-\infty}^{t-R/c} \frac{\nabla \cdot \mathbf{J}_s(\mathbf{r}, t')}{R} dt' ds^\Delta \\
& + \frac{1}{4\pi} \nabla \int_{S^\Delta} \left\{ \frac{\gamma(t) \otimes \left[ \int_{-\infty}^{t-R/c} \nabla \cdot \mathbf{J}_s(\mathbf{r}, t') dt' \right]}{R} \right\} ds^\Delta
\end{aligned} \right\} \tan \\
& = \mathbf{E}_{\tan}^{inc}(\mathbf{r}, t). \tag{12}
\end{aligned}$$

Here, the recursive convolution scheme [12] is applied. In order to solve (12) conveniently, a new parameter  $\mathbf{P}(\mathbf{r}, t)$  is defined as:

$$\mathbf{P}(\mathbf{r}, t) = \gamma(t) \otimes \mathbf{J}_s(\mathbf{r}, t). \tag{13}$$

Now,  $\mathbf{P}(\mathbf{r}, t)$  and  $\mathbf{J}_s(\mathbf{r}, t)$  are expanded with spatial and temporal basis functions as follows:

$$\mathbf{P}(\mathbf{r}, t) = \sum_{n=1}^{N_s} \sum_{j=1}^{N_t} P_{j,n} T_j(t) \mathbf{f}_n^s(\mathbf{r}). \tag{14}$$

$$\mathbf{J}_s(\mathbf{r}, t) = \sum_{n=1}^{N_s} \sum_{j=1}^{N_t} J_{j,n} T_j(t) \mathbf{f}_n^s(\mathbf{r}). \tag{15}$$

Here, RWG [13] basis functions  $\mathbf{f}_n^s(\mathbf{r})$  are used as the spatial expansion basis functions. High order time shifted Lagrange interpolant  $T(t)$  [14] is used for time expansion, and the order is chosen to be 4.  $N_s$  is the number of unknowns.  $N_t$  is the number of time steps.

Substituting (13), (14), (15) into (12), and referencing the treatments of convolution operations discussed in [12], after applying the Galerkin and point testing procedures in the space and time domains, respectively, the system of equations can be written as:

$$\tilde{Z}_0^d I_i^d = V_i - \sum_{j=1}^{i-1} \tilde{Z}_j I_{i-j}^d + \sum_{j=1}^{i-1} Z_j^p I_{i-j}^p + \tilde{Z}_{i-1}, \tag{16}$$

where,

$$I_i^d = [J_{i,1}, \dots, J_{i,N_s}], \tag{17}$$

$$I_i^p = [P_{i,1}, \dots, P_{i,N_s}], \tag{18}$$

$$\tilde{I}_{i-1} = [\tilde{P}_{i-1,1}, \dots, \tilde{P}_{i-1,N_s}], \tag{19}$$

$$\tilde{Z}_{j,m} = \begin{cases} Z_{j,mn}^d - Z_{0,mn}^p \beta_{j,n}^d & 0 \leq j \leq K \\ Z_{j,mn}^d & j > K \end{cases} \quad (m, n = 1, 2, 3, \dots, N_s), \tag{20}$$

$$\begin{aligned}
Z_{i-j,mn}^d &= \frac{\mu_0}{4\pi} \int_S \mathbf{f}_m^s(\mathbf{r}) \cdot \left\{ \int_S \frac{\sum_{n=1}^{N_s} \mathbf{f}_n^s(\mathbf{r}') \partial_t T_j(i\Delta t - R/c)}{R} ds' \right\} ds \\
& - \frac{\mu_0}{4\pi} \int_S \mathbf{f}_m^s(\mathbf{r}) \cdot \left\{ \int_V \frac{\sum_{n=1}^{N_s} [\nabla' \cdot \mathbf{f}_n^s(\mathbf{r}')] \partial_t T_j(i\Delta t - R/c)}{R} \cdot \mathbf{n}(\mathbf{r}') dv' \right\} ds \\
& - \frac{1}{4\pi\epsilon_0} \int_S \mathbf{f}_m^s(\mathbf{r}) \cdot \left\{ \nabla \int_{S^\Delta} \frac{\sum_{n=1}^{N_s} [\nabla' \cdot \mathbf{f}_n^s(\mathbf{r}')] \partial_t^{-1} T_j(i\Delta t - R/c)}{R} ds'^\Delta \right\} ds
\end{aligned} \quad (m, n = 1, 2, 3, \dots, N_s), \tag{21}$$

$$\tilde{Z}_{nm} = Z_{0,mn}^p \quad (m, n = 1, 2, 3, \dots, N_s), \tag{22}$$

$$\begin{aligned}
Z_{i-j,mn}^p &= \epsilon_0 \frac{1}{4\pi\epsilon_0} \int_S \mathbf{f}_m^s(\mathbf{r}) \cdot \left\{ \nabla \int_S \frac{\sum_{n=1}^{N_s} [\nabla' \cdot \mathbf{f}_n^s(\mathbf{r}')] \partial_t^{-1} T_j(i\Delta t - R/c)}{R} ds' \right\} ds \\
& - \epsilon_0 \frac{\mu_0}{4\pi} \int_S \mathbf{f}_m^s(\mathbf{r}) \cdot \left\{ \int_V \frac{\sum_{n=1}^{N_s} [\nabla' \cdot \mathbf{f}_n^s(\mathbf{r}')] \partial_t T_j(i\Delta t - R/c)}{R} \cdot \mathbf{n}(\mathbf{r}') dv' \right\} ds \\
& - \epsilon_0 \frac{1}{4\pi\epsilon_0} \int_S \mathbf{f}_m^s(\mathbf{r}) \cdot \left\{ \nabla \int_{S^\Delta} \frac{\sum_{n=1}^{N_s} [\nabla' \cdot \mathbf{f}_n^s(\mathbf{r}')] \partial_t^{-1} T_j(i\Delta t - R/c)}{R} ds'^\Delta \right\} ds
\end{aligned} \quad (m, n = 1, 2, 3, \dots, N_s), \tag{23}$$

$$V_{i,m} = \int_S \mathbf{f}_m^s(\mathbf{r}) \cdot \mathbf{E}^{inc}(\mathbf{r}, i\Delta t) ds \quad (m = 1, 2, 3, \dots, N_s), \tag{24}$$

$K$  denotes the order of temporal basis function. The parameters  $\beta^d$  and  $\tilde{I}$  are both the intermediate variables, which are used to represent the material property and recursive convolution, respectively. The definitions of them are both expounded in [12].

### III. NUMERICAL RESULTS

In order to demonstrate the validity of the proposed scheme, three numerical examples for Debye, Drude and Lorentz materials will be presented in this section. The incident wave is the modulated Gaussian plane wave and defined as:

$$\begin{aligned}
\mathbf{E}^{inc}(\mathbf{r}, t) &= \hat{\mathbf{p}}^{inc} \cos[2\pi f_0(t - \mathbf{r} \cdot \hat{\mathbf{k}}^{inc}/c)] \\
& \quad \times \exp[-0.5(t - t_p - \mathbf{r} \cdot \hat{\mathbf{k}}^{inc}/c)^2/\sigma^2]. \tag{25}
\end{aligned}$$

Here,  $f_0$ ,  $f_{\max}$  and  $f_{bw}$  denote the center frequency, maximum frequency and bandwidth of the incident

wave, respectively.  $f_{max} = f_0 + f_{bw}/2$ .  $\hat{\mathbf{k}}^{inc}$  and  $\hat{\mathbf{p}}^{inc}$  are the incident and polarization directions.

To test the accuracy and late-time stability of the proposed method, a Debye material coated PEC sphere is considered. The radius of coated PEC sphere is 0.8m, and the thickness is 0.05m, 0.07m, 0.09m, respectively. The incident wave is modulated with  $f_0 = 165\text{MHz}$ ,  $f_{bw} = 270\text{MHz}$ ,  $\hat{\mathbf{k}}^{inc} = \hat{\mathbf{z}}$ ,  $\hat{\mathbf{p}}^{inc} = \hat{\mathbf{x}}$ ,  $\sigma = 6 / (\pi f_{bw})$ . The number of total unknowns is 3834. The static relative permittivity  $\epsilon_{r,s}$  of the coated layer is 4.2, the permittivity of the medium at infinite frequency  $\epsilon_{r,\infty}$  is 1, and the relaxation time is  $t_0 = 5 \times 10^{-9}\text{s}$ . The time step  $\Delta t$  is  $0.05Lm$ , total time is  $3000\Delta t$ . In order to verify the precision of the proposed scheme further, root mean square (RMS) error is introduced, the parameter of RMS error is defined as:

$$\text{RMS} = \sqrt{\sum_{i=1}^n |f_{c,i} - f_{r,i}|^2 / \sum_{i=1}^n |f_{r,i}|^2}, \quad (26)$$

where  $f_{c,i}$  are the values of bi-static RCS which are computed by the proposed TDS scheme, and  $f_{r,i}$  are the reference Mie series values,  $n$  is the number of observed angles. The RMS values of different thickness are shown in Fig. 3, it can be found that the accuracy of the proposed method at the most frequencies is ensured when the thickness of coated layer is no more than 0.09 wavelength of  $f_{max}$ . The bi-static RCS curves of the coated sphere with 0.07m layer are shown in Fig. 4, which agree well with the exact Mie solutions. From the magnitude of time domain current coefficient shown in Fig. 5, it can be observed that the late-time stability is also ensured.

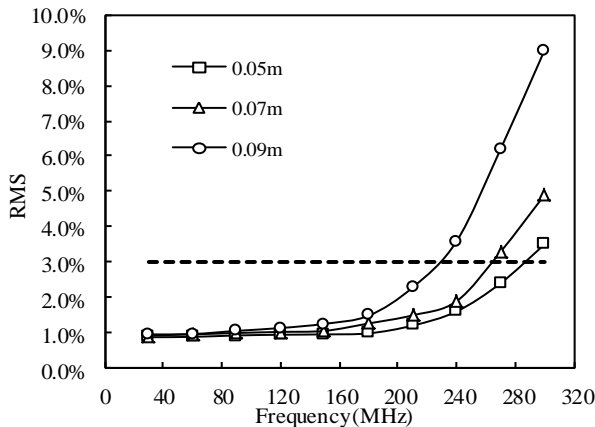


Fig. 3. RMS values of different thickness.

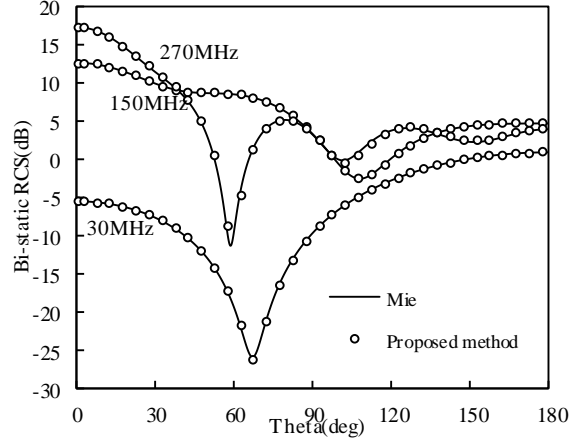


Fig. 4. Bi-static RCS ( $\phi = 0^\circ$ ) of the coated sphere structure at 30, 150, and 270 MHz.

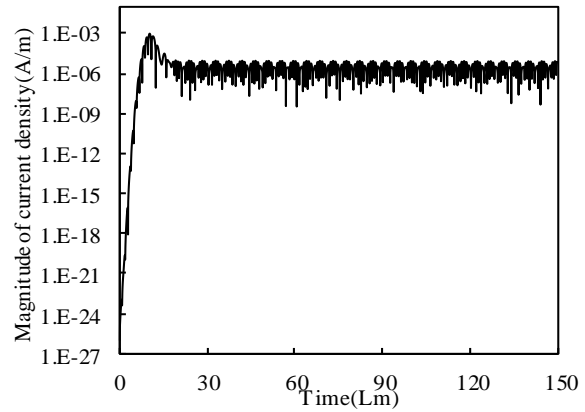


Fig. 5. Magnitude of time domain current coefficient at (0.001971m, -0.797540m, -0.062647m).

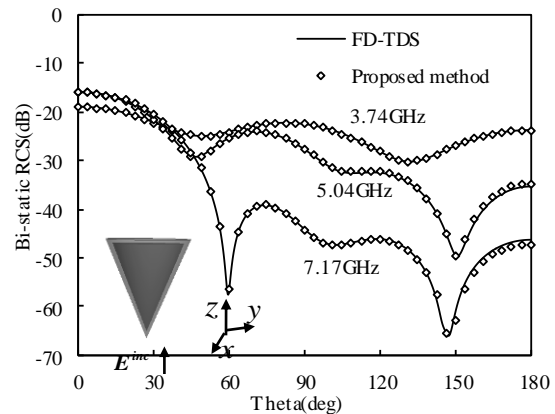


Fig. 6. Bi-static RCS ( $\phi = 0^\circ$ ) of the coated cone structure at 3.74, 5.04, and 7.17 GHz.

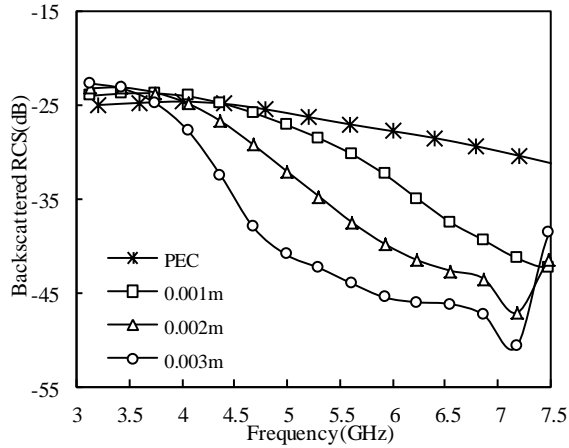


Fig. 7. The comparisons of backscattered RCS with different thickness.

Next, considering the scattering from a coated cone structure, and the coated layer is modeled as a Drude medium with  $\epsilon_{r,\infty} = 1$ , collision frequency  $\delta_p = 8 \times 10^9 s^{-1}$ , plasma frequency  $\omega_p = 8 \times 10^9 s^{-1}$ . The radius of inner PEC cone is 0.02m and the height is 0.06m. The thickness of coated dielectric layer is 0.001m, 0.002m, 0.003m, respectively. The center frequency of incident wave is 5.5 GHz, bandwidth is 5 GHz. The time step  $\Delta t$  is  $0.01875Lm$ , total time is  $500\Delta t$ . The unknown number of this object is 2376. It is a considerable reduction compared with TD-VSIE [3], which has 15806 unknowns. The bi-static RCS curves of the coated cone with 0.002m layer at different frequencies obtained after the discrete Fourier transform are shown in Fig. 6. Compared with its frequency domain counterpart, it is apparent that there is a good agreement between them. Moreover, the curves of backscattered RCS with different thickness are shown in Fig. 7, the stealth performance of coated structure is evident when compared with the PEC cone, and the performance will be more obvious with the increase of thickness.

At last, an airplane model coated with Lorentz material is considered, as shown in Fig. 8. The dimension of it is  $x \times y \times z = 2.56m \times 1.32m \times 6m$ . The coating's thickness is 0.02m. For this composite coated structure, the modeling of the coated layer is troublesome. If the TD-VSIE method is used to analyze it, the modeling of the coated layer is inevitable. However, the proposed method is used, the modeling process is simple. For analyzing the electromagnetic scattering from this composite structure by the proposed method, the time step  $\Delta t$  is  $1/20Lm$ , total time is  $600\Delta t$ . The incident wave is modulated with  $f_0 = 165MHz$ ,  $f_{bw} = 270MHz$ ,  $\hat{\mathbf{k}}^{inc} = -\hat{\mathbf{z}}$ ,  $\hat{\mathbf{p}}^{inc} = -\hat{\mathbf{x}}$ ,  $\sigma = 3/(\pi f_{bw})$  and  $t_p = 15\sigma$ .

The number of total unknowns is 4782. The damping coefficient  $\delta_p$  of the coated layer is  $1 \times 10^8 s^{-1}$ , resonant frequency  $\omega_p$  is  $1.2 \times 10^8 s^{-1}$ , static permittivity  $\epsilon_{r,s} = 2.8$ ,  $\epsilon_{r,\infty} = 1$ . Figure 9 shows the curves of bi-static RCS at different frequencies. Good agreement is obtained again when compared with corresponding frequency domain method, and the ability of analyzing complex target by the proposed method is proven.

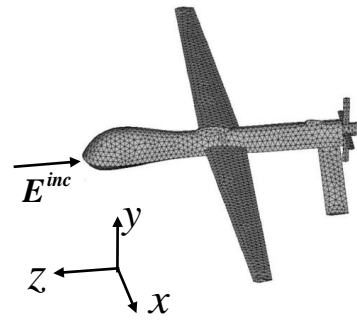
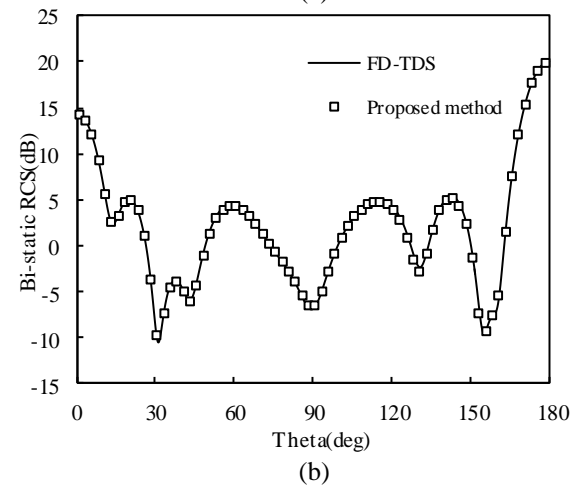
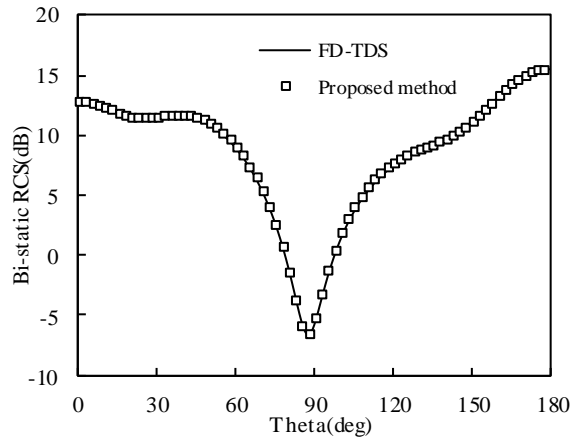


Fig. 8. Geometry of an airplane model.



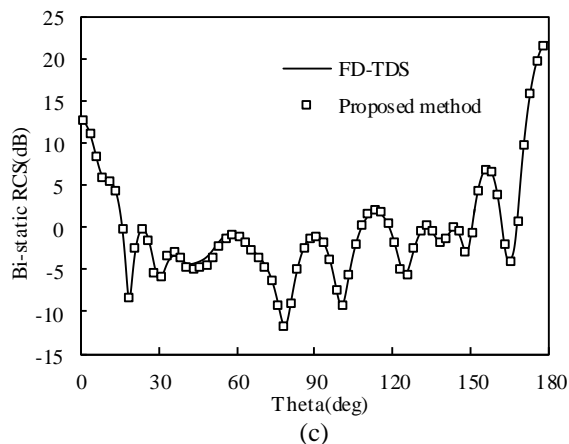


Fig. 9. (a) Bi-static RCS ( $\phi = 0^\circ$ ) of the coated airplane structure at 54 MHz, (b) bi-static RCS ( $\phi = 0^\circ$ ) of the coated airplane structure at 153 MHz, and (c) bi-static RCS ( $\phi = 0^\circ$ ) of the coated airplane structure at 252 MHz.

#### IV. CONCLUSION

In this paper, a modified MOT based TD-TDS method is proposed, it extends the practicality of conventional time domain TDS method to analyze the transient electromagnetic scattering from the conductors coated with thin dispersive dielectric. Because the sources in the thin dispersive material layer are all replaced by the current densities on PEC surface, the number of unknowns will be reduced markedly. From the numerical results, the accuracy and stability of the proposed scheme are ensured, and the ability of analyze the coated stealth structure is demonstrated.

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## RFID Humidity Sensor Tag for Low-cost Applications

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**Abstract** — This article presents a low-cost, flexible, chipless Radio Frequency Identification (RFID) tag for humidity monitoring applications. The tag exhibits moisture sensing feature within a compact geometrical dimension of 20mm x 17.6mm. The design is loaded with 12 resonators, where each resonator represents 1 bit in the frequency domain. For the designed 12-bit tag, 11 inverted C-shaped resonators are dedicated for encoding 11-bit information in their spectral signature.

An integrated meandered-shaped resonator, covered with moisture sensitive Kapton<sup>®</sup> HN film, functions as a 1-bit moisture sensor. It is deployed for monitoring relative humidity (RH) levels, simultaneously. The passive RFID tag is realized on Taconic TLX-0 and has an operational bandwidth of 2.62 GHz. Furthermore, the design is modeled and analyzed for multiple substrates. The performance of the sensor tag for various humidity levels indicates that it is a potential solution for inexpensive sensing applications.

**Index Terms** — Moisture sensor, radar cross section, relative humidity, RFID.

### I. INTRODUCTION

Radio Frequency Identification (RFID) is a wireless data collecting method deployed for automatic identification of distant targets. Billions of items are anticipated to use RFID tags for recognition in near future [1]. This technology has found its applications in identification [2], security and access control, health monitoring systems [3] and asset tracking [4].

Internet of Things (IoT) is an emerging technology which connects multiple smart objects to internet via wired or wireless networks [5]. Wireless sensor networks and RFID constitutes the IoT system. Recent

advancements in RFID technology have facilitated their deployment in IoT [6].

The existing RFID technology is categorized as chip-based RFID and chipless RFID. The chip-based RFID tag consists of a silicon chip, used for data encoding, connected to a transceiver antenna. The presence of microelectronic chip makes the system costly and complex [7]. Therefore, research efforts are made to reduce the cost and complexity of individual tags. Passive RFID technique is the ultimate solution for low-cost system design. Integrating passive sensing feature makes RFID tags attractive for various application-specific utilities [8]. The sensing parameters may include temperature, pressure, methane sensing [9], moisture, crack, and strain sensing, etc. Relative humidity (RH) monitoring is an important physical parameter for drugs and food storage, water detection in damaged walls, archives, and buildings [10]. Therefore, the market has a demand for low-cost, reliable and robust RH sensors. Re-transmission based RFID tags perform moisture monitoring by detecting the change in the phase/amplitude of the retransmitted signal. However, limited coding capacity and restricted number of terminating sensors impose the restriction for sensing of multiples nodes simultaneously [11]. To overcome these limitations, frequency-signature based RFID tags with enhanced coding capacity and miniaturized structure are presented in [12-15]. RH monitoring is achieved by the variations resulted in electromagnetic footprint of the tag. Various moisture sensitive materials, e.g., polyvinyl alcohol, gelatin, dextrin and Kapton<sup>®</sup> HN film are deployed for RH sensing. These polymers show dielectric or conductive variations with humidity levels. Humidity sensor reported in [12] uses hygroscopic, polyvinyl alcohol (PVA), which absorbs moisture and a shift in the

resonance frequency is observed. A sensor tag designed on paper substrate exhibits moisture sensitivity [13]. A stepped impedance resonator covered with Kapton® HN film polyamide for humidity monitoring in ubiquitous environment is proposed in [14]. A humidity sensor based on polyvinyl alcohol film polyamide as a sensing material is reported in [15].

In this paper, a miniaturized passive RFID tag is presented. Moisture sensing is integrated in the tag by deploying Kapton® HN film. Section II introduces the RFID tag design, followed by data encoding principal in Section III. The detailed sensing mechanism and results are presented in Section IV and Section V respectively. Section VI reflects the conclusion.

## II. GEOMETRICAL CONFIGURATION

Based on the data encoding mechanism, the frequency signature tags can be categorized into ‘re-transmission’ or ‘backscattered’ tags. The re-transmission tags consist of a transceiver antenna and resonators for ID encoding [16]. In the backscattered RFID tags, the structure is excited by an incident plane wave. Each tag generates a unique electromagnetic (EM) response when the plane wave is incident upon it. The information encoded in backscattered signal is evaluated as radar cross section (RCS). This mechanism is used in the presented sensor tag design.

The proposed RFID tag is designed on substrate Taconic TLX-0 having relative permittivity  $\epsilon_r=2.45$  and loss tangent  $\tan \delta=0.0019$  and a thickness of 0.5mm. The total dimension of the tag is 20mm x 17.6mm. Resonance at a particular frequency is achieved by etching a slot in copper sheet of thickness 0.035mm. The resonator length  $l$  corresponding to the desired frequency  $f$  is calculated by using the following set of equations:

$$c = f \lambda, \quad (1)$$

$$\lambda_g = \lambda \sqrt{\frac{2}{1 + \epsilon_r}}, \quad (2)$$

$$l = \frac{\lambda_g}{4}, \quad (3)$$

where  $\lambda$  indicates the calculated wavelength at central frequency,  $\lambda_g$  is guided wavelength,  $\epsilon_r$  is relative permittivity and  $c$  is the speed of light in free space.

Initially, a single inverted C-shaped slot is designed and optimized by using Equations (1)–(3). The same design approach is followed to achieve 11-bit encoding RFID tag by using eleven inverted C-shaped slots. The outermost slot  $R_1$  produces a resonance at the lowest frequency, whereas, the inner slot  $R_{11}$  resonates at highest frequency. For sensing purpose, a meandered S-shaped structure is incorporated inside the repetitive inverted C-shaped resonators as shown in Fig. 1. The width of individual slots is kept 0.3mm which allows ease of fabrication using the conventional PCB technology.

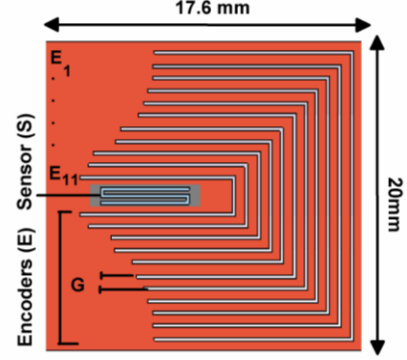


Fig. 1. The proposed RFID sensor tag.

It is observed that if two adjacent slots are placed in close vicinity, then the mutual coupling becomes significant and results in a distorted spectral signature. The effect of noise is minimized by introducing ‘guard bands’ between consecutive resonances. A spectral separation of 200 MHz is maintained between all resonances to eliminate the impact of coupling. This can be controlled by the length  $l$  and the separation  $G$  between the adjacent slots. Each slot is of different length responsible to produce resonance at particular frequency. The optimized dimensions of RFID tag are summarized in Table 1.

Table 1: Dimensions of RFID tag

Resonator	Length $l$ (mm)	Frequency $f$ (GHz)	Gap	Gaps $G$ (mm)
$R_1$	41.6	2.98	$G_1$	0.7
$R_2$	39.6	3.22	$G_2$	0.5
$R_3$	35.8	3.47	$G_3$	0.6
$R_4$	33.4	3.72	$G_4$	0.6
$R_5$	31.1	3.99	$G_5$	0.6
$R_6$	29	4.26	$G_6$	0.5
$R_7$	28	4.55	$G_7$	0.6
$R_8$	26.2	4.79	$G_8$	0.7
$R_9$	26	5.01	$G_9$	0.6
$R_{10}$	23.4	5.31	$G_{10}$	0.55
$R_{11}$	21.8	5.6	$G_{11}$	0.7
S	16.2	6.5		

## III. DATA ENCODING MECHANISM

The simulations are carried out in commercially available EM software CST MICROWAVE STUDIO® (CST MWS). The tag is excited by using horizontally polarized plane wave. RCS probes are placed at a  $d = 50\text{mm}$  to observe the backscattered encoded signal. If  $D$  is the maximum dimension of the tag and  $\lambda$  is wavelength calculated at central frequency, then far-field distance is calculated by using Equation (4):

$$d = \frac{2D^2}{\lambda}. \quad (4)$$

The tag comprising of only C-shaped encoder slots is simulated and its RCS is studied. Eleven distinct bits, distributed over a frequency range of 2.62 GHz, ranging from 2.98 GHz to 5.60 GHz, are achieved.

### A. Multiple ID combinations

The proposed 11-bit ID encoder tag can produce  $2^{11}$  different combinations using logic '1' and '0' which corresponds to presence or absence of a data bit respectively. Figure 2 shows the spectral signature of RFID tag sending all '1'.

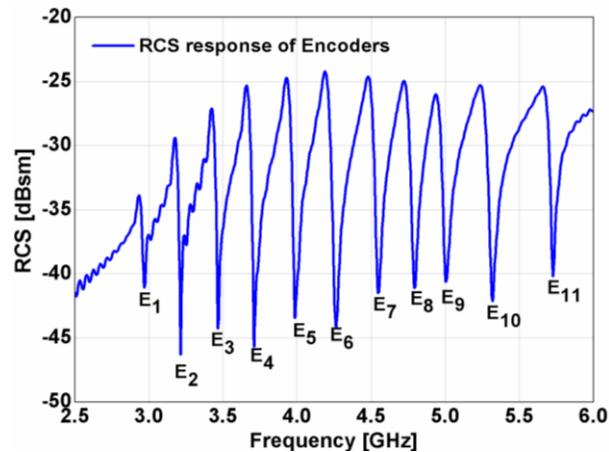


Fig. 2. The RCS response of 11-bit encoder tag.

Different data combinations can be achieved by altering the physical structure of the tag. Figure 3 shows three independent tag IDs '10111111111', '11101111111' and '11111111111'. The RCS comparison in Fig. 3 shows when a slot from a particular position is removed then only the logic state corresponding to that resonator is changed from '1' to '0'. The remaining resonators continue to resonate on the same frequencies.

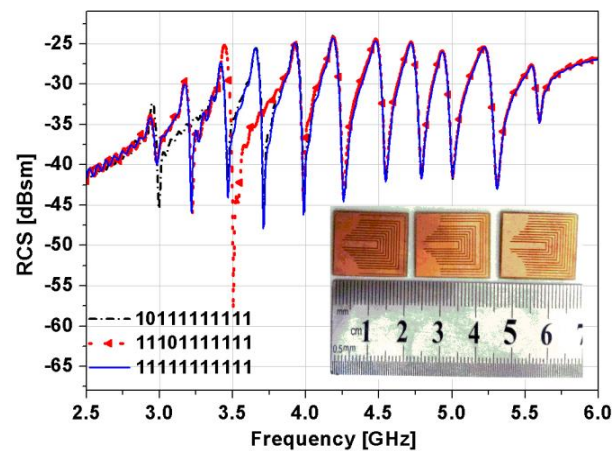


Fig. 3. Shorted tag structures and their RCS.

### B. Performance analysis of different tags

Initially, the tag is designed on Taconic TLX-0 substrate with copper as a radiating material having thickness 0.035mm. Furthermore, the tag is designed and analyzed on low-cost substrates, Rogers RT Duroid 5880 and Polyethylene Terephthalate (PET). The comparison for multiple substrates is presented in Fig. 4, without altering the geometrical parameters of the designed tag presented in Table. 1.

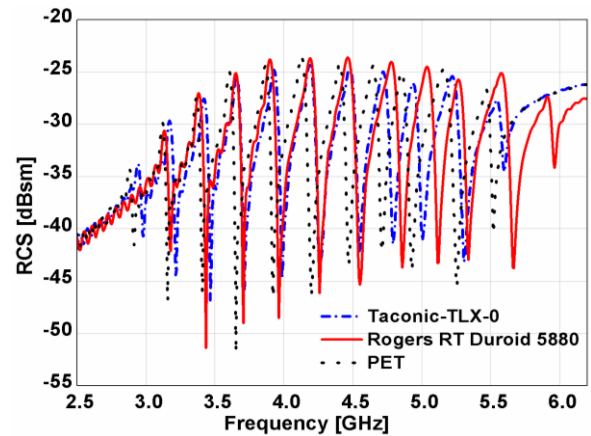


Fig. 4. RCS response for different substrates.

#### 1) Taconic TLX-0

The properties and analysis for Taconic TLX-0 substrate are presented in Section A. It can be seen in Fig. 4 that the overall bandwidth of designed tag is 2.62 GHz, where most significant bit (MSB) produces a resonance at 2.98 GHz, and least significant bit (LSB) resonates at 5.6 GHz.

#### 2) Rogers RT Duroid 5880

The same prototype is designed on Rogers RT Duroid 5880 substrate with a dielectric constant  $\epsilon_r=2.33$ , loss tangent  $\tan \delta=0.0002$  and thickness 0.254mm. The complete band of operation is 2.77 GHz, where MSB resonates at 3.19 GHz, and LSB resonates at 5.96 GHz depicted by Fig. 4.

#### 3) Polyethylene Terephthalate (PET)

The presented RFID tag is designed on a thin, low-cost substrate, PET, to incorporate the flexibility feature. The substrate has a thickness of 0.255mm, dielectric constant  $\epsilon_r=3.2$  and loss tangent  $\tan \delta=0.022$ . The analysis presented in Fig. 4 shows that the tag designed on PET requires the minimum frequency band for data encoding. In this case, the MSB is observed at 2.91 GHz, and LSB is present at 5.52 GHz.

The comparison of multiple substrates proves that the tag designed on PET substrate provides the best solution in term of flexibility and cost. Another prominent feature of the PET-based tag is the efficient

band utilization. It uses a spectrum of 2.61 GHz to data.

#### IV. HUMIDITY MONITORING

The integration of a sensing mechanism in RFID tag enables it to provide information about the state of identified object. Relative humidity  $RH$  is an important parameter for various moisture sensitive applications such as drugs storage, cold storage for a large quantity of food, etc. For this purpose Kapton<sup>®</sup> HN film with loss tangent  $\tan \delta=0.0026$ , dielectric constant  $\epsilon_r=3.5$  and thickness 0.1mm is used as a hygroscopic sensitive polymer. Its relative permittivity varies with the slight variations in humidity level. Kapton<sup>®</sup> HN film possesses linear dielectric change with humidity [17] given by Equation (5):

$$\epsilon_r = 3.05 + 0.008 \times RH. \quad (5)$$

For sensing purpose, a meandered S-shaped structure has been proposed inside the repetitive inverted C-shaped resonators. The S-shaped structure is covered with an adhesive tape of Kapton<sup>®</sup> HN dielectric film of thickness 0.1mm. The presented sensor tag has compact size and efficient band utilization.

The moisture sensitive nature of the film introduces drift in the resonance frequency of the slot, as indicated in Fig. 5. The sensing trend of the moisture tag in an ambient environment is investigated for different humidity levels. The resonance frequency is shifted to lower values with the increase in %RH. When the humidity level is incremented from 30% to 100%, the resonance is drifted from 6.7 GHz to 6.55 GHz. An overall shift of 150 MHz in the frequency spectrum is observed for the variations in %RH. However, the bits generated by the encoder slots remain unaffected by the change in the sensor resonance frequency.

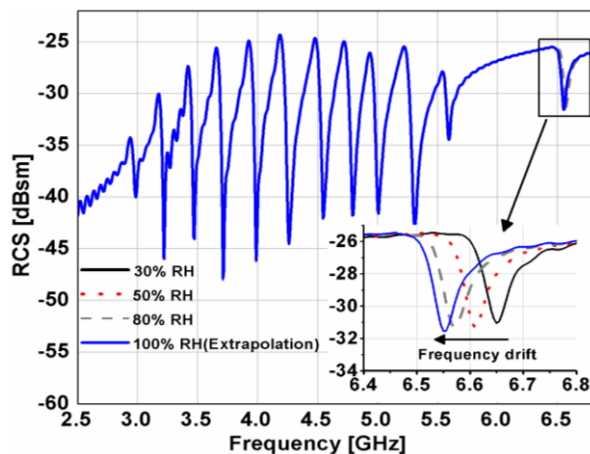


Fig. 5. RCS response of the sensor tag.

The additional sensing functionality in the designed RFID tag equips a conventional package with intelligence

to monitor environmental humidity variations and inform the customer about product condition. Henceforth, the sensor integration enables the deployment of the proposed design for various Internet of Things (IoT) and smart sensing applications. A comparison of presented research and previously proposed work [12-14] is shown in Table 2.

Table 2: Comparison of designed tag

Feature	Ref. [12]	Ref. [13]	Ref. [14]	Present Work
Substrate	Taconic TLX-0	Taconic TLX-0	Taconic TLX-0	PET, Roger 5880, Taconic TLX-0
Size (mm)	17x7.5	15x6.8	50x3.45	17.6x20
Trans. bits	4	6	4	12
Flexibility	No	No	No	Yes
Sensing	Yes	Yes	Yes	Yes

#### V. RESULTS AND DISCUSSION

The proof of concept RFID tag is fabricated on Taconic TLX-0 substrate. The experimental setup deployed, is shown in Fig. 6.

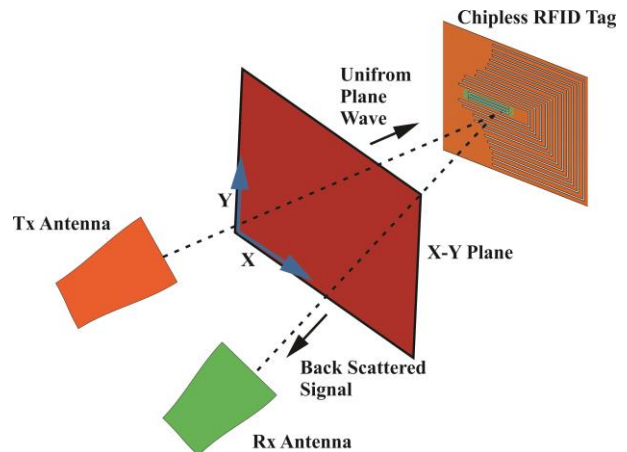


Fig. 6. Experimental setup.

It consists of two horn antennas; Tx is used to illuminate the chipless RFID tag with an incident plane wave, whereas the backscattered response observed by the antenna Rx. The two port vector network analyzer (VNA) R&S<sup>®</sup>ZVL13is used inside the climate chamber Weiss Technik WK11-180 to investigate the variations in the frequency response of the sensor tag for various humidity levels.

The experimental setup represented by the block diagram in Fig. 6 is used to calibrate the results. A comparison between the computed and measured results is shown in Fig. 7. It can be seen that the measured results are in good agreement with the simulated results.

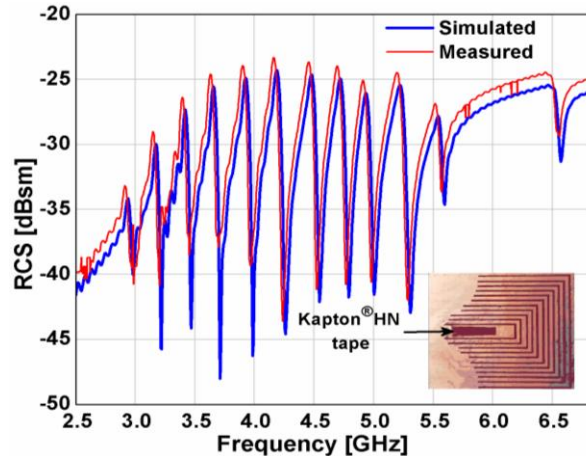


Fig. 7. Computed and measured results.

## V. CONCLUSION

A passive RFID tag for 12-bit information encoding is proposed. It provides an inexpensive solution for RH monitoring in wireless communication. The prototype is designed and analyzed on Taconic TLX-0, Rogers RT Duroid 5880 and flexible Polyethylene Terephthalate (PET). In the proposed design, the slots are arrayed so that mutual coupling can be avoided among the resonators and optimized lengths of slots yield the resonances in a squeezed band. Kapton<sup>®</sup> HN adhesive film is used as sensing material. A shift in resonant frequency is observed with the variation in humidity levels. Therefore, the proposed tag is a potential candidate for data encoding and humidity monitoring for low-cost applications.

## ACKNOWLEDGMENT

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# Modal Analysis of Different Stator Configurations to Mitigate Electromagnetically Excited Audible Noise and Vibrations of Switched Reluctance Motors

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**Abstract** — The primary objective of this paper was to investigate different stator configurations of switched reluctance motors (SRM) in order to mitigate electromagnetically excited audible noise and vibration. We analyzed natural frequencies of different SRM stator configurations by virtue of modal analysis theory. The three-dimensional numerical modeling of SRM stator geometry was performed by using finite element method. Based on the output results, we propose the solutions on how to select an appropriate stator configuration in order to increase its natural frequencies beyond the resonant operational frequency, and thus, to mitigate the resulting audible noise and vibration. The numerical and analytical results are successfully compared to the published data.

**Index Terms** — Electromagnetic force, finite element analysis, modal analysis, natural frequencies, noise and vibration, switched reluctance machines.

## I. INTRODUCTION

A quest for energy efficiency, reliability and cost effectiveness has fostered research and development of switched reluctance motors (SRM) due to the fact that they are comparable or more advantageous over their counterparts [1,2]. Numerous studies demonstrated that the SRM may be competitive electric drive system candidate in different automotive and aerospace applications where high-performance and variable-speed is required [1-5]. In addition, the SRM and the SRM drive systems are receiving considerable attention from academia and industry [6-8]. However, among the major drawbacks that prevent the SRM from being more prominent in many industrial and other applications are acoustic noise and vibrations.

Theoretical analysis predicted and experimental measurements of airborne audible acoustic noise confirmed that dominant component of the SRM acoustic noise and vibration signal is emitted from the machine's stator [9]. It has been found that pronounced acoustic

noise occurs when the frequencies of exiting stator radial magnetic forces acting on itself coincide with its natural frequencies. In particular, the level of acoustic noise and vibration strongly depends on the geometry design and material properties of the stator configuration of switched reluctance motors. Therefore, the stator configuration need to be carefully planned and designed within the pre-construction stage of the SRM machine. It has been demonstrated that the acoustic noise and vibrations can be successfully predicted with analytical and numerical mathematical models of the SRM, which were validated with experimental measurements [12-15]. For example, the issue of acoustic noise production have been addressed by varying the combination of stator-rotor poles [10]. The influence of windings and end-bells has also been studied on a commercial SRM configuration with 8/6 stator vs. rotor pole ratio [9]. However, although a large body of literature already exists on the analysis of acoustic noise and vibration of the SRM, an optimum SRM configuration is still a subject of scientific investigation [2,9-14].

The primary objective of this study is to mitigate the electromagnetically excited audible noise and vibration of SRM by altering the natural frequencies of different 3-phase SRM configurations. This study is focused on the natural frequencies for the most critical modal shapes of the small size SRM stator configurations (i.e., up to few kW [12]) by virtue of modal analysis theory. First, the natural frequencies of 3-phase SRM stator SRM configurations with different ratios between stator vs. rotor pole number are numerically analyzed based on 3D finite element modeling. Second, the impact of the SRM stator geometry modifications on its natural frequencies was studied by increasing the numbers of stator poles, by changing the thickness of the stator yoke and by adding the spacers and the end-bells to the analyzed stator geometry. The influence of the stator poles geometry modification on the natural frequencies of the resulting stator geometry was also investigated and new stator geometries are proposed. Finally, the numerical results

of 3D simulations were compared to the analytical calculations based on Jordan's law. Based on the obtained results the solutions are proposed on how to design an appropriate stator geometry in order to increase its natural frequencies beyond the resonant vibration frequency, and consequently to avoid or mitigate the undesirable acoustic noise.

## II. THEORETICAL CONSIDERATION

In our present study we focused on the magnetically excited vibration and acoustic noise of small size (i.e., up to few kW [12]) 3-phase switched reluctance motors by means of modal analysis theory. We investigated the impact of different SRM stator geometry modifications on their natural frequencies and the corresponding modal shapes that are responsible for the majority of vibration and acoustic noise. The stators of SRMs with three different ratios between the number of stator poles  $N_s$  vs. number of rotor poles are investigated:  $N_s/N_r = 6/4, 8/6$  and  $12/8$ . (Note: The number of the SRM stator poles is equal to the number of the SRM stator teeth).

The maximum noise of the switched reluctance motors is produced when the harmonics of the excitation frequencies of the magnetic radial force  $f_{exc}(n)$  coincide with the natural frequencies of the SRM stator. The excitation frequencies  $f_{exc}(n)$  in Hz for harmonic numbers  $n$  can be calculated according to Equation (1) [12]:

$$f_{exc}(n) = n \cdot f_p = \frac{n \cdot N_{rp} \cdot \omega_m}{60}, \quad (1)$$

where  $n$  is harmonic number ( $n = 1, 3, 5, \dots$ ),  $f_p$  is the fundamental frequency of the phase current in Hz,  $N_{rp}$  is the number of the machine rotor poles and the  $\omega_m$  is the rotational speed of the machine in rotations per minute.

The most critical modes of vibration of small size machines [12] are the first several modal shapes  $m$  (typically  $m = 0, m = 2, m = 4$ ), while for the medium and large size machines also higher vibrational modal shapes ( $m > 4$ ) may become also important. The contribution of the modal shapes to the acoustic noise and vibration depend on the stator/rotor configuration of the SRM. Specifically, for the 3-phase SRM with the ratio  $6/4$  and  $8/6$  the second-order cylindrical mode, also termed in this case as fundamental mode  $m = 2$  is predominant due to the fact that two opposite stator poles are excited simultaneously. Thus, the radial force acts on the stator between these two poles and further on mechanically ovalises the circumference of the stator's yoke. The 3-phase SRM with  $N_s/N_r$  ratio  $12/8$  the fourth-order mode  $m = 4$  causing double oval deformation (i.e., double ovalisation) on the stator's circumference, due to the fact that four opposite stator poles are excited simultaneously. In this case,  $m = 4$  is responsible for the majority of the emitted acoustic noise and vibration. In this study the most critical modal shapes and the corresponding natural frequencies for  $6/4, 8/6$  and  $12/8$

SRM configurations are calculated and compared by virtue of modal analysis.

The determined natural frequencies and mode shapes are solutions of an eigenvalue problem [14, 18] described by a general system of equations of motion, which assumes free vibration and ignores the dumping, Equation (2):

$$\{[H] - (\omega)^2 [M]\} \{X\} = \{0\}, \quad (2)$$

where  $[H]$  and  $[M]$  are stiffness matrix and the mass matrix, respectively. The square roots of the eigenvalues are the natural frequencies  $\omega$  (in radians/sec) of the SRM stator structure. The natural frequencies  $f$  in cycles/sec are then calculated as  $f = \omega/2\pi$ . The eigenvectors  $\{X\}$  represent the mode shapes – the shape assumed by the SRM stator structure when vibrating at frequency  $f$ .

The modal analysis of 3D SRM stator structures was performed by numerically solving the Eq. (2) using a commercial finite element based software package Autodesk Inventor Professional 2016 and Matlab R2013a. All numerical simulations were run on a computer platform with 2.7 GHz of CPU speed and 3GB of RAM. The accuracy of the numerical results was controlled by the selection of number of finite elements within the model. The final density of finite element mesh was created by increasing the number of elements until the results of the calculated natural frequencies changed less than 0.5%, and thus the numerical error was negligible. However, it should be noted that the models with high finite element number (high quality of finite element mesh) require significant computational time and computer resources.

Within the final part of the study the natural frequencies of the studied baseline SRM stator geometries were also analytically calculated by employing Jordan's law [14,17] in order to find out wheatear it can be used as a more rapid but approximate solution for calculation of natural frequencies of SRM. The analytical results were then compared to the numerically calculated natural frequencies of the SRM stators in 3D (the geometry data are taken from Table 1 with M270-35A material properties). According to the Jordan's law [14,17] the natural frequency for zero modal shape ( $m = 0$ ) of the stator can be calculated with Equation (3) [14,17]:

$$f_0 = \frac{1}{2 \cdot \pi} \cdot \sqrt{\frac{E}{\rho \cdot R_c^2 \cdot \Delta_m}}, \quad (3)$$

where  $E$  is Young's modulus of the stator core material,  $\rho$  is the density of the material composing the stator core,  $R_c$  is the mean stator yoke radius, and the  $\Delta_m$  is the corrective factor [14] that accounts for the effect of the stator poles' mass on the stator natural frequencies. The corrective factor  $\Delta_m$  can be calculated by Equation (4) [14]:

$$\Delta_m = \frac{m_{sy} + N_s \cdot m_{sp}}{m_{sy}}, \quad (4)$$



where the  $m_{sy}$  and  $m_{sp}$  stand for the stator yoke mass and the stator pole mass, respectively, whereas the  $N_s$  is the number of the stator poles.

According to the Jordan's law [14,17], the natural frequencies for modal shapes  $m \geq 2$  are expressed with Equation (5):

$$f_m = \frac{f_0 \cdot w_{sy} \cdot m \cdot (m^2 - 1)}{2 \cdot \sqrt{3} \cdot R_c \cdot \sqrt{m^2 + 1}}, \quad (5)$$

where  $f_0$  is natural frequency for zero modal shape,  $w_{sy}$  is the thickness of the stator yoke,  $m$  is the number of the modal shape ( $m \geq 2$ ), and  $R_c$  is the mean stator radius without considering the stator poles. The Equations (3) and (5) are derived for 2D structures that represent an approximation of the electrical machine stator with a ring (the stator teeth (i.e., poles) are taken into account by introducing the corrective factor  $\Delta_m$  (Equation (4)).

### III. RESULTS AND DISCUSSION

#### A. Verification of 3D numerical modeling

Within the first part of the study a preliminary 3D model numerical analysis was done based on the reference SRM configuration geometry and its material properties taken from the literature [9,15]. The stator configuration (the SRM with  $N_s/N_r = 8/6$  and the ribbed frame (i.e., stator housing)) of the reference SRM model [9] is based on the modal analysis theory and on experimental validation performed using a shaker-accelerometer system. In order to verify the accuracy of our 3D numerical results, we compared the results of our numerical simulations to the results from the reference study [9]. The 3D model of the SRM stator ( $N_s = 8$ ) with the ribbed frame geometry used as a reference model for numerical results verification (the model geometry was built according to the dimensions of the model developed in [9]) is shown in Fig. 1 (a). The dimensions are given in millimeters. The 2D view in the central XY cross section of the SRM stator only is shown in Fig. 1 (b). The geometrical parameters in the Fig. 1 (b), the stator yoke thickness  $w_{sy}$ , the height of the stator pole  $h_p$  (i.e., the stator tooth), the width of the stator pole  $w_{sp}$ , the pole arc of stator  $\beta$ , stator outer radius  $R_{so}$ , the yoke inner radius  $R_{sy}$ , the stator bore radius  $R_{si}$  and the length of stator  $L$ . According to [9], the values of the geometrical parameters are  $w_{sy} = 11.4$  mm,  $h_p = 30$  mm,  $w_{sp} = 16.9$  mm,  $\beta = 20.2^\circ$ ,  $R_{so} = 89.8$  mm,  $R_{sy} = 78.4$  mm,  $R_{si} = 48.18$  mm, and  $L = 151$  mm. According to [9], the stator was modelled as a ferromagnetic material with the Young's elasticity modulus  $E = 2.07 \cdot 10^{11}$  N/m<sup>2</sup>, while the material for the stator frame is cast iron with  $E = 1.65 \cdot 10^{11}$  N/m<sup>2</sup>. The mass density of both materials is 7800 kg/m<sup>3</sup>.

The calculated natural frequencies for  $m = 2$  and  $m = 4$  obtained in this study were 1331 Hz and 4237 Hz, respectively (as shown in Figs. 2 (a) and 2 (b)). These two frequencies differed from the calculated and measured natural frequencies in [9] by 1.3% and 2.8% for the modal

shape 2 and 4, respectively. Thus, a good agreement was obtained between our results and the results previously obtained from [9]. This small difference can be attributed to the difference in geometry of the modeled geometry dimensions that were not available to us from [9]. Namely, the geometry of ribs at the bottom part of the stator frame (present in [9]) was omitted in the model developed in this study. Based on this comparison the preliminary numerical models built in this study were verified. Based on compared results we concluded that the preliminary mode can be successfully used and extended for further analysis performed in this study.

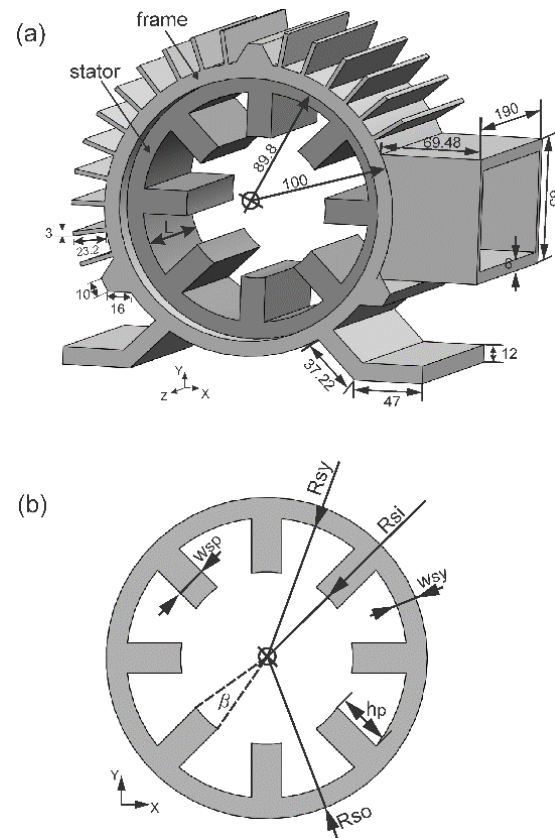


Fig. 1. (a) The 3D model of the SRM stator ( $N_s = 8$ ) with the ribbed frame geometry used as a reference model for numerical results verification (the model geometry was built according to the dimensions of the model developed in [9]; the dimensions are given in millimeters), and (b) the 2D view in XY cross section of the SRM model with the geometrical parameters.

The numerical results are shown in Fig. 2. The results (i.e., the deflection profiles at the calculated natural frequencies) are displayed in central XY cross-section plane of the 3D finite element models. The grayscale color bar shows relative movement/displacement profile from the minimum (light gray) to maximum (black) value as a result of numerical modal analysis performed

in Autodesk Inventor software we used in our study. The white color in figures represents the stator area where no movement/displacement occurs. The color chart shows relative movement/displacement values based on which the modal shapes are formed. Therefore, the units are not applicable (they have no actual physical value), since the mode shapes values are relative. (Note: This is also valid for the results shown in the Fig. 6 and Fig. 8.)

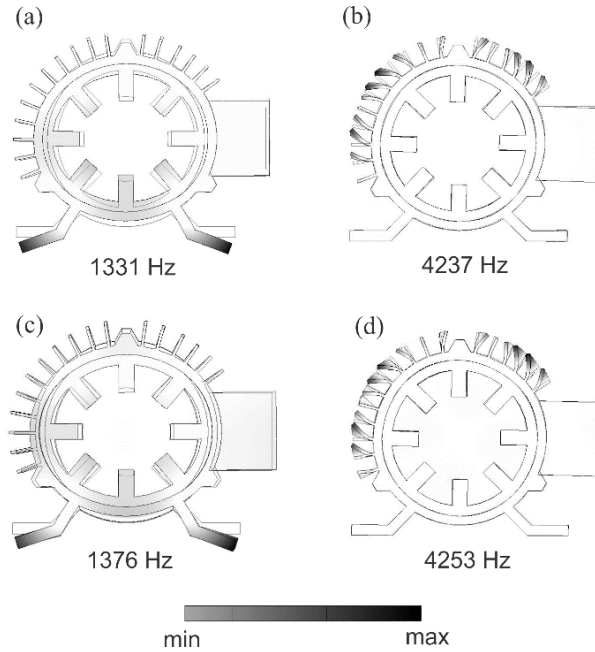


Fig. 2. The calculated deflection profiles and the corresponding frequencies of SRM stators with ribbed frame without end-bells: (a) 1331 Hz and (b) 4237 Hz; and with end-bells: (c) 1376 Hz and (d) 4523 Hz.

The second modal shape ( $f = 1331$  Hz) of the modeled assembly of the SRM stator with ribbed frame is shown in Fig. 2 (a). From the Fig. 2 (a) it can be seen that at lower natural frequency the stator deforms according to the second modal shape, while the stator frame remains still. It is interesting to note that at higher natural frequencies ( $f = 4237$  Hz) the stator remains still while the mechanical vibration and deformation of the stator frame ribs considerably increases, as shown in Fig. 2 (b). In addition, in order to increase the stiffness of the stator geometry the end-bell geometry (illustrated Fig. 5 (b)) was also added to the baseline stator geometry and the natural frequencies of the whole assembly were calculated. The calculated and visualized results for the stator assembly with one end-bell are displayed in Figs. 2 (c) and 2 (d). The calculated numerical results shown in Figs. 2 (c) and 2 (d) indicate that if the end-bells are added to the analyzed assembly the vibration of the whole assembly is somewhat constrained, resulting in increase of the natural frequency as compared to the

stator assembly without the end-bell. Due to the presence of end-bell the SRM assembly is expected to emit the vibration of higher frequencies compared to the stator assembly without the end-bell. Interestingly, at higher frequencies the ribs of the stator frame starts to vibrate, and thus the mechanical noise and vibration is emitted predominantly due to the frame.

**B. Influence of SRM stator geometry modification on its natural frequencies**

The SRM stator geometries (with  $N_s = 6, 8$  and  $12$ ) given in Fig. 3 served as reference/baseline models for all further geometry modifications and corresponding natural frequency calculations.

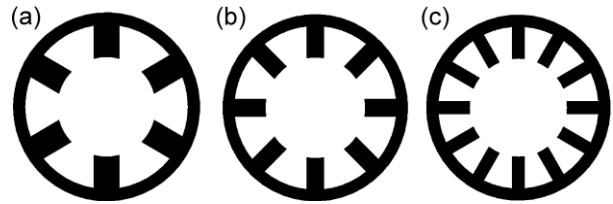


Fig. 3. The XY view of the SRM stators with: (a)  $N_s = 6$ , (b)  $N_s = 8$ , and (c)  $N_s = 12$ .

The XY view of the SRM stators geometry with  $N_s = 6, 8$  and  $12$  is given in Fig. 3 (a), Fig. 3 (b) and Fig. 3 (c), respectively.

Table 1: Dimensions of the analyzed SRM stators

$N_s$	6	8	12
$w_{sp}$	25.82 mm	15.64 mm	13.05 mm
$\beta$	30°	18°	15°
$R_{so}$	95 mm		
$R_{sy}$	82.5 mm		
$L$	114 mm		
$w_{sy}$	12.5 mm		
$R_{si}$	50 mm		
$h_p$	32 mm		

The width of the stator pole and the stator pole arc were adjusted to each of the stator geometry according to the design recommendations for the SRM stator sizing [15]. The stator was modeled as a solid/bulk material without taking into account the laminations. The used material for the stator yoke is M270-35A. The dimensions of the SRM stators geometries analyzed in this study are listed in Table 1.

Within this part of the study, the influence of modification of the stator yoke thickness  $w_{sy}$  and the height of the stator poles (i.e., stator teeth)  $h_p$  on the stator natural frequencies was numerically investigated. The parameterization of the SRM stator geometry was carried out by modifying the stator yoke thickness and height of the stator pole with respect to the dimension

of the baseline model dimensions  $w_{sy} = 12.5$  mm and  $h_p = 32$  mm by increments of 0.5 mm. In this way the functional dependency of the studied SRM natural frequencies  $f$  [Hz] on yoke thickness  $f(w_{sy})$  and pole height  $f(h_p)$  was obtained, as shown in Fig. 4. The numerical simulations show that the increase in height of the stator's pole  $h_p$  results in decrease of natural frequency for all studied SRM stator configurations as shown in Fig. 4 (a) and Fig. 4 (b). On the other hand, the Figs. 4 (a) and 4 (b) show that higher number of stator poles (i.e., 6/4 vs. 8/6) yields higher natural frequency of the SRM stator configuration. As for the parameterization of the stator yoke thickness, the SRM stator natural frequencies increased with the increase of the stator yoke thickness Fig. 4 (c) and Fig. 4 (d). This is in agreement with previous findings which demonstrated that undesirable vibrations of the SRM machines can be reduced by employing relatively thick stator yokes to minimize deflection, which in turn increases the mechanical stiffness of the machine [17].

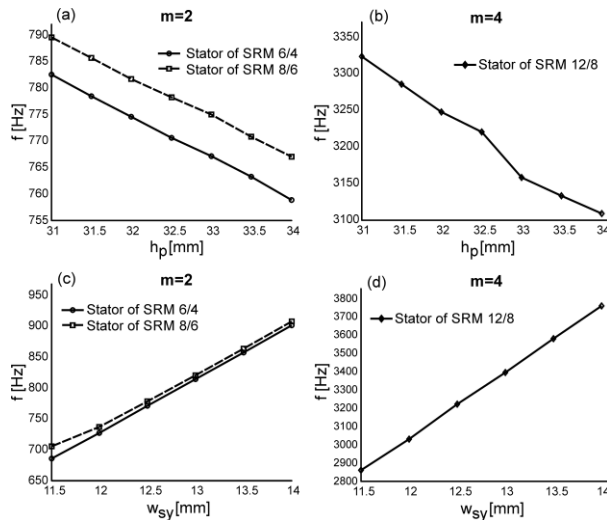


Fig. 4. The calculated relationships: (a)  $f(h_p)$  and (b)  $f(w_{sy})$  for the SRM designs with  $N_s = 6$  and  $N_s = 8$ , and (c)  $f(h_p)$  and (d)  $f(w_{sy})$  for the SRM design with  $N_s = 12$ .

Furthermore, the increase of natural frequencies of SRM configuration by adding the aluminum spacers between the stator poles and the end-bells is also numerically calculated. The geometry and placement of the spacers and the end bells added to the reference/baseline SRM geometries (from Fig. 3) are shown in Fig. 5 (a) and Fig. 5 (b), respectively.

The number of stator poles in SRM designs depends on the number of the excitation voltage phases used (i.e., three phases are needed for the 6/4 SRM design, four phases are needed for the 8/6 SRM; the 6/4 design is two times repeated in the 12/8 design) [15]. The thickness of the spacers inserted between the stator poles depends on

the available space within the stators' slots (as well as on the properties of the windings, fill factor, current density and  $B(H)$  characteristic of the stator material). In this study the width of the spacers is 5 mm, while the width of the end-bell is 10 mm, Fig. 5. The number of spacers is equal to number of stator poles. Since the end-bells represent the bearing housing, their dimensions thus need to be calculated according to the bearing dimensions. When calculating the end-bell dimensions it is also important to achieve a high enough mechanical strength of the SRM housing in order to efficiently support the mechanical loadings of the shaft. In order to precisely calculate the natural frequencies a 3D numerical modeling need to be used by taking into account the material and geometrical properties of the SRM stator structure.

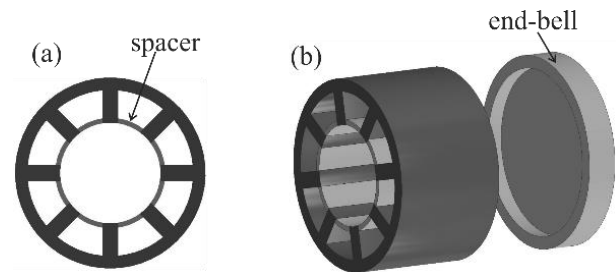


Fig. 5. Geometry and placement of: (a) the spacers between poles, and (b) the spacers between poles and the end-bell added to the baseline geometry of the SRM stators. The width of the spacers is 5 mm. The width of the end-bell is 10 mm.

The results of 3D numerical calculations of stator motion profiles in ZX cross section plane at circumferential mode shapes  $m = 2$  (for SRMs 6/4 and 8/6) and  $m = 4$  (for SRM 12/8) and the corresponding natural frequencies are shown in Fig. 6. The results are displayed in increasing order of the calculated natural frequencies. The calculated natural frequencies and the deflection profiles of the baseline SRM stators with  $N_s = 6$ ,  $N_s = 8$  and  $N_s = 12$  are shown Figs. 6 (a), 6 (b) and 6 (c). The numerical results show the natural frequencies can be increased by adding the end-bells to the stators as shown in Figs. 6 (d), 6 (e) and 6 (f).

The natural frequencies can be further increased by adding the spacers between the stator poles (Figs. 6 (g), 6 (h) and 6 (i)). The integration of the spacers and end-bell together yields the highest increase in natural frequencies as shown in Figs. 6 (j), 6 (k) and 6 (l). The increase in natural frequencies of the stators due to different stator modifications with respect to the natural frequencies of the reference stator models with  $N_s = 6$ ,  $N_s = 8$  and  $N_s = 12$  is evaluated and compared in Table 2. The natural frequency ( $f$ ) of the modified stators is divided by the natural frequency of the reference stator model ( $f_0$ ) (i.e., factor  $f/f_0$  in Table 2). The factor  $f/f_0$  is

calculated for the stators with one end-bell (no spacers), with spacers (and no end-end-bells), with one end-bell and spacers, two end-bells (and no spacers) and with two end-bells and spacers. As shown in Table 2, the integration of two end-bells and spacers yields the highest increase in natural frequencies (i.e., by factors 3.97, 4.15 and 2.44).

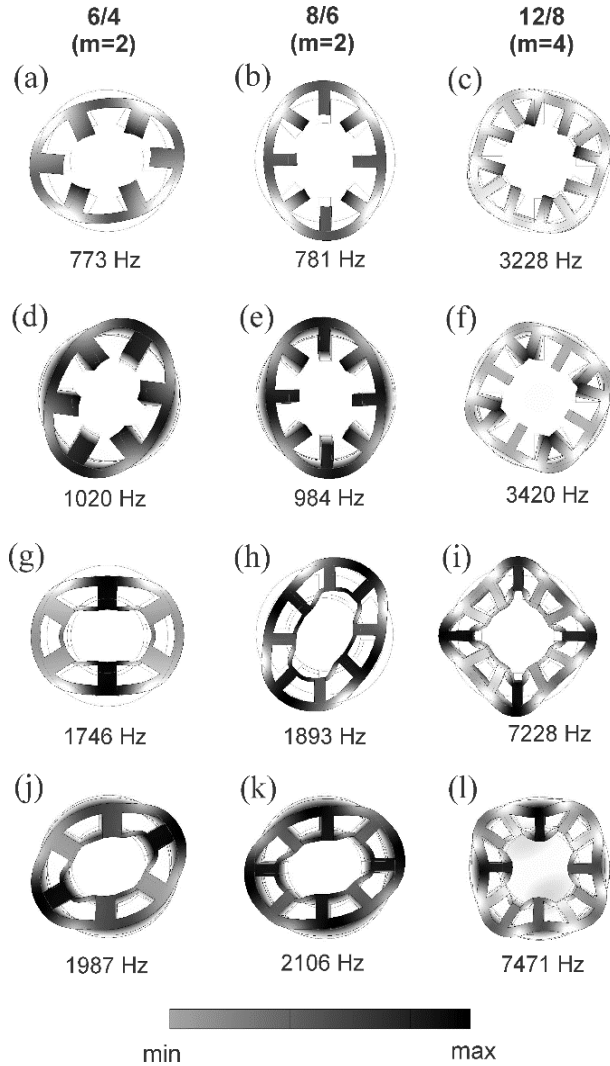


Fig. 6. Comparison of 3D numerical modeling results for the SRM stators with  $N_s = 6, 8$  and  $12$ .

In conclusion, the results of 3D numerical modeling (Fig. 6 and Table 3) demonstrate that the stator modifications such as integration of spacers and/or end-bells constraint the intensity of the stators' motion and significantly increase the natural frequencies of the SRM stators. This may in turn result in reduction of undesirable noise and vibration that may be produced by the SRM.

Table 2: Natural frequencies of the modified SRM stator models ( $f$ ) with respect to the natural frequencies of the reference/baseline SRM stator models ( $f_0$ ) with  $N_s = 6, 8$  and  $12$

SRM 6/4		SRM 8/6		SRM 12/8	
$m = 2$		$m = 2$		$m = 4$	
(Hz)	$f/f_0$	(Hz)	$f/f_0$	(Hz)	$f/f_0$
Reference Stator ( $f_0$ ):					
773	1	781	1	3228	1
With One End-bell ( $f$ ):					
1020	1.32	984	1.26	3420	1.06
With Spacers ( $f$ ):					
1746	2.26	1893	2.42	7228	2.24
With One End-bell and Spacers ( $f$ ):					
1987	2.58	2106	2.70	7472	2.31
With Two End-bells ( $f$ ):					
2496	3.23	2595	3.32	7785	2.41
With Two End-bells and Spacers ( $f$ ):					
3068	3.97	3241	4.15	7878	2.44

**C. Influence of stator pole geometry modifications on natural frequencies**

Within the third part of the study, the influence of modification of stator pole geometry (i.e., stator teeth) on the natural frequencies of the SRM stator was numerically investigated. The stator geometry modifications were done on the SRM stator with  $N_s = 12$ . Three different modifications of the reference/baseline stator pole geometry (shown in Fig. 7 (a)) were done as follows: the first modification of the baseline stator pole was a trapezoidal pole (Fig. 7 (b)), the second one was a trapezoidal pole with a circumference of pole and the pole's root (Fig. 7 (c)) and the third one was a trapezoidal pole with a circumference of pole and the pole's root (Fig. 7 (d)). The geometrical modifications with respect to the reference/baseline geometry are marked with red color in Fig. 7. The results of numerical modeling are displayed in order of increasing natural frequency for different studied geometries in Fig. 8.

The numerical results in Fig. 8 show that the natural frequencies of the stator can be altered by modifying the stator pole geometry. The highest natural frequency as compared to the reference/baseline model (shown in Fig. 8 (a)) is obtained with the trapezoidal pole with a circumference of pole and pole's root as shown in Fig. 8 (d) (i.e., from 3228 Hz to 3593 Hz). These results can be particularly useful for the SRM designs which require higher input currents and thus more space for the windings with a larger cross section area of the wire. Similar increase in natural frequency by modifying the stator pole geometry can be obtained in other SRM configurations (e.g., 6/4 and 8/6 SRM stator designs). However, the geometrical modifications of the poles

need to be selected according to the desired winding fill factor and the stator magnetic load ability.

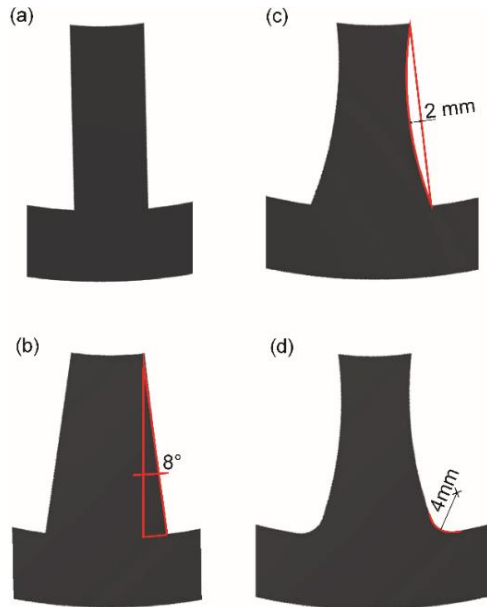


Fig. 7. The reference/baseline stator pole geometry the SRM 12/8 (a) and the modified stator poles (b) trapezoidal pole; (c) trapezoidal pole with a circumference, and (d) trapezoidal pole with a circumference of pole and the pole's root.

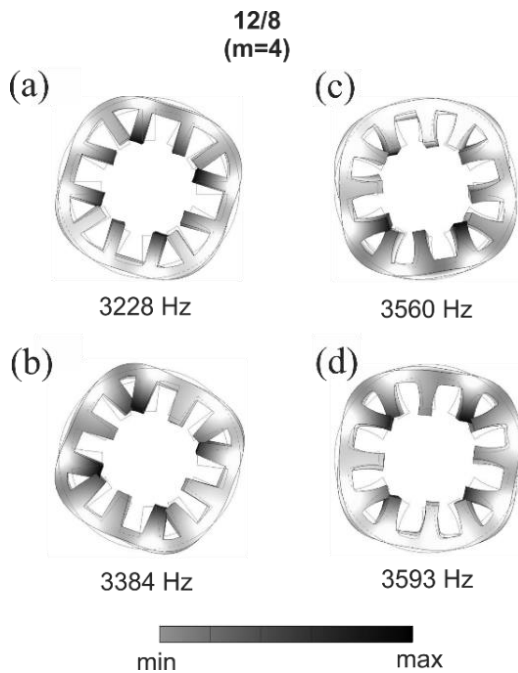


Fig. 8. Numerical results: (a) the reference/baseline stator pole geometry, (b) trapezoidal pole, (c) trapezoidal pole with a circumference, and (d) trapezoidal pole with a circumference of pole and pole's root.

#### D. Comparison between numerical and analytical results

Within the final part of the study the natural frequencies of the studied baseline SRM stator geometries were also analytically calculated by employing Jordan's law (Equations (2), (3) and (4)). The analytical results (i.e., calculated natural frequencies of the stators approximated with cylindrical rings in 2D) are compared to the numerically calculated natural frequencies of the SRM stators in 3D (Table 3).

Table 3: Comparison between numerical and analytical results

$N_s$ and $m$	Analyt. (Hz)	Numer. (Hz)	Difference (%)
$N_s = 0, m = 2$	962	982	2
$N_s = 0, m = 4$	5218	5174	0.85
$N_s = 6, m = 2$	736	773	5
$N_s = 8, m = 2$	768	781	1.8
$N_s = 12, m = 4$	3991	3228	23

The smallest difference is obtained between the numerical model of hollow cylinder (stator without poles  $N_s = 0$ ) and the analytical solution (with the corrective factor  $\Delta_m = 0$ ) (i.e., 2% and 0.85% for  $m = 2$  and  $m = 4$ , respectively). However, this model can result in erroneous calculations of natural frequencies if used as an approximation for the SRM stators with teeth/poles (e.g.,  $f = 733$  Hz for the SRM ( $N_s = 6, m = 2$ ) vs.  $f = 962$  Hz for the SRM ( $N_s = 0, m = 2$ ), Table 3).

As for the stator models with poles, better agreement was obtained between numerical and analytical results for the modal shapes  $m = 2$  (the difference between the numerical and analytical results was 5% and 1.8%, for 6/4 and 8/6 SRM, respectively) as compared to the results calculated for modal shapes  $m = 4$  for 12/8 SRM stators (the difference between the numerical and analytical results was 23%). The comparison between the analytical and numerical results (Table 3) implies that analytical solution obtained by using Jordan's law (if the corrective factor is taken into account) is good enough to be used as a rapid, but rough estimation of natural frequencies of the SRM stators for modal shapes  $m = 2$ . A precise calculation of natural frequencies requires a detailed 3D numerical modeling by considering all geometrical and material properties.

The difference between the analytical and numerical results occurs due to the fact that the analytical results are obtained using Jordan's law which describes the natural frequency of a cylindrical ring which approximates the geometry of the SRM with multiple poles in 2D, while the numerical calculation are performed for the SRM stator geometry in 3D [14]. Better agreement between the two sets of results can be obtained by considering solutions of extended Jordan's law [14], or more complex analytical models [12,14].

#### IV. CONCLUSION

The natural frequencies and thus the level of acoustic noise and vibration strongly depend on geometrical properties of the SRM stator which need to be carefully planned and designed within the pre-construction stage. This study proposes the solutions on how to select an appropriate stator SRM geometry in order to increase its natural frequencies beyond the operational frequencies, and thus to mitigate the electromagnetically excited noise and vibration of the SRM. The natural frequencies of different SRM stator configurations were numerically and analytically investigated by virtue of modal analysis theory. The obtained results are successfully verified based on the previously published theoretical and experimental data.

The obtained results demonstrate that the natural frequencies of the SRM stator can be increased, and thus the noise and vibration can be mitigated by performing the following stator modifications: 1) by increasing the number of the stator pole, 2) by decreasing the height of the stator pole, 3) by increasing the thickness of the stator yoke, 4) by inserting the spacers between the poles, 5) by adding the end-bells, and 6) by modification of the stator pole geometry. On the other hand, the numerical simulations show that the increase in height of the stator's poles results in decrease of natural frequency.

Based on the comparison between the numerical and analytical results it can be concluded that that analytical solution by using Jordan's law is good enough to be used as a rapid first estimation of natural frequencies for the modal shape  $m = 2$ . However, for precise calculation of natural frequencies the detailed 3D numerical analysis considering all geometrical and material properties of SRM need to be used.

The results obtained in this study can provide useful guidelines for further experimental testing and design of SRM machines as well as for other types of electrical machines. Namely, the results can be successfully used in further investigations on the undesirable noise and vibration emitted from the SRM which could also consider the whole stator assembly comprising the windings, the insulation components as well as the stator laminations needed for axial modal shape analysis. In addition, further investigation of adequate SRM configurations can also involve different optimization procedures; the results obtained in this study can therefore be successfully used as input data into the optimization algorithms. To conclude, the findings provided in this study can have important implication in efficient machine design planning and manufacturing process of a SRM with significantly reduced magnetically excited noise and vibration.

#### ACKNOWLEDGEMENT

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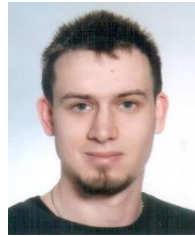
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# Shielding Effectiveness Improvement of Non-Metallic Transparent Enclosures Using Gold Nano-Layer Deposition

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**Abstract** — In this paper we present the results of a simple optically transparent enclosure with efficient electromagnetic shielding. Firstly, Indium Tin Oxide (ITO) enclosures are designed for simultaneously providing high transmittance within a visible range and good shielding effectiveness (SE), where weight reduction, small size, and transparency are challenges. In the next step, gold Nano-layer films of various thicknesses are deposited on the patterned ITO glass using the thermal evaporation method. Then, by making a trade-off between optical transparency and SE, an optimized gold film thickness is selected. To assess the proposed approach, the final experimental results of copper, ITO, and gold deposited enclosures are compared. The results showed that a 9nm gold deposited ITO enclosure, provides efficient electromagnetic shielding better than the ITO enclosure (about 15 dB better at the resonance frequency and much closer to the copper enclosure) and at the same time, acceptable optical transparency.

**Index Terms** — Deposition, gold Nano-layer, shielding effectiveness, transparent enclosure.

## I. INTRODUCTION

Today, the growth of wireless communications, such as satellites and Wi-Fi connections, electromagnetic shielding of high frequencies is needed. Electrical and electronic devices are commonly housed into metallic enclosures to reduce emissions or improve their immunity. From the viewpoint of electromagnetic compatibility due to the existence of slots and apertures for signal and power cable penetration and heat dissipation, the

shielding efficiency of electromagnetic enclosures is degraded [1]. The ability of an electromagnetic enclosure to reduce the effect of undesired emission is represented by shielding effectiveness (SE). SE is defined as the ratio of field strength in the absence of an enclosure to the presence of an enclosure [2]. Metallic enclosures with apertures have been studied extensively in other previous work [3, 4]. Also, analytical models have been developed for a fast and accurate estimation of the metallic enclosures SE [5-7]. Nonetheless, a metallic enclosure cannot be considered as a good candidate for electromagnetic shielding, where weight, size reduction, and transparency are challenging.

Although meta-material slabs can provide good shielding with a reduced weight and size as well as can be tailored to obtain frequency selectivity [8], they are challenging especially when an optimum condition between optical transparency and SE is necessary.

Recent attempts to find transparent materials with high SE and good optical transparency have led to technological solutions [9] for applications such as displays of electronic devices, monitors, electrical panels, fashion electronic devices and integration with solar cells in Nano and cube satellites. Transparent enclosures can be realized on transparent conductive films, such as Indium tin oxide (ITO), fluorine-doped tin oxide, and silver-coated polymer film. Among the transparent conductors, ITO is more desirable as it offers a reasonable trade-off between optical transparency and minimum electrical resistivity [10-12]. However, the practical application of transparent ITO enclosures has been limited by their lower SE rather than metallic enclosures [13]. In recent years, a lot of effort has been



made to fabricate transparent composite films with high SE. This has led to the usage of nanoparticle materials. For example graphene based composites have been used to obtain better SE for transparent conductive films [14–16]. The authors in [17] have attached a monolayer CVD graphene film on quartz substrate to a transparent ITO enclosure. Although, result of the proposed enclosure has demonstrated better electromagnetic shielding, this graphene loaded ITO enclosure has been considered entirely closed as there were no apertures in it. So, its practical application has been reduced dramatically.

In this paper, we provide a practical and simple solution to improve the SE of the transparent ITO enclosure with a circular aperture. In the first step, CST simulations are used to find the SE of ITO enclosures. Then, an optimum gold Nano-layer thickness is selected to make a trade-off between SE and the optical transparency of gold deposited ITO enclosure. Finally, in order to demonstrate the effectiveness of the method, performance results of copper, ITO glass, and gold Nano-layer deposited enclosures are compared.

This paper is organized as follows: In Section II, we discuss the characteristics of transparent conductors and the design and analysis of transparent enclosures is presented. In Section III, the fabrication of transparent ITO enclosure is described. Gold Nano-layer deposition using thermal evaporation technique is described in Section IV. Also, the selection of optimum gold Nano-layer thickness is presented in this section. Measurement results of copper, ITO and gold Nano-layer deposited enclosures are provided in Section V. The conclusion is given in Section VI.

## II. TRANSPARENT ENCLOSURE

### A. Transparent conductive films

In studies involving optically transparent conductive films, the performance of different materials has been assessed, among which, ITO is highly regarded. ITO is an n-type semiconductor with a wide energy gap (more than 3.5 eV) that offers a good trade-off between high optical transparency and minimum electrical resistivity in the visible light spectrum. An ITO film without layering on a substrate can be considered a good conductor to use in an enclosure, if its sheet resistance  $R_{sh}$  is low to restrict the ohmic losses and its thickness is high enough to limit skin depth losses. The optical transparency of the ITO film is estimated by using (1) [18]:

$$T \cong e^{-\frac{2x_c}{\delta}}, \quad (1)$$

where,  $T_c$  is the conductive film thickness and  $\delta$  is the skin depth for visible wavelengths. According to [19], the sheet resistance of the ITO film can be obtained by using (2), where  $\rho$  is the resistivity of the ITO film:

$$R_{sh} = \frac{\rho}{T_c}. \quad (2)$$

One can see from (1), that a higher optical transparency needs thinner films while (2) shows that lower sheet resistance requires thicker films. In this paper, in order to determine the effects of gold deposition, we have used a 160nm thick ITO film with high optical transparency of about 89% and sheet resistance of 10  $\Omega/\text{sq}$ .

### B. Transparent enclosure design

The Robinson et al. model is a simple analytical model based on transmission line theory that can be used for SE estimation of metallic enclosures with infinite conductivity. It has been developed for practical metallic enclosures with finite conductivity [20]. It still cannot analyse transparent enclosures accurately, due to its inability to take into account multi-layer panel effects. In this paper, we used three layer transparent panels including transparent dielectric (such as Glass, PET polymer sheet, and Plexiglas) and conductive transparent films (see Fig. 1). Hence, to analyse transparent enclosures, we have performed a 3D full wave electromagnetic analysis using the CST Microwave Studio based on the Finite Integration Technique and setting the proper characteristics of the materials constituting the enclosure.

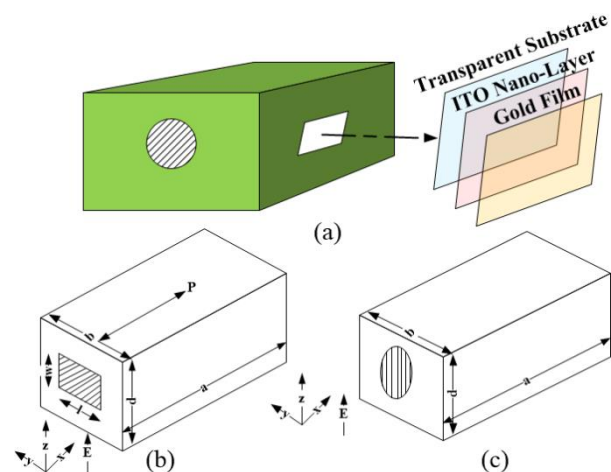


Fig. 1. Transparent enclosure: (a) panel configuration, (b) rectangular aperture, and (c) circular aperture.

SE is the most important factor in metallic enclosure performance while, optical transparency as well as the SE factor should also be considered in transparent enclosures. Therefore, by making a trade-off between these two factors, optimized ITO and glass thickness selection is necessary. So, based on CST optimization tools and Equations (1)–(2), we have selected optimum thicknesses of layers. As a result, we have selected

160nm thick ITO films that are mounted on a transparent dielectric substrate to make two-layer transparent panels. Such transparent enclosures with various dimensions, substrate thickness, and apertures (see Table 1) has been simulated and exposed to the normal incident of the plane wave. The electric field has been calculated in the center of enclosures. Also, for a better comparison between SE of transparent and metallic enclosures, copper enclosures with the same dimensions, apertures and thickness (equal to the substrate thickness) have been analysed, too.

Table 1: Characteristics of ITO Enclosures

Sample	Substrate	Substrate Thickness	Enclosure Dimensions
1	PET Polymer	0.18mm	100mm × 100mm × 100mm
2	Plexiglas	1mm	300mm × 120mm × 300mm
3	Glass	1.5mm	90mm × 40mm × 90mm
4	Glass	2.2mm	100mm × 50mm × 100mm

SE results of ITO enclosures have been given in Fig. 2. One can see there is a significant difference between the SE of ITO and copper enclosures. Therefore, in practical applications, ITO enclosure shielding performance should be improved as will be discussed in the Section V.

### III. ITO GLASS ENCLOSURE FABRICATION

Based on previous results, one can see that an ITO enclosure with square and circular aperture has better SE than one with a rectangular aperture (similar to a metallic one) [20, 21]. As observed in Fig. 2, as the thickness of enclosures is increased, the SE difference between transparent and metallic cases is higher. However, in practical application of transparent enclosure, the thickness should be lower than 2 mm, because of weight and size challenges. Hence, in order to consider the worst case, we have chosen 2.2 mm thick borosilicate glass as a substrate for transparent panels. Additionally, in order to simplify the manufacturing process (cutting and drilling); we have taken ITO enclosures with a circular aperture. Without any damages to the generality of the work, we have considered ITO enclosure dimensions of 100mm × 50mm × 100mm with a circular aperture radius of 5mm. 150nm thick ITO films have been deposited on the substrates using RF sputtering deposition technique.

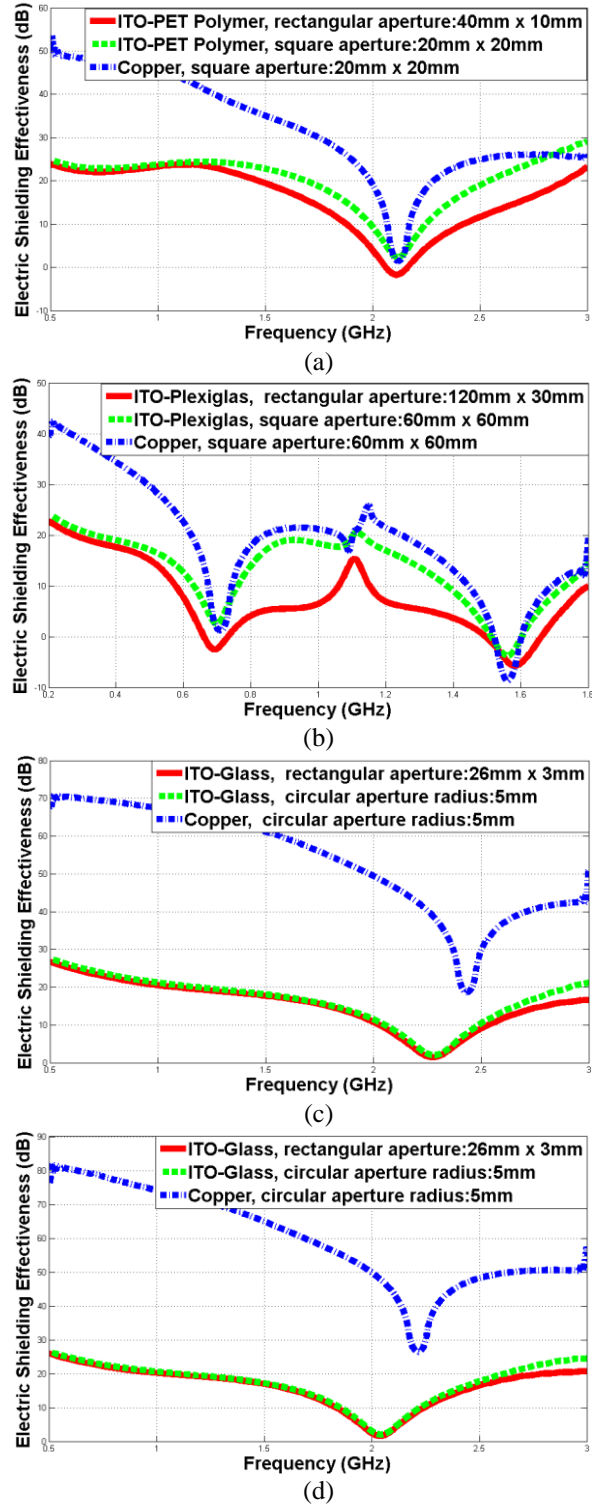


Fig. 2. SE calculation of transparent enclosures with various panels and dimensions: (a) sample 1, (b) sample 2, (c) sample 3, and (d) sample 4.

In order to maintain conductivity at the edges of the enclosures, panels are connected by a silver paste internally and glue strips of copper externally (see sample A, Fig. 3).



Fig. 3. Transparent enclosures have been fabricated using various transparent panels (see Table 2).

#### IV. GOLD NANO-LAYER DEPOSITED ITO ENCLOSURE

##### A. Gold Nano-layer deposition

As predicted in previous section, based on our measurement results (as observed in Fig. 4), the SE of ITO glass enclosure is very low compared to the metallic enclosure (with the same dimensions, aperture and the thickness equal to the substrate thickness).

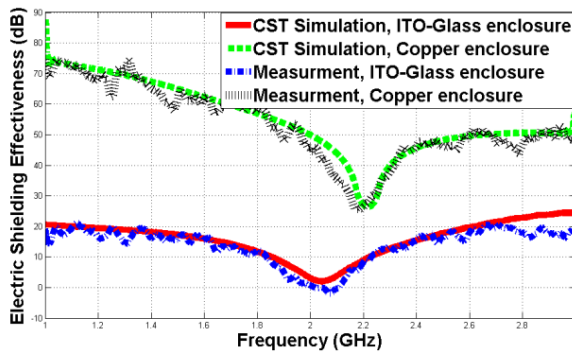


Fig. 4. SE Comparison of copper and ITO enclosures with the same dimensions, thickness and aperture.

To improve the SE, we used gold Nano-layers that homogeneously deposited on the ITO glass panels, using thermal evaporation technique. We choose gold due to its good oxidation resistance, low penetration depth, and high electrical conductivity. This method is based on reducing the surface resistance of the panels to obtain a better conductor and therefore good SE. One technique is to use multilayer conductors. In such a case, several layers are placed in parallel, one on top of the other, and therefore the overall sheet resistance is decreased as observed in (3) [19]:

$$\frac{1}{R_{sh-Total}} = \frac{1}{R_{sh-ITO}} + \frac{1}{R_{sh-Gold}}, \quad (3)$$

where,  $R_{sh-ITO}$  and  $R_{sh-Gold}$  are the sheet resistances of the ITO film and gold Nano-layer, respectively.

##### B. Thermal evaporation deposition technique

Thermal evaporation is utilized extensively to deposit Nano-layer films of materials on the substrates in semiconductor production. In this paper, gold Nano-layer films of various thicknesses have been deposited on the ITO glass panels by this technique. The process has been performed at an atmospheric pressure of  $1e-5$  and supply current of 100A. Before deposition, the panels were cleaned by rinsing them in a detergent solution and later in deionized water. They were finally dried with  $N_2$  gas.

##### C. Selection the optimum thickness of gold film

Here we propose a practical approach to improve SE, based on gold Nano-layer deposition on the patterned ITO glass panels, the configuration of which has been shown in Fig. 1. The first step is to make a trade-off between SE and optical transparency, and the second step is the selection of the appropriate gold Nano-layer thickness. Therefore, gold Nano-layers of various thicknesses in the range of 3nm to 19nm have been deposited on the patterned ITO glass using the thermal evaporation technique.

Results from optical transparency measurements show that thicknesses higher than 13nm lead to very low transparency of about 50% (Fig. 5). On the other hand, based on the results of our simulation and also Equation (3), thicknesses lower than 3nm cannot improve SE, effectively.

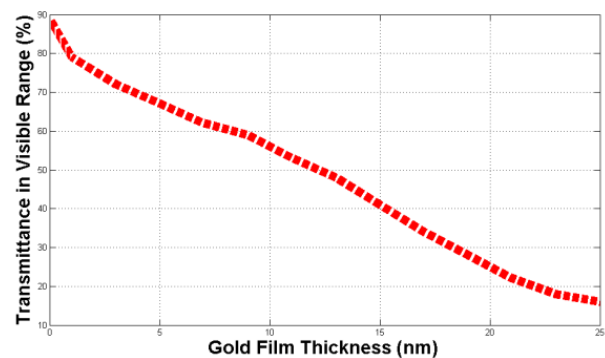


Fig. 5. Optical transparency results of gold deposited ITO glass panels at 550 nm wavelength, versus gold film thickness.

In the final step, 5nm, 7nm, and 9nm gold deposited panels were selected to fabricate transparent enclosures. As mentioned in Section IV, to connect the edges, gold deposited ITO glass panels were connected using silver glue internally and copper adhesive tape externally. Fabricated transparent enclosures can be seen in Fig. 3. Also, panels used in these enclosures have been described in Table 2.

Table 2: Transparent panel samples

Sample	Panel Layers	Layer Thicknesses
A	ITO-Glass	160nm - 2.2mm
B	Gold-ITO-Glass	5nm - 160nm - 2.2mm
C	Gold-ITO-Glass	7nm - 160nm - 2.2mm
D	Gold-ITO-Glass	9nm - 160nm - 2.2mm

## V. RESULT AND DISCUSSION

### A. Measurement

The SE of transparent enclosures was measured within a frequency range of 1 GHz to 3 GHz in an EMC chamber. A standard horn antenna was used to radiate electromagnetic waves, and an electric field measurement probe has been placed at the center of the enclosure (Fig. 6). The SE results of transparent enclosures have been shown in Fig. 7 (a).



Fig. 6. SE measurement setup in EMC chamber.

Also, the optical properties of the composite panels are characterized by spectrophotometry. The transmittance spectra can be recorded within the 300nm–1000nm range by a spectrophotometer for normal incidence. The optical transparency measurements for the mentioned panels, shown in Fig. 7 (b), were conducted using a high resolution optical spectrometer in the optical laboratory.

### B. Discussion

ITO glass enclosures can provide proper shielding and good transparency, simultaneously, but as discussed in the previous section its SE should still be improved to obtain SE closer to metallic enclosures. A homogenous deposition base on gold Nano-layer has been proposed in this paper. To assess the performance of the improvement method, a 2.2mm thick copper enclosure with the same dimensions and aperture (equal to the ITO enclosures) has been fabricated and measured too. The performance results of copper, ITO glass, and gold deposited ITO glass enclosures are compared in Fig. 8 (a) and Fig. 8 (b). As predicted by analysis results, measurement results show that, gold Nano-layer deposition increases the SE significantly and reduces optical transparency which verifies the theoretical relationships (1) to (3).

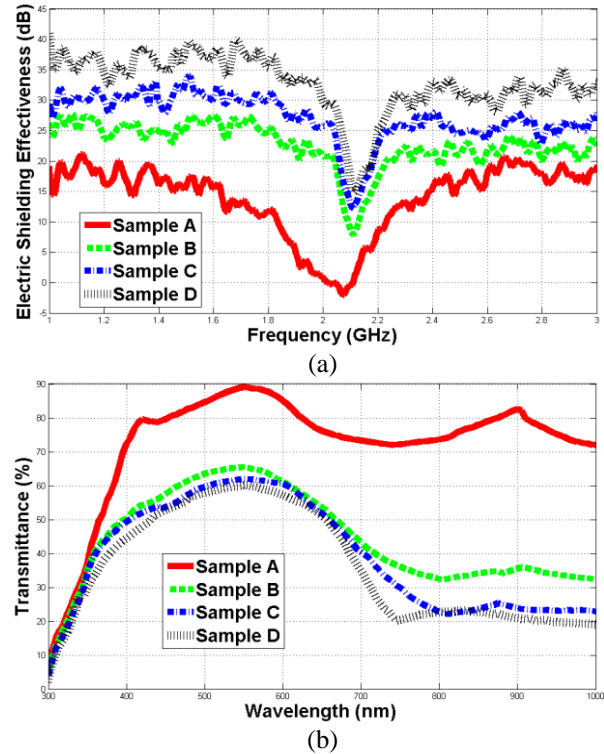


Fig. 7. Performance of transparent enclosures: (a) measured SE, and (b) measured optical transparency.

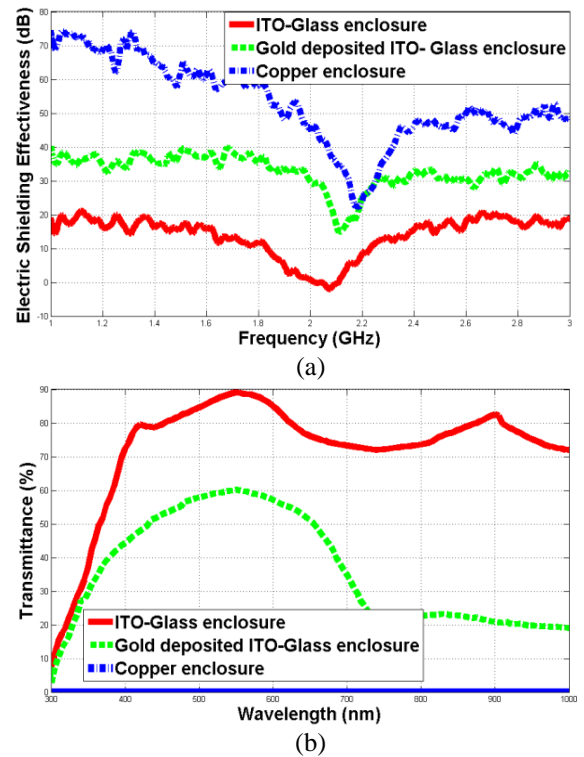


Fig. 8. Performance comparison between transparent and metallic enclosures: (a) SE and (b) optical transparency.

Therefore, optimized performance for transparent enclosures has been obtained by making a trade-off between SE and optical transparency factors. From the results, it can be inferred that while ITO glass enclosure does not have desirable SE, 9nm gold Nano-layer deposited ITO glass enclosure has more than 15 dB better SE at the resonance frequency (and much closer to the SE of metallic enclosure) as well as acceptable optical transparencies for practical applications (about 60% transmittance in 550nm wavelength), while a copper enclosure cannot provide any transmittance in the visible range. As shown in this paper, the proposed method has been used for 2.2mm enclosure (which is usually the worst case for transparent applications). This method improves the SE of ITO glass enclosures more efficiently (compared to the metallic enclosure performance), especially when the thickness of the enclosure walls is reduced.

## VI. CONCLUSION

Metallic enclosures cannot be good candidates for electromagnetic shielding, where weight, size reduction, and transparency are challenging.

In this paper, we have proposed gold-ITO enclosures that can overcome mentioned challenges. Since an ITO enclosure cannot provide the desirable SE, in this paper we have proposed a practical solution to improve the performance of transparent enclosures. Gold Nano-layer films of various thicknesses were deposited on the patterned ITO glass panels to achieve better SE. Finally, by making a trade-off between SE and optical transparency, the optimum thickness of gold Nano-layer was selected. To assess the performance improvement method, copper, ITO, and gold deposited ITO enclosures have been compared. The results showed that a transparent enclosure with 9nm gold Nano-layer deposited ITO glass panels provides efficient SE and the same time acceptable transmittance within a visible range, while metallic enclosures cannot overcome transparency challenges. The proposed technique can improve the SE of ITO glass enclosures efficiently, especially when the thickness of the enclosure walls is reduced.

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# Characterization of 4 Element Compact Microstrip Patch Antenna Array for Efficient Null Steering

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**Abstract** — Null steering in radiation pattern of linear antenna array is essential for minimizing degradation in signal-to-noise ratio performance due to undesired interference. High directivity and miniaturization are other key factors for proficient design of antenna arrays. Thus, the development of small sized antenna elements with high directivity is strategic point of interest. Consequently, the four element bowtie patch antenna array has been designed and simulated using IE3D software, and compared with conventional rectangular patch antenna array. Null steering is performed by varying element excitations in isotropic antenna array, rectangular patch antenna array and bowtie patch antenna array with main emphasis on position and depth of nulls to analyze the radiation characteristics of the designed models. The mutual coupling effects at different inter-element spacings are also presented for both the designed arrays. It has been observed that using bowtie patches as the element in the array brings about comparatively more effective null steering, and enhances the performance of the array as compared to the conventional rectangular patches.

**Index Terms** — Antenna array, isotropic elements, mutual coupling, non-isotropic elements, null steering, Schelkunoff polynomial method.

## I. INTRODUCTION

The demand for extended functionalities provided by wireless communications has risen beyond all expectations over the last decade. With the growing demand for wireless communications, the need for better coverage, improved capacity and higher transmission has risen. In wireless communication systems, null steering is of great significance for rejecting unwanted interference while receiving the desired signal. Null steering in antenna radiation pattern finds its applications in radar, sonar and many communication systems for minimizing degradation in signal-to-noise ratio performance due to undesired interference [1]. Further, the electromagnetic interferences between radiating elements in an antenna array, is expressed by the

modification of the surface currents distribution. This phenomenon, called mutual coupling, depends on the antenna type and the distance between its elements. The coupling between the elements of antenna array has a great importance in the design of antenna arrays, because it may cause a change in the radiation pattern.

Methods of null steering in antenna arrays including controlling the amplitude-only, phase only, position only and complex amplitude, i.e., amplitude and phase both have been extensively used for isotropic antenna arrays. Null steering by controlling the complex weights involves larger degree of freedom, which makes it the most effective but also most costly method because of the controllers used for phase shifters and variable attenuators for each array element [2]. Moreover, the computational time to find the values of element amplitudes and phases also increases with the increase in number of elements in the array [3]. The phase-only and position-only nulling methods are inherently non-linear and, thus cannot be solved by analytical methods without any approximation. The nulling equations for phase only control can be linearized by assuming that the phase perturbations are small, but it cannot be used to place nulls at symmetric location with respect to the main beam. Methods based on non-linear optimization techniques [4] have been proposed for steering the nulls symmetrically with respect to the main beam, but the resultant patterns of these methods have significant pattern distortion because of large phase perturbations used. Moreover, null steering with small phase perturbations results in increased side lobe level (SLL) in the direction symmetric to nulling direction with respect to the main beam [5]. The nulls can be steered in symmetrical directions with respect to the main beam by element position control method using a mechanical driving system such as servomotors [6]. Null steering with amplitude-only control, make use of a set of variable attenuators to adjust the element amplitudes. The number of attenuators and the computational time are halved, if the element amplitudes have even symmetry about the centre of the array [7].

There has been a lot of research work being carried out on null steering of antenna array of isotropic elements using various optimization techniques such as Genetic Algorithm (GA), Simulated Annealing (SA), Particle Swarm Optimization (PSO), Artificial Bee Colony (ABC), Tabu Search Optimization, Bacterial Foraging Algorithm, Backtracking Search Optimization Algorithm [2, 8-12] and many more. All these methods have their own advantages and drawbacks, but have been less explored for null steering of arrays of non-isotropic elements. Choudhari et al. have used Schelkunoff polynomial method (SPM) for phase controlled null steering of microstrip patch antenna array and found it to be an effective analytic approach to synthesize the null controlled patterns [13]. Dwivedi and Banerjee have applied SPM in combination with PSO for null steering of array of isotropic elements [14]. Much of the work on null steering has been implemented for isotropic antenna arrays. However, there has been relatively little effort in analyzing the use of non-isotropic antenna elements in the array design. An antenna element to be used in array environments should have small size and broad patterns. The scanning performance of an array is generally determined by the element spacing which, in turn, is limited by the element size, hence, the physical size of the element is an important consideration in the design of an array. The better approach to address these issues is to explore the design and analysis of antenna array of non-isotropic antenna elements.

In this paper, a performance comparison of null steering of isotropic antenna array and non-isotropic linear antenna arrays consisting of rectangular and bowtie patches is presented. The nulls are imposed in the direction of interference by using Schelkunoff polynomial method for controlling the excitation amplitudes of array elements instead of phase and position control due to their limitations as discussed above. Hence, a Matlab code has been generated to derive the excitation amplitudes for isotropic and non-isotropic elements and the data is then utilized in the design of 4-element isotropic and non-isotropic microstrip patch antenna array for demonstrating the null steering of the designed antenna arrays.

## II. DESIGN SPECIFICATIONS OF ARRAY AND COPLANAR WAVEGUIDE FEED SECTION

Antenna array size reduction has attracted increasing interest in wireless communication so as to reduce the total size of devices. One way to reduce the total size of an array antenna is to place elements of an antenna array close to each other. But the factor called mutual coupling that effects the radiation pattern, depends on inter element spacing and causes undesirable effects on antenna characteristics [15]. Therefore, the selection of

appropriate design parameters is prerequisite for performance evaluation of the designed models. The process of antenna array design involves the selection of elements and geometry of array, and the determination of the excitations of array elements required for achieving desired radiation characteristics with prescribed location of the major lobe and nulls, SLL, and beamwidth [16].

As reported in literature the antenna pattern synthesis can be classified into several categories. One of these groups requires antenna patterns with narrow beams and low side lobes. This guarantees the radiating or receiving energy to be more focused in specific directions. Various techniques such as the binomial method, Dolph–Chebyshev method, and Taylor line-source are proposed to serve this purpose.

Another group requires that the antenna patterns exhibit a desired distribution in the entire visible region, which is also referred to as beam shaping. A typical example is the design of a sector beam pattern, which allows the antenna array to have a wider angular coverage. This is usually accomplished by using the Fourier transform technique and the Woodward–Lawson method.

A third group usually requires that the antenna patterns possess nulls in desired directions. This property is widely used in smart antenna systems to eliminate the interference from specific noise directions. The Schelkunoff polynomial method is an effective approach to synthesize the null controlled patterns.

Although existing designs offer excellent performance, many other considerations such as miniaturization of antennas and increased directivity have become important due to the increasing demand for small antennas as a result of the rapid development in wireless communications. The bowtie antenna originally proposed by Lodge demonstrates these benefits [17].

Considering the performed investigation in wideband antenna categories, two antenna geometry types: rectangular and bowtie patch antennas have been selected for present study. Both the rectangular patch antenna and bowtie patch antenna shown in Fig. 1 (i) have been designed to operate at resonant frequency of 4.9 GHz. The substrate material that is used is FR4 with dielectric constant 4.4, thickness of 1.6 mm and loss tangent 0.02. The essential parameters for the design of the two antennas, calculated using analytical equations given in [18] and [19] are presented in Table 1.

The coplanar waveguide antenna feeding operates well in wideband frequency to field matching and impedance transforming and they can also be printed on single side of the substrate. For the antenna arrays being studied, the 50 ohm CPW transmission line has been designed with metal strip width  $2a = 1.8$  mm and gap dimension of 0.22 mm, calculated using equations given in [20]. The width of CPW feeding structure is



similar to the width of the patch. For rectangular patch antenna  $L_g = 12.9$  mm and  $W_g = 8.18$  mm. For bowtie patch antenna  $L_g = 8$  mm and  $W_g = 3.88$  mm.

Table 1: Dimensions of antenna elements

Antenna Element	Dimensions	
Rectangular Patch	Length of rectangular patch (L)	14.06 mm
	Width of rectangular patch (W)	18.6 mm
	Area	261.52 mm <sup>2</sup>
Bowtie Patch	L/2	7.07 mm
	W	10 mm
	W <sub>C</sub>	1.0 mm
	S	10 mm
	Area	55 mm <sup>2</sup>

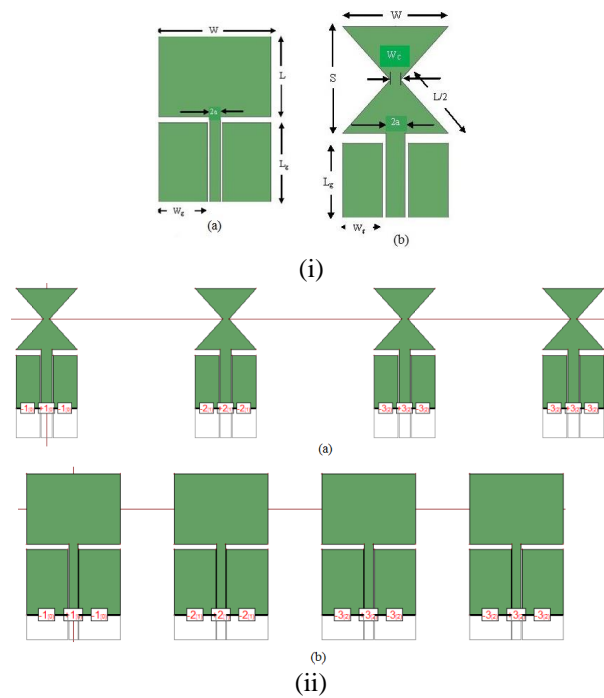


Fig. 1. (i) Geometries of CPW fed: (a) rectangular patch antenna, and (b) bowtie patch antenna. (ii) Geometry of 4-element: (a) bowtie patch antenna array, and (b) rectangular patch antenna array.

The theoretical calculations are verified by simulating the antenna in IE3D software and thereafter arrays of 4-elements with inter element spacing of 29.18 mm ( $\lambda/2$ ) are designed and simulated.

### III. ARRAY SYNTHESIS USING SCHELKUNOFF POLYNOMIAL METHOD

The null controlled patterns can be generated using Schelkunoff polynomial method [16]. The method is

used to calculate the elements excitations from the required number of nulls and their positions taken as input. The array factor for an N-element array with equal spacing, non-uniform amplitude, and progressive phase excitation, is given by:

$$AF = \sum_{n=1}^N a_n e^{j(n-1)(kd \cos \theta + \beta)} = \sum_{n=1}^N a_n e^{j(n-1)\Psi}, \quad (1)$$

where,

N = number of elements;

a<sub>n</sub> = amplitude weight at element n;

$\Psi = kdcos\theta + \beta$ ;

k =  $2\pi/\lambda$  = wave number;

$\lambda$  = signal wavelength;

d = spacing between the elements;

$\theta$  = an incidence angle of signal from the array normal;

$\beta$  = progressive phase.

The array factor of an N-element array is a polynomial of degree N-1 and therefore it has N - 1 zeros. By proper placement of the zeros on the z-plane, a desired array factor can be designed.

Let  $Z = e^{j\Psi} = e^{j(kd \cos \theta + \beta)}$ ,

$$AF = \sum_{n=1}^N a_n Z^{(n-1)} = a_1 + a_2 Z + a_3 Z^2 + \dots + a_N Z^{(N-1)}, \quad (2)$$

The above equation is simply a polynomial in the complex variable z. Recall that a polynomial of order N has N zeros which may be complex. The polynomial for the AF above is of order N-1 zeros. If the zeros are numbered starting from one, the zeros will be 1, 2, ..., N-1. The AF is then rewritten as:

$$AF = a_n (Z - Z_1)(Z - Z_2)(Z - Z_3) \dots (Z - Z_{(N-1)}), \quad (3)$$

where  $Z_1, Z_2, Z_3 \dots Z_{n-1}$  are roots of the polynomial. The roots  $Z_n$  of the polynomial can be positioned on, inside or outside the unit circle. The roots  $Z_n$  that lie on the unit circle contribute to the nulls in the radiation pattern in the fixed directions  $\theta_1, \theta_2, \theta_3 \dots \theta_{N-1}$ , and  $a_1, a_2 \dots a_n$  are the corresponding element excitations which can be obtained by equating (2) and (3).

Hence, we can choose the zeros  $Z_n$  to be whatever we want corresponding to the nulls  $\theta_n$  as:

$$Z_n = e^{j(kd \cos \theta_n + \beta)},$$

and then figure out what the weights  $a_n$  should be to give us the same pattern.

### IV. RESULTS AND DISCUSSION

For investigating the null placement in non isotropic antenna array, the proposed array geometries (Fig. 1 (ii)) of bowtie microstrip patch and rectangular microstrip patch elements shown in Fig. 1 (i) have been designed and simulated using IE3D on a personal computer with Intel Core i7 processor running at 3.40 GHz with 4 GB RAM, and the performance of the

two has been evaluated for different amplitude ratios.

Based on Schellkunoff polynomial method, in 4 element array 3 nulls can be oriented along direction of interference. In this work it is intended to impose two nulls at the peaks of the two side lobes ( $\theta_1 = 45^\circ$  and  $\theta_2 = -45^\circ$ ) and the third null at  $155^\circ$  by changing the amplitude ratio to 0.395:1:1:0.395 as calculated by Schelkunoff polynomial method.

The elevation pattern gain plots shown in Fig. 2 (i) and Fig. 3 depict the symmetric patterns with respect to the main beam because of the element-amplitude's symmetry around the center element of the array. Hence, null placement at one side of the main beam results in occurrence of an image null at the other side of the main beam.

In all the three cases, three nulls have been steered to  $45^\circ$ ,  $-45^\circ$  and  $155^\circ$  by controlling only the element amplitudes as calculated using Equations (2) and (3). With the change in amplitude, the side lobe level decreases from  $-12.29$  dB to  $-25.34$  dB for rectangular antenna array and from  $-13.31$  dB to  $-25.5$  dB for bowtie antenna array. For rectangular antenna array the NDL decreases from  $-16.57$  dB to  $-27.14$  dB at  $45^\circ$ . Null steering is achieved better in bowtie antenna array with considerable decrease in NDL from  $-22.59$  dB to  $-31.75$  dB.

The 2D radiation patterns of the two arrays obtained for different amplitudes are shown in Fig. 2 (ii) and Fig. 4. It can be clearly observed from the 2D radiation patterns that nulls have been successfully steered to desired location by changing the amplitude as obtained by Schelkunoff polynomial method. Further the maximum gain of the major lobe decreases from 6.76 dBi to 6.3 dBi for bowtie antenna array, whereas for rectangular antenna array it decreases from 7.92 dBi to 7.3 dBi.

It can be observed that null steering has been successfully achieved for both the isotropic and non-isotropic patch arrays. The comparative analysis on the basis of position and depth of nulls is presented in Table 2. It can be clearly depicted from the Fig. 2 (i) and Fig. 3 that the pattern symmetry is retained in all the three cases. For isotropic antenna array, the nulls have been steered to  $45^\circ$ ,  $-45^\circ$  and  $155^\circ$  with significant decrease in the depth of nulls. For the rectangular and bowtie patch antenna arrays, the two nulls have been successfully steered towards the peaks of the two side lobes ( $45^\circ$  and  $-45^\circ$ ) except for the third nulling location steered towards the peak of the side lobe at  $135^\circ$  instead of  $155^\circ$ .

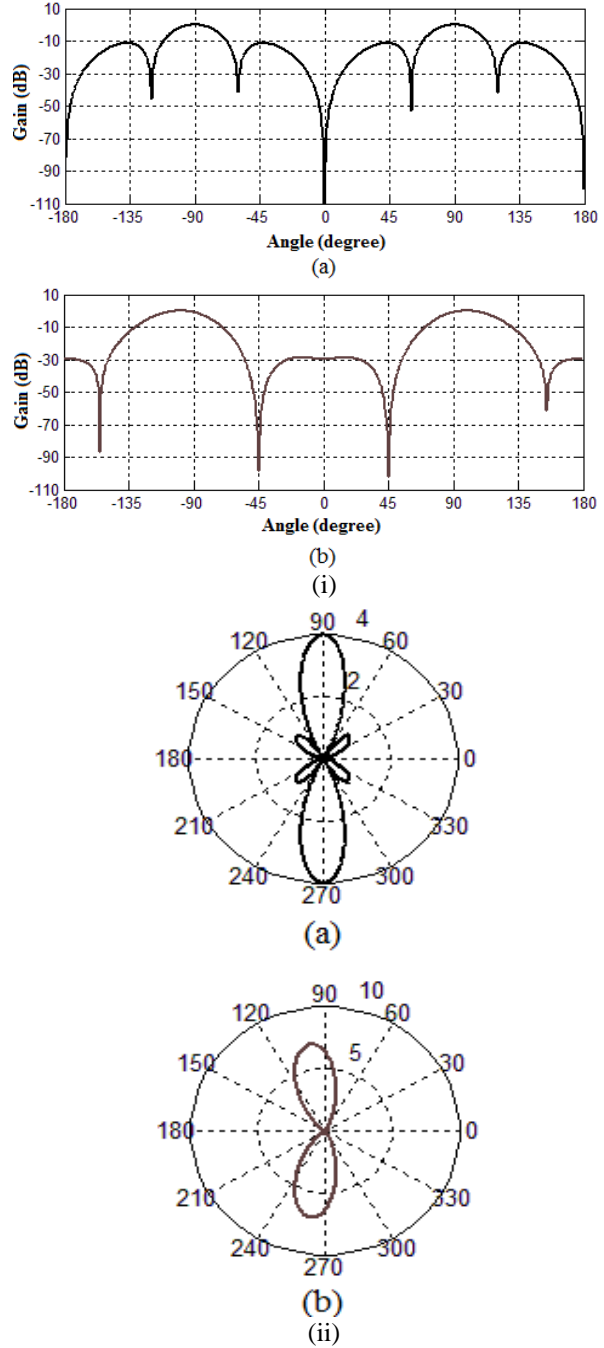


Fig. 2. (i) Gain versus angle plots of 4-element isotropic antenna array with: (a) amplitude 1:1:1:1 and (b) amplitude 0.395:1:1:0.395. (ii) 2D radiation patterns of 4-element isotropic antenna array with: (a) amplitude 1:1:1:1 and (b) amplitude 0.395:1:1:0.395.

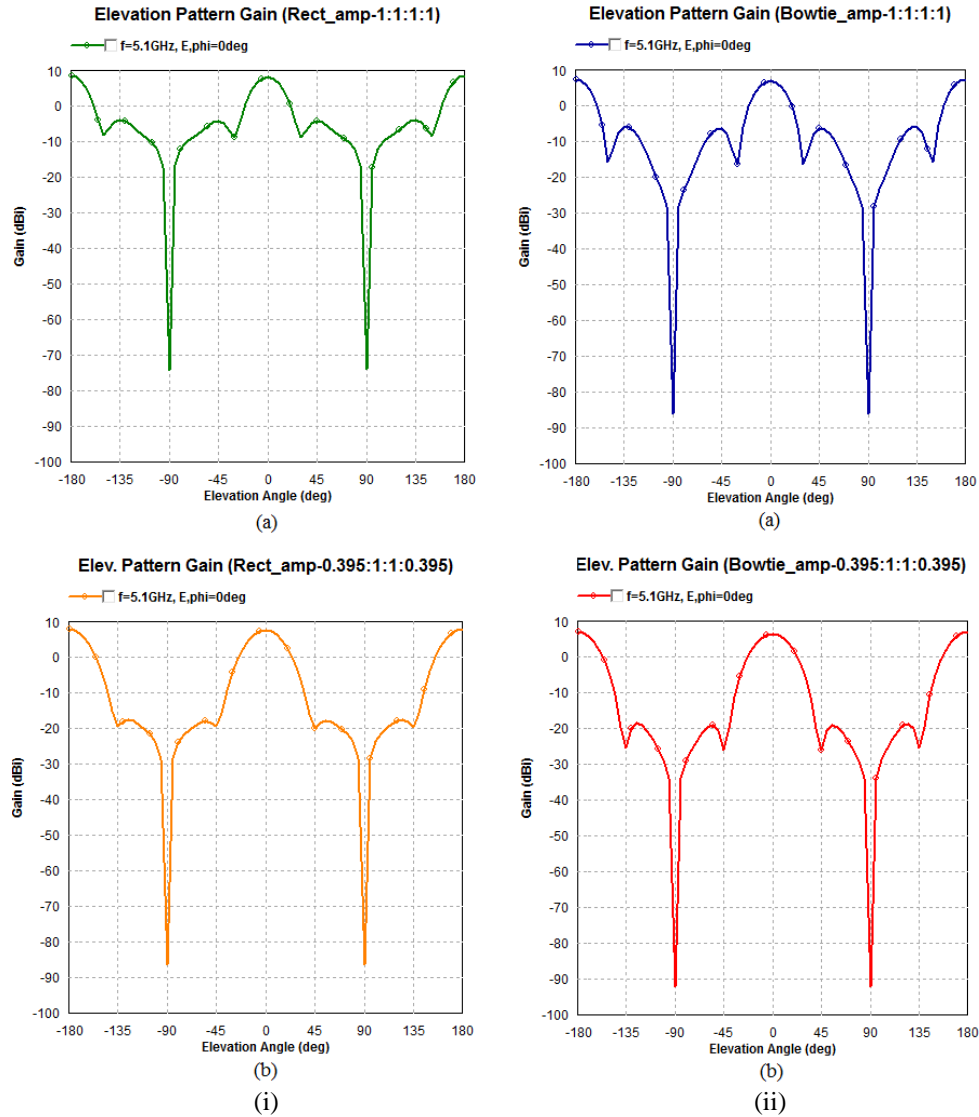


Fig. 3. Gain versus angle plots of (i) 4-element rectangular patch antenna array with: (a) amplitude 1:1:1:1 and (b) amplitude 0.395:1:1:0.395; (ii) 4-element bowtie patch antenna array: (a) amplitude 1:1:1:1 and (b) amplitude 0.395:1:1:0.395.

Table 2: Position and depth of nulls in radiation pattern of 4x1 isotropic and non-isotropic antenna array for different amplitudes

Antenna Element	Amplitude Ratio	Null 1		Null 2		Null 3	
		Position	Depth (dB)	Position	Depth (dB)	Position	Depth (dB)
Isotropic	1:1:1:1	60°	-55	-60°	-40	120°	-40
	0.395:1:1:0.395	45°	-105	-45°	-100	155°	-60
Rectangular	1:1:1:1	30°	-16.57	-30°	-16.49	150°	-16.14
	0.395:1:1:0.395	45°	-27.14	-45°	-26.81	135°	-26.59
Bowtie	1:1:1:1	30°	-22.59	-30°	-22.16	150°	-22.16
	0.395:1:1:0.395	45°	-31.75	-45°	-31.74	135°	-31.39

As can be seen from the plots shown in Figs. 4 and 5 for both the designed antenna arrays, i.e., rectangular as well as bowtie, when we change amplitude ratio to 0.395:1:1:0.395, the gain at  $155^\circ$  and  $-155^\circ$  undergo a slight variation; i.e., from  $-5.6$  dB to  $-1.08$  dB for bowtie antenna array and shows a variation from  $-4.2$  dB to  $-0.15$  dB for rectangular antenna array. However, for both the arrays the gain changes significantly at  $135^\circ$  and  $-135^\circ$  as the nulls present at  $149^\circ$  and  $-149^\circ$  have got steered towards  $135^\circ$  and  $-135^\circ$ , i.e., the direction of maxima of minor lobes (direction of interference).

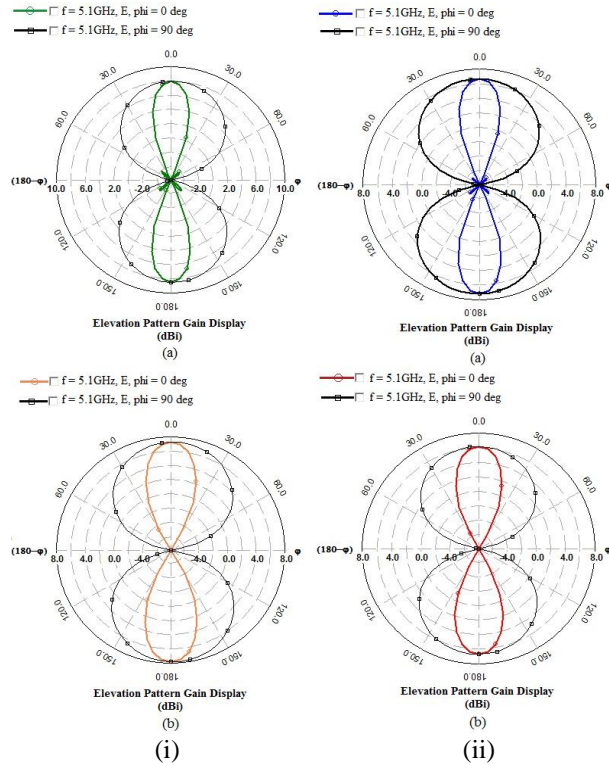


Fig. 4. 2D radiation patterns of (i) 4-element rectangular patch antenna array with: (a) amplitude 1:1:1:1 and (b) amplitude 0.395:1:1:0.395; (ii) 4-element bowtie patch antenna array with: (a) amplitude 1:1:1:1 and (b) amplitude 0.395:1:1:0.395.

Effect of mutual coupling have to be introduced while designing antenna array because in practical arrays this effect changes radiation pattern significantly. Coupling effect produced due to elements interaction while all the elements in array are excited causes elements pattern alter from isolated pattern [21]. Hence, it is important to investigate mutual coupling effect to calculate optimum inter-element spacing and IE3D is a good platform to analyze this effect [22]. To explore the mutual coupling effect, an analysis of the variation of  $S_{12}$  with respect to inter-element spacing has been

presented in Table 3 and the plots are shown in Fig. 5.

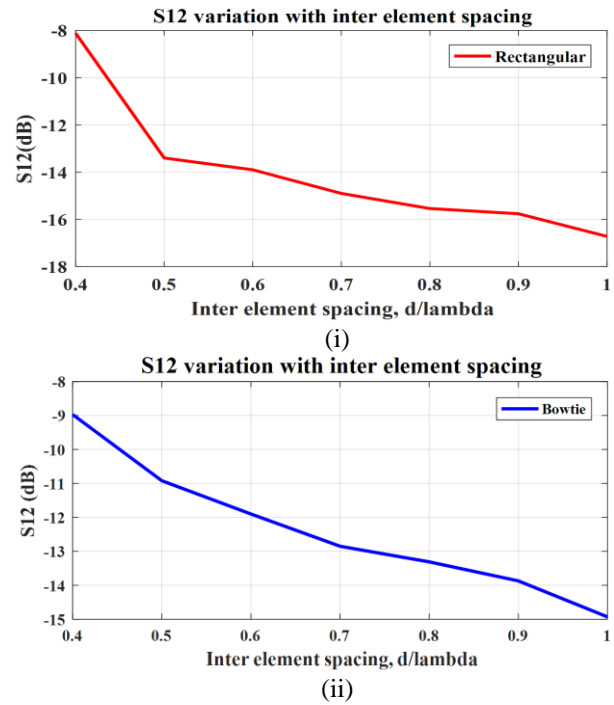


Fig. 5. Mutual coupling of: (i) 4-element rectangular patch antenna array, and (ii) 4-element bowtie patch antenna array.

It is observed from the variation of  $S_{12}$  as a function of inter element spacing, that the mutual coupling gets reduced with increase in the element spacing. Furthermore, both the curves plotted in Fig. 5 shows a significant variation in  $S_{12}$  for the change in spacing from  $0.4\lambda$  to  $0.5\lambda$ . Whereas the variation gets reduced beyond  $0.5\lambda$  as depicted by the gradual slope of the two curves, thus indicating half wavelength distance to be the optimum one.

The comparative analysis performed for non-isotropic antenna arrays of rectangular and bowtie patches indicates better performance exhibited by the bowtie antenna array not only in terms of occupied area but also improved radiation characteristics. The dimensions of the two elements given in Table 1 show that the area of bowtie patch is  $55 \text{ mm}^2$  which is approximately 80% less than that of rectangular patch which leads to much reduced computational time of 31 min 58 sec for bowtie array, whereas the execution takes 10 hr 2 min 29 sec for rectangular array performed on Intel Core™ i7, 3.40 GHz processor. Further, from the performance comparison detailed in Table 3, it can be seen that the value of  $S_{11}$  at the optimum spacing ( $\lambda/2$ ) for bowtie antenna array is  $-20.57$  dB, whereas for rectangular antenna array it is  $-13.72$  dB.

Table 3:  $S_{11}$  and  $S_{12}$  for different inter element spacing  
Element Excitation (Amplitude Ratio) = 0.395:1:1:0.395

Inter Element Spacing	4 Element Bowtie Array			4 Element Rectangular Array		
	Resonant Frequency (GHz)	At Resonant Frequency		Resonant Frequency (GHz)	At Resonant Frequency	
		$S_{11}$ (dB)	$S_{12}$ (dB)		$S_{11}$ (dB)	$S_{12}$ (dB)
$0.4\lambda$	5.15	-17.97	-8.97	4.43	-21	-8.12
$0.5\lambda$	5.12	-20.57	-10.92	4.93	-13.72	-13.4
$0.6\lambda$	5.02	-22.13	-11.9	4.61	-12.19	-13.9
$0.7\lambda$	4.93	-19.67	-12.85	5.53	-16.2	-14.9
$0.8\lambda$	5.07	-17.24	-13.31	5.17	-16.10	-15.54
$0.9\lambda$	5.12	-18.66	-13.87	4.98	-15.74	-15.76
$1\lambda$	5.08	-20.36	-14.93	4.83	-13.73	-16.72

## V. CONCLUSION

Null steering in antenna array of isotropic and non-isotropic elements has been presented in this paper. The algorithm for amplitude controlled null steering of linear antenna array using Schelkunoff polynomial method has been developed in MATLAB. The simulation results demonstrate that by using Schelkunoff method the null steering has been effectively attained in 4-element rectangular antenna array and bowtie antenna array to obtain the desired radiation pattern with nulls imposed at the direction of interferences. The single element bowtie antenna offers the advantage of reduced area but it also achieves high directivity. Further, the results show that using bowtie patches as the element in the array brings about comparatively more effective null steering, and enhances the performance of the array as compared to the conventional rectangular patches, with a decrease in SLL and NDL to  $-25.47$  dB and  $-31.75$  dB respectively, with a significant reduction of the area of the array, which is desirable for application where small antennas are required. It has been observed that the mutual coupling between elements decreases substantially with significant variation at half wavelength spacing.

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